

MATERIAL PROPERTIES IMPORTANT TO THE DESIGN  
OF A LARGE SUPERCONDUCTING MAGNET

Introduction:

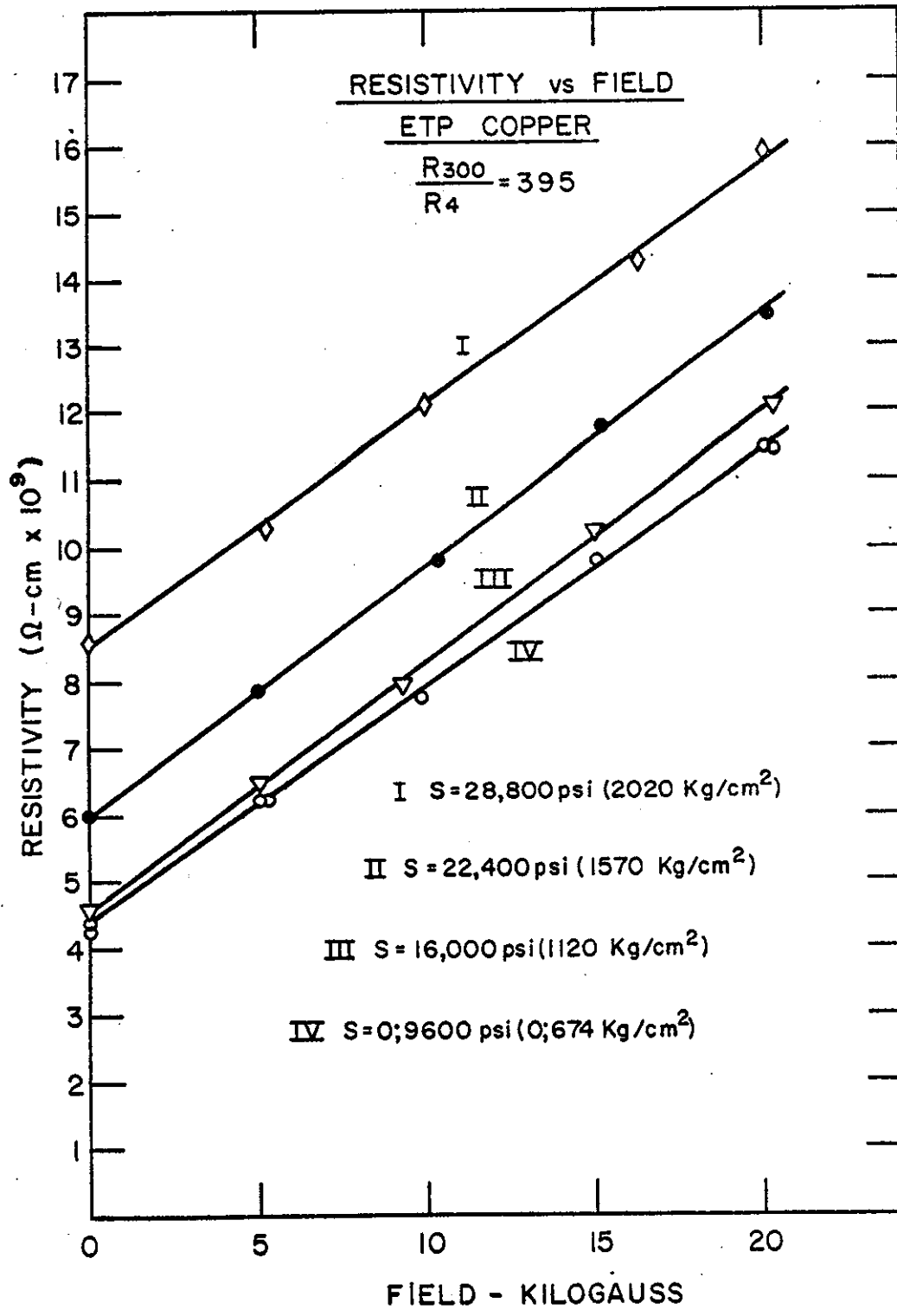
This monograph is a compilation from the literature, from more general compilations, and from private sources, of electrical and mechanical properties of materials important to the design of a large superconducting magnet for a bubble chamber. It is intended for use by the Magnet Committee of the Bubble Chamber Group, Physics Department, Brookhaven National Laboratory, as a source document during the design of large superconducting magnets for a bubble chamber of 14-foot diameter and a test facility of 7-foot diameter. It will be revised and extended as needed during the design effort.

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Brooks, J. M. and Purcell, J. R.,  
Stress vs. Resistivity at Liquid  
Helium Temperature, Argonne Natl.  
Lab., paper presented at 1966  
Cryogenic Conference

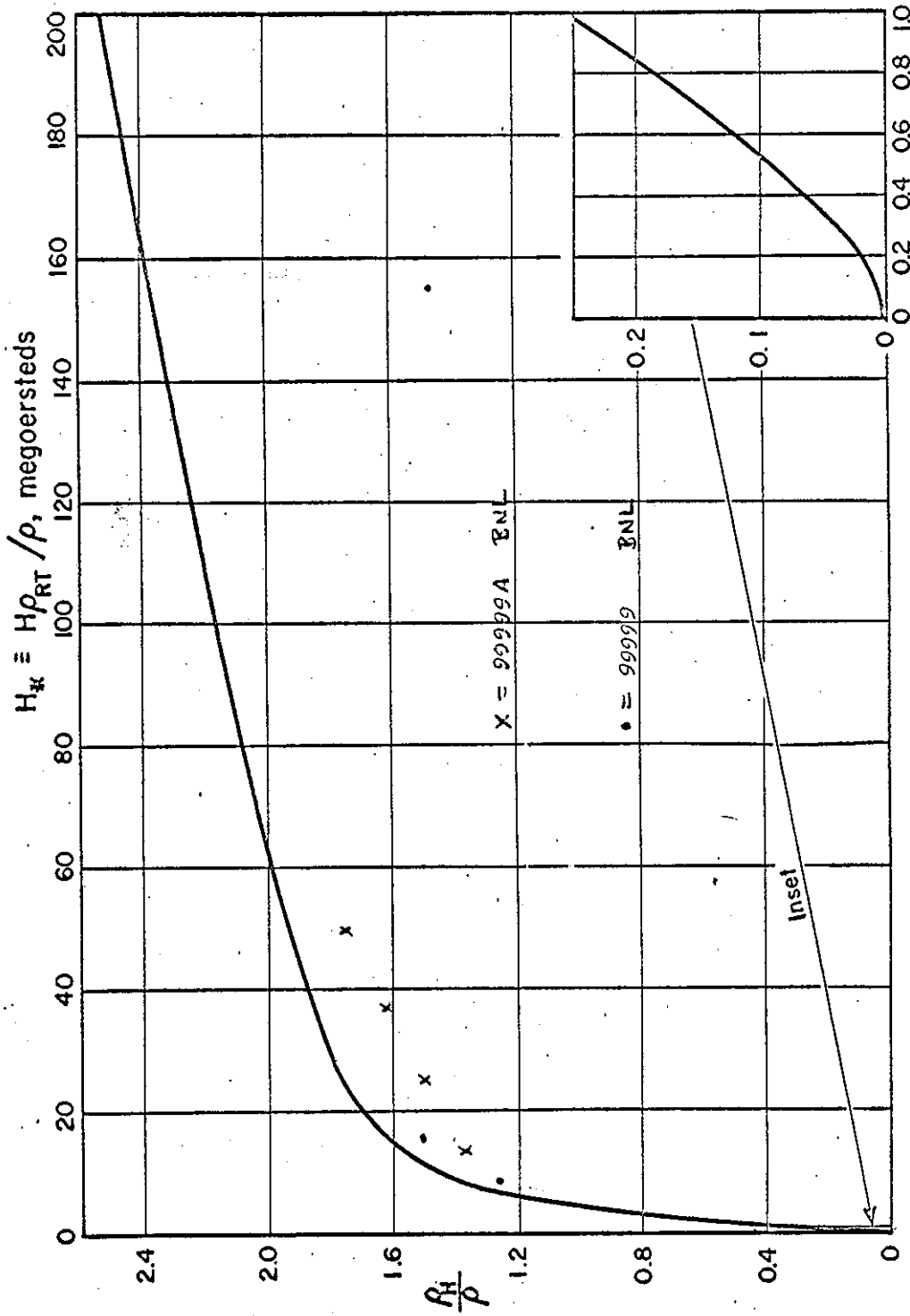


Fig. 3 Transverse magnetoresistance of aluminum as given by equation (6).

$$\rho_B = \rho \frac{H_*^2 (1 + 0.00177 H_*^2)}{1.8 + 1.6 H_*^2 + 0.53 H_*^4}$$

N.B.S. TECH NOTE 218  
CORRUCCINI

10 - 8 CM.

MAGNETO-RESISTIVITY

$\rho(\beta) - \rho(\beta=0) = \rho_{\beta}$

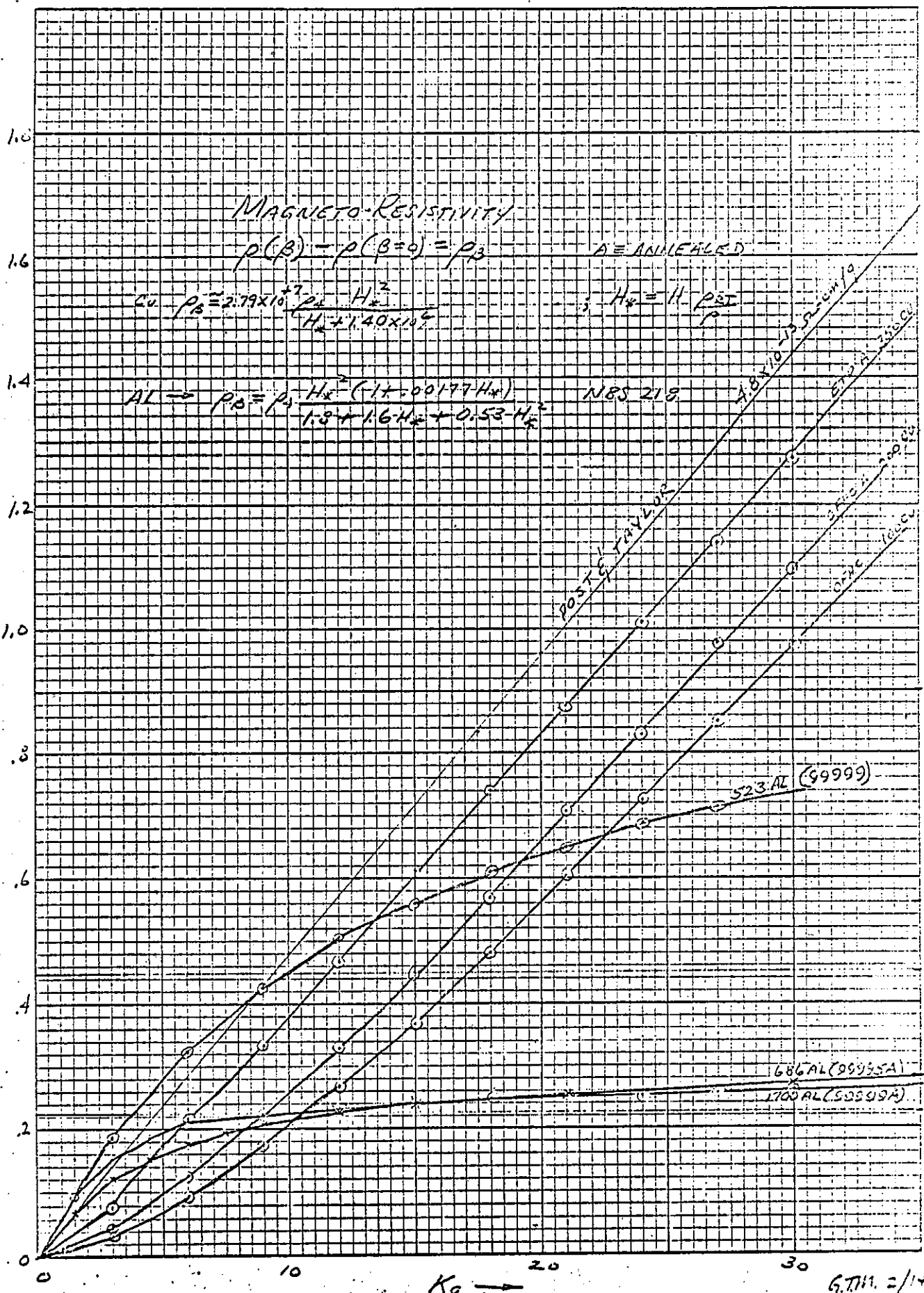
A = ANNEALED

$\rho_{\beta} = 2.79 \times 10^{-7} \frac{\rho_0 H_x^2}{H_x^2 + 1.40 \times 10^6}$

$H_x = 11 \frac{\rho_{\beta T}}{\rho}$

$\rho_{\beta} = \rho_0 \frac{H_x^2 (1 - 0.0177 H_x)}{1.8 + 1.6 H_x + 0.53 H_x^2}$

NBS 219



100

200

300

400

500

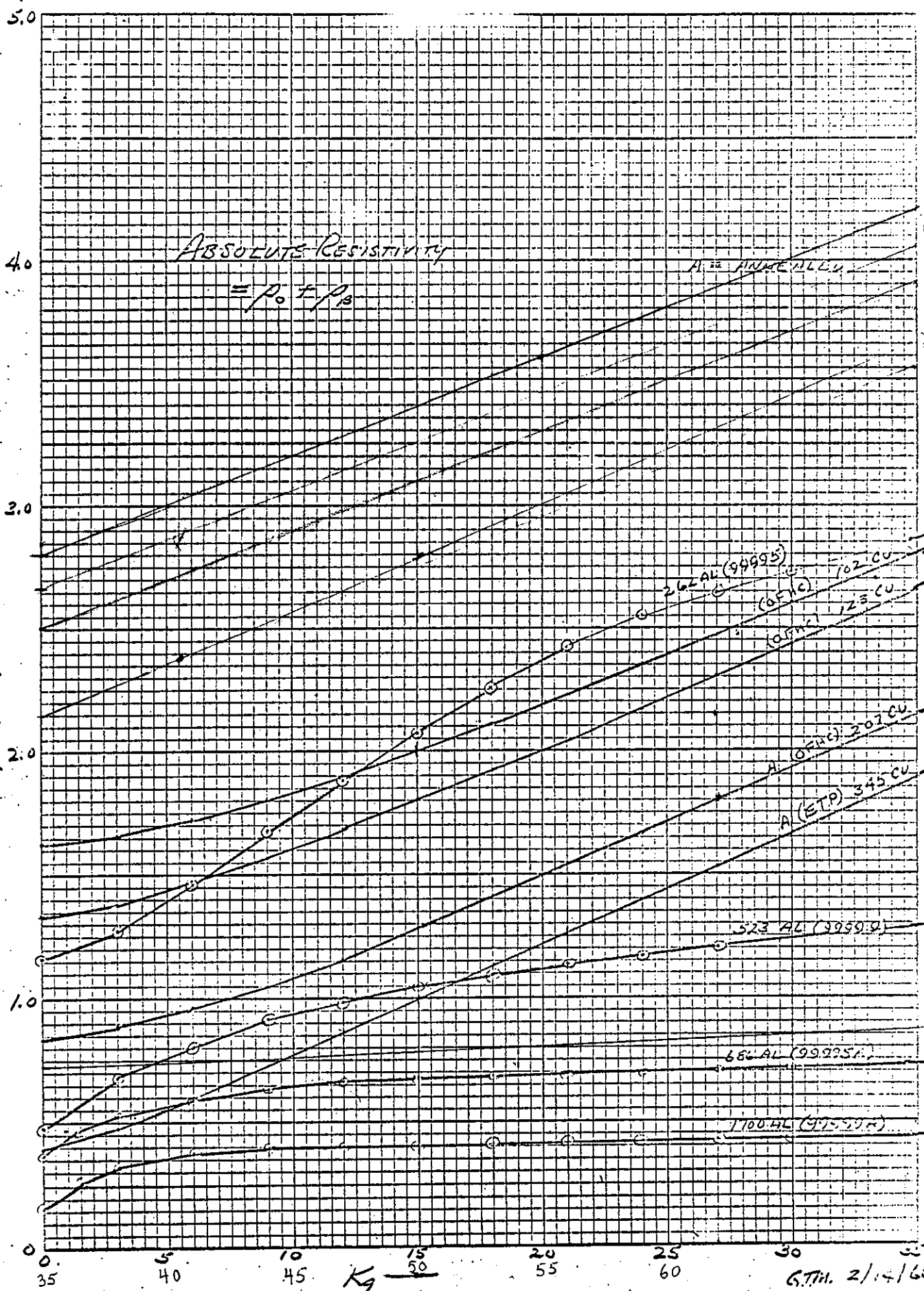
600

G.T.H. = 1/14/66

9 R

K<sub>10</sub> 10 X 10 TO THE INCH 35B-5  
 KUFFEL & ESSER CO. MADE IN U.S.A.

RESISTIVITY (μΩ-INCH) X 10<sup>-8</sup> Ω-CM.

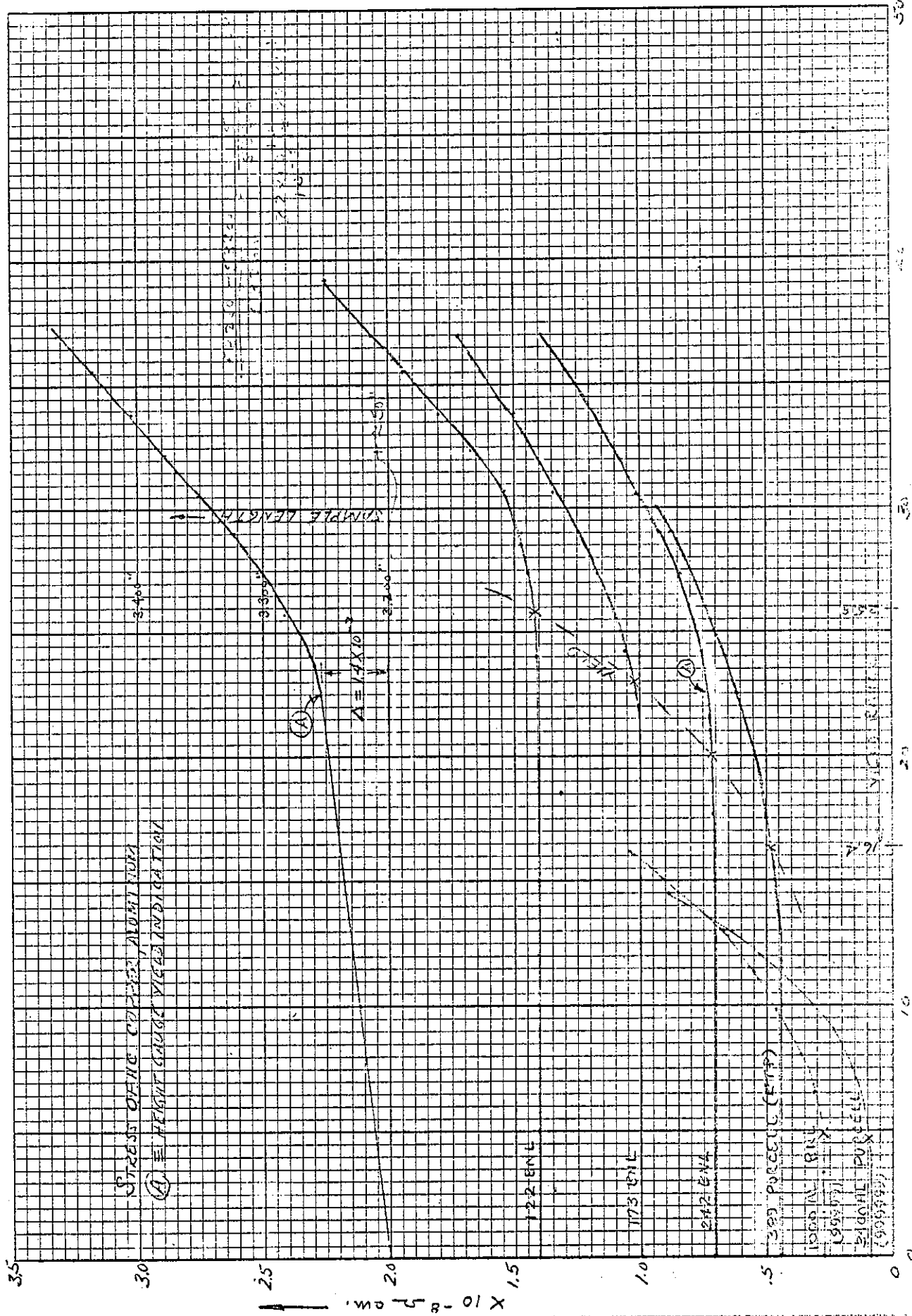


Cu (123)  
 Cu (150)  
 Al (202)

GTH. 2/14/66

019





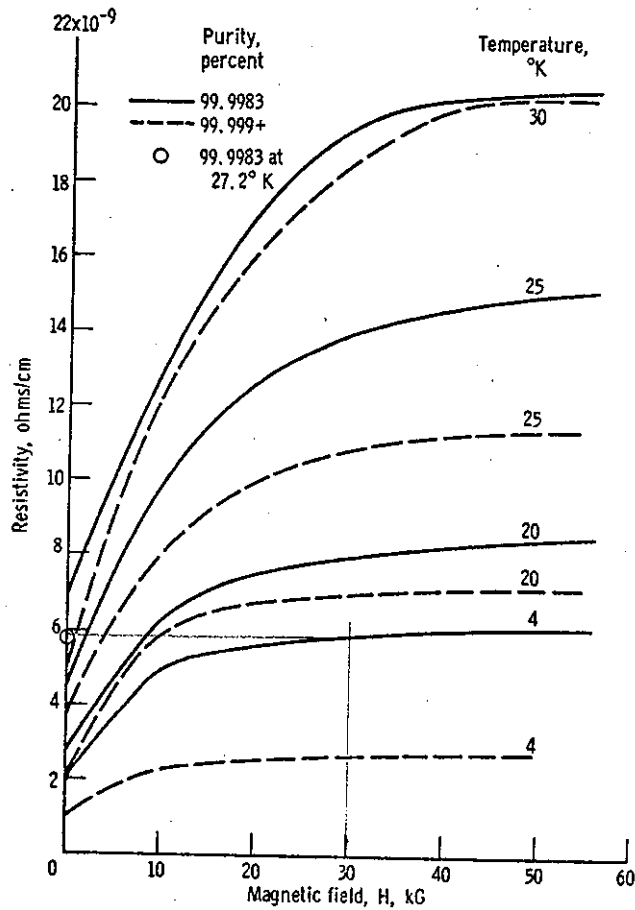


Fig. 1. Magnetoresistance of aluminum.

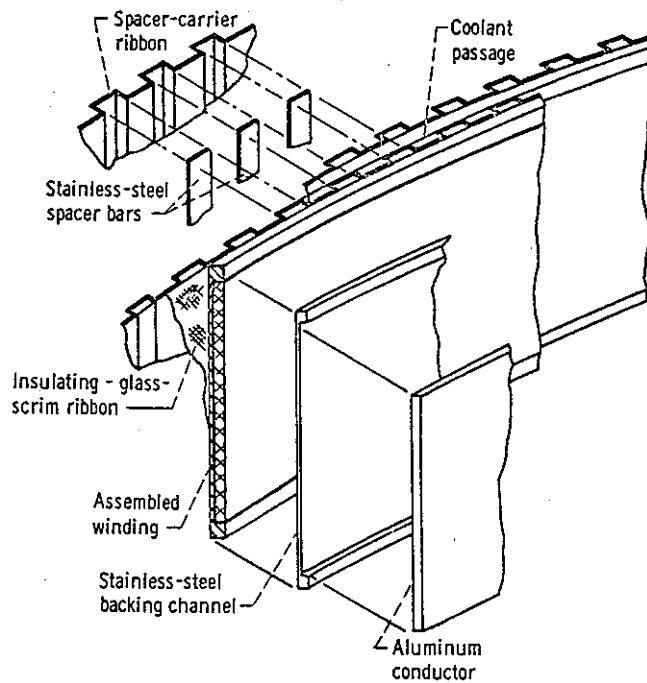


Fig. 2. - Construction of coils of cryogenically-cooled aluminum magnet.

Laurence, J. C. and Coles, W. D., Design, Construction, and Performance of Cryogenically Cooled and Superconducting Electromagnets, NASA TM X-52121 Sept. 1965

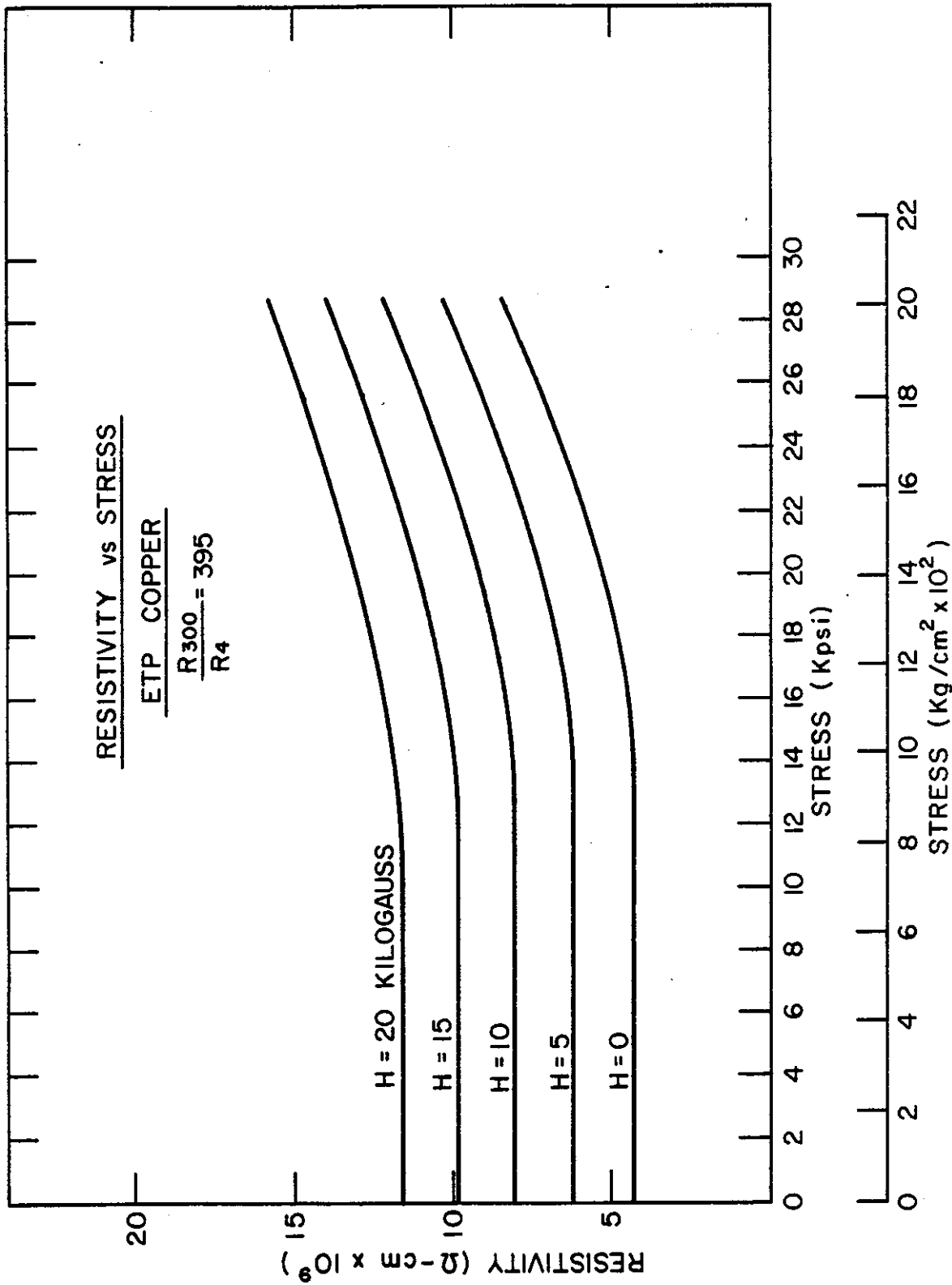


FIGURE 3

STRESS vs RESISTIVITY  
 BRUNNEN 1000, AVE  
 1966 CALIF. STATE

BY GTM. DATE 6/66

SUBJECT.....

SHEET NO.....OF.....

CHKD. BY.....DATE.....

DEPT. OR PROJECT.....

JOB NO.....

YIELD VS RESISTIVITY RATIO  
FOR SMALL C.S.A.

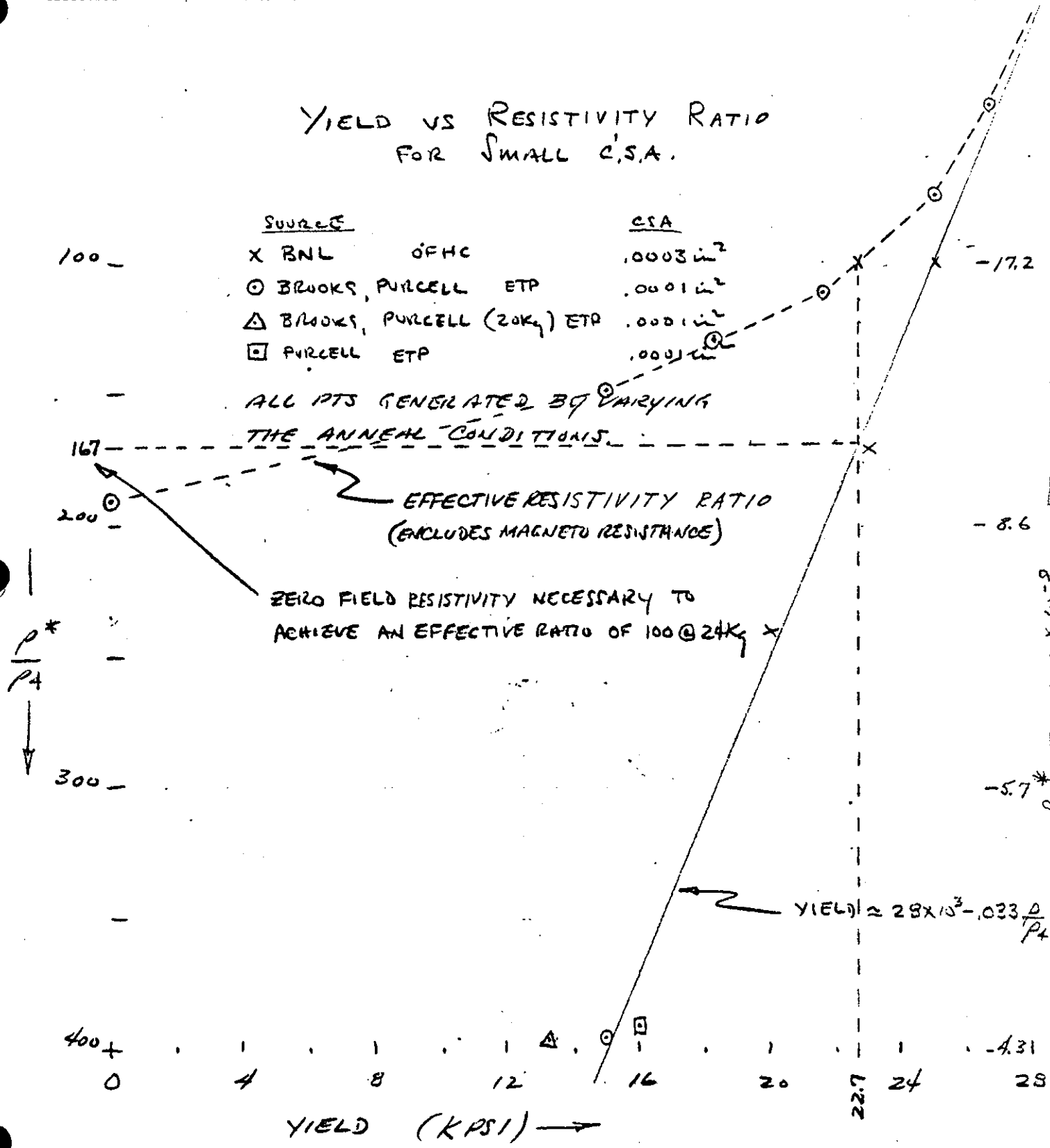
SOURCE	CSA
X BNL OFHC	.0003 in <sup>2</sup>
⊙ BROOKS, PURCELL ETP	.0001 in <sup>2</sup>
△ BROOKS, PURCELL (20K <sub>g</sub> ) ETP	.0001 in <sup>2</sup>
□ PURCELL ETP	.0001 in <sup>2</sup>

ALL PTS GENERATED BY VARYING THE ANNEAL CONDITIONS.

EFFECTIVE RESISTIVITY RATIO (INCLUDES MAGNETO RESISTANCE)

ZERO FIELD RESISTIVITY NECESSARY TO ACHIEVE AN EFFECTIVE RATIO OF 100 @ 24K<sub>g</sub>

YIELD  $\approx 28 \times 10^3 - .033 \frac{\rho}{\rho_0}$



\* MAGNETO-R NOT INCLUDED ;  $\rho_M \approx 4 \times 10^{-10} \Omega\text{-cm/Kg}$   
i.e.  $\approx 8 \times 10^{-9}$  @ 20 K<sub>g</sub>

AVCO SUPERGENIC STRIP  
 9 Nb-Zr 10 mil wires  
 SHORT SAMPLE PERFORMANCE

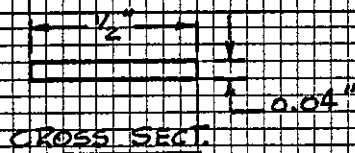
Oct. 1965

RECOVERY CURRENT - AMPERES

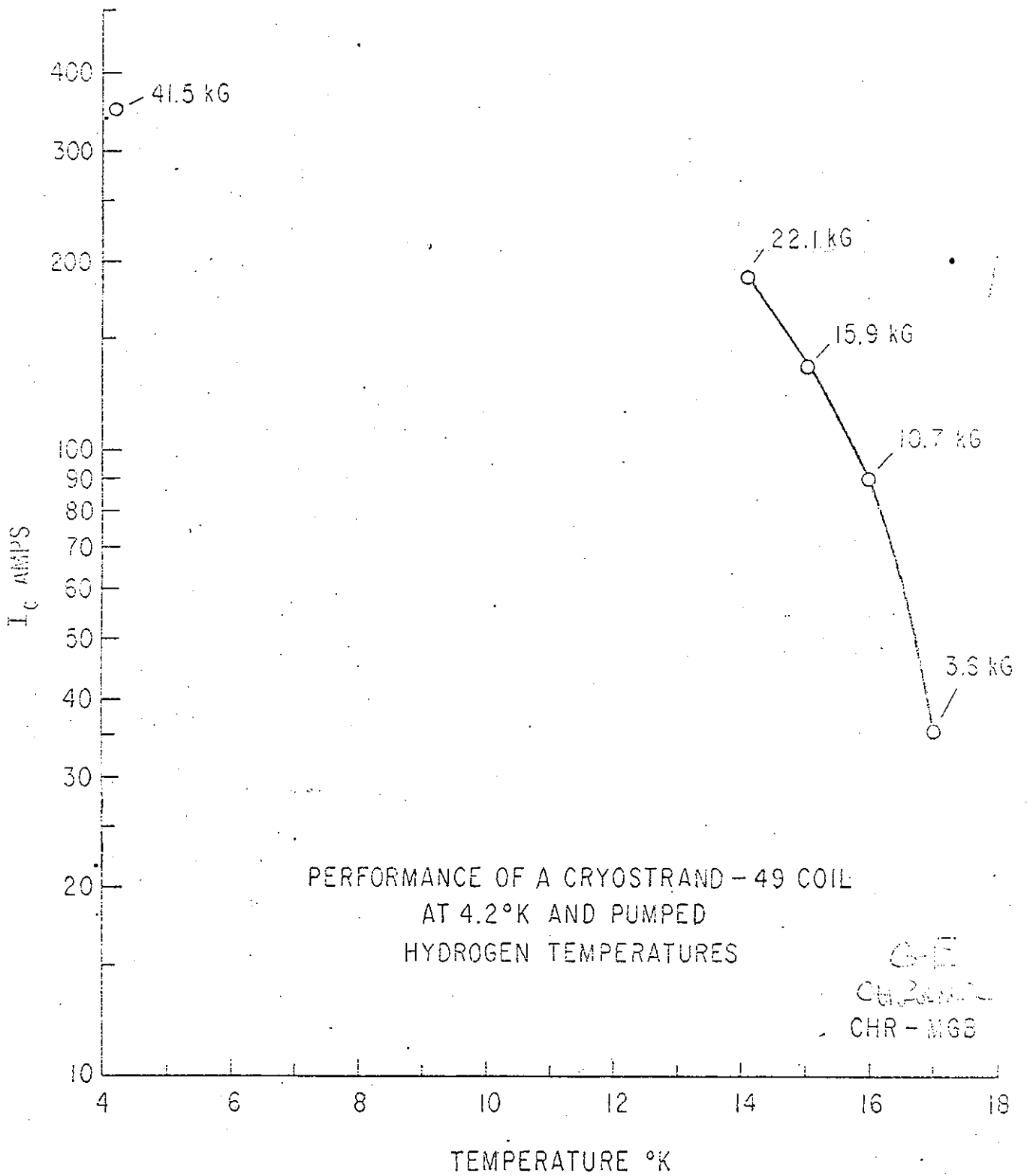
2000  
 1750  
 1500  
 1250  
 1000  
 750  
 500  
 250  
 0

KILOGAUSS

4.2°K  
 GUARANTEED  
 PERFORMANCE: +30%  
 -10%

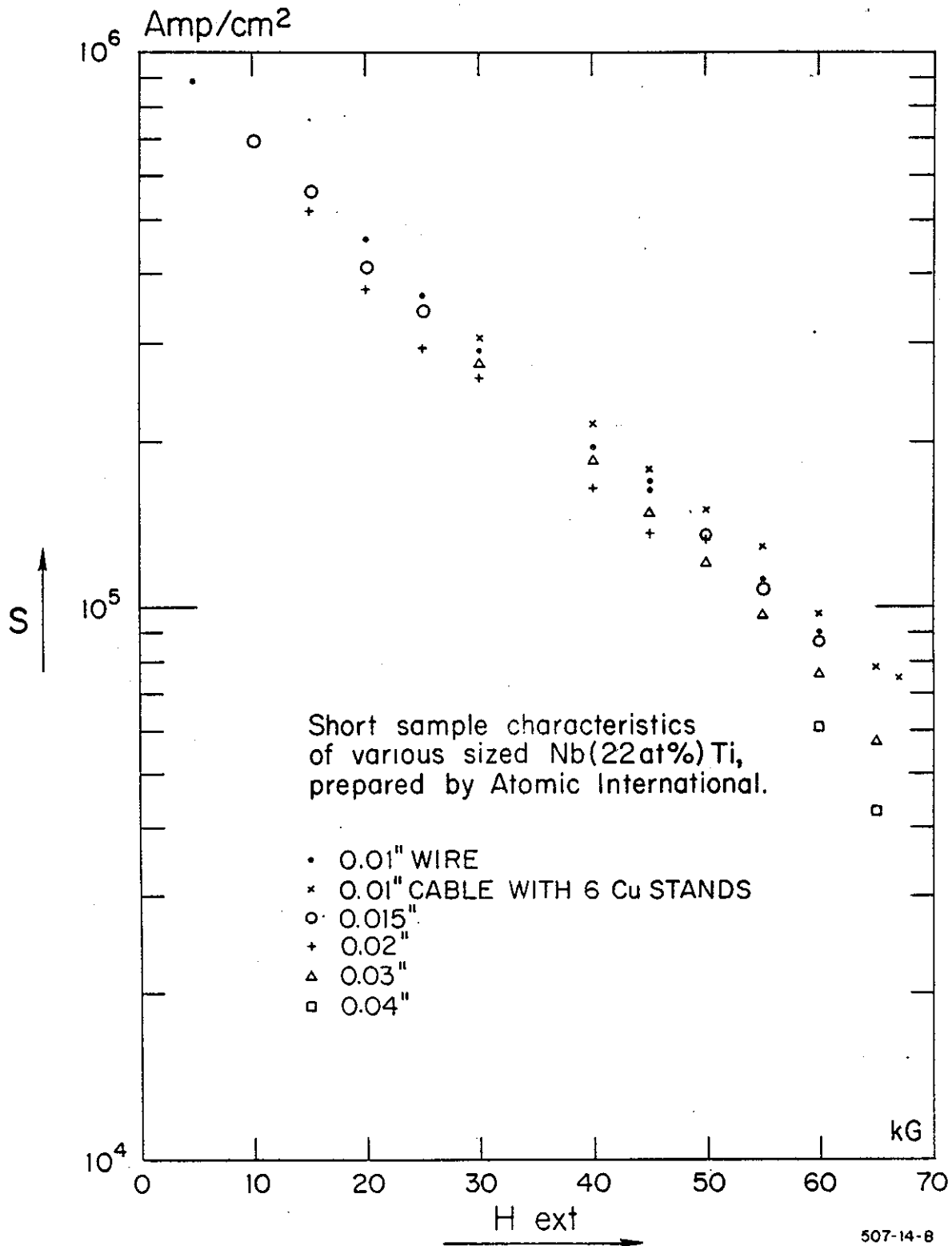


228 10/11/65



Dr. C. H. Rosner, The General  
Electric Co.

4.2



Brechna, H., A High Field 1.3m Superconducting Split  
 Coil Magnet with Forced Liquid Helium Cooling,  
 Stanford Linear Accelerator Center, SLAC-PUB-182,  
 April 1966

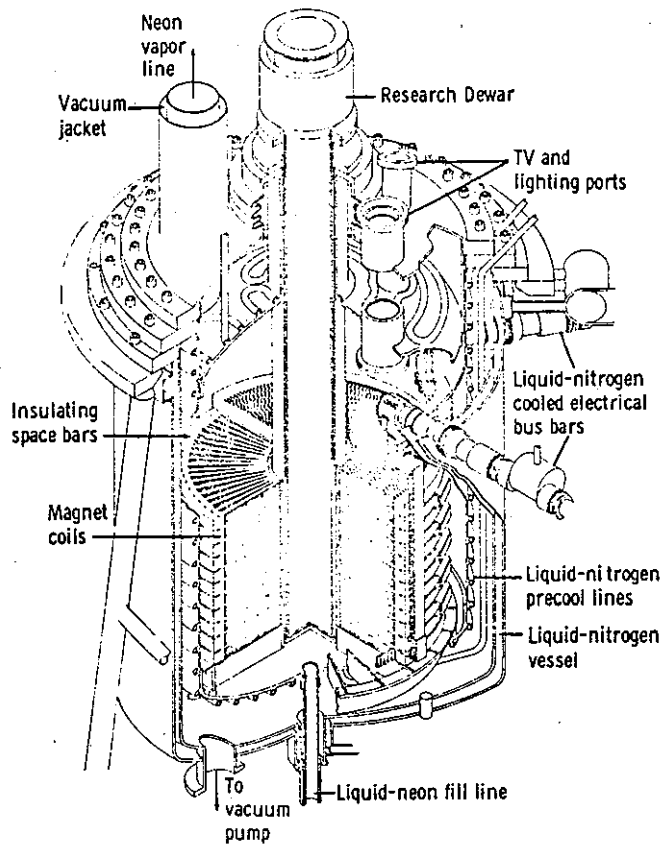


Fig. 1. - 12-in. Coils in magnet vessel.

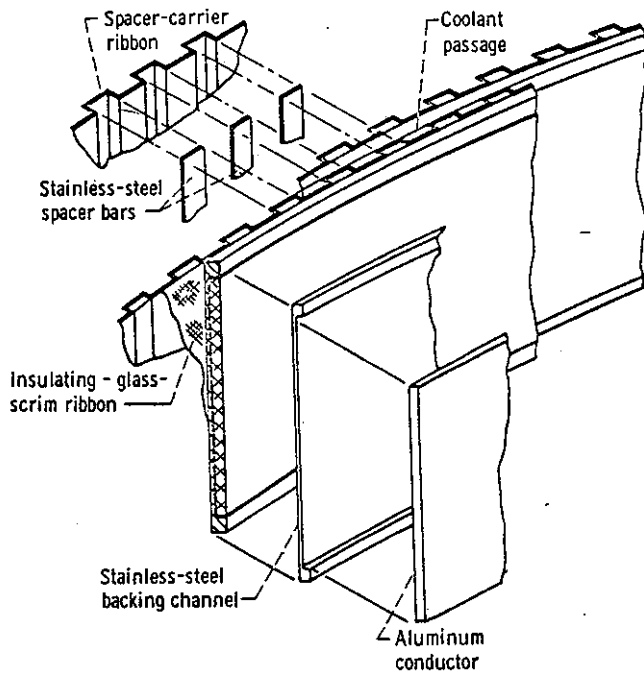


Fig. 2. - Details of coil construction.



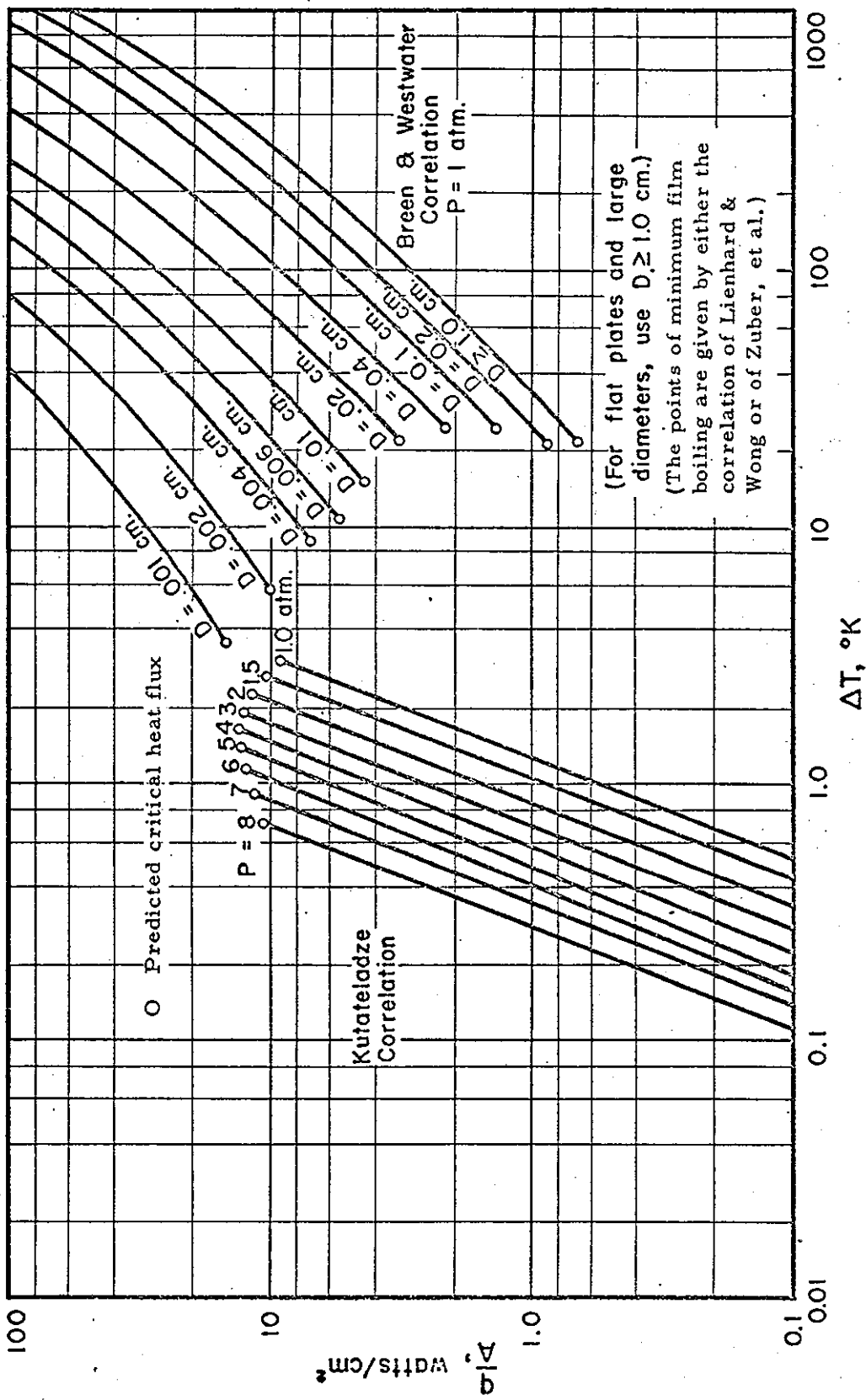


FIGURE 2.7  
Predictive Nucleate and Film Pool Boiling Correlations for Hydrogen

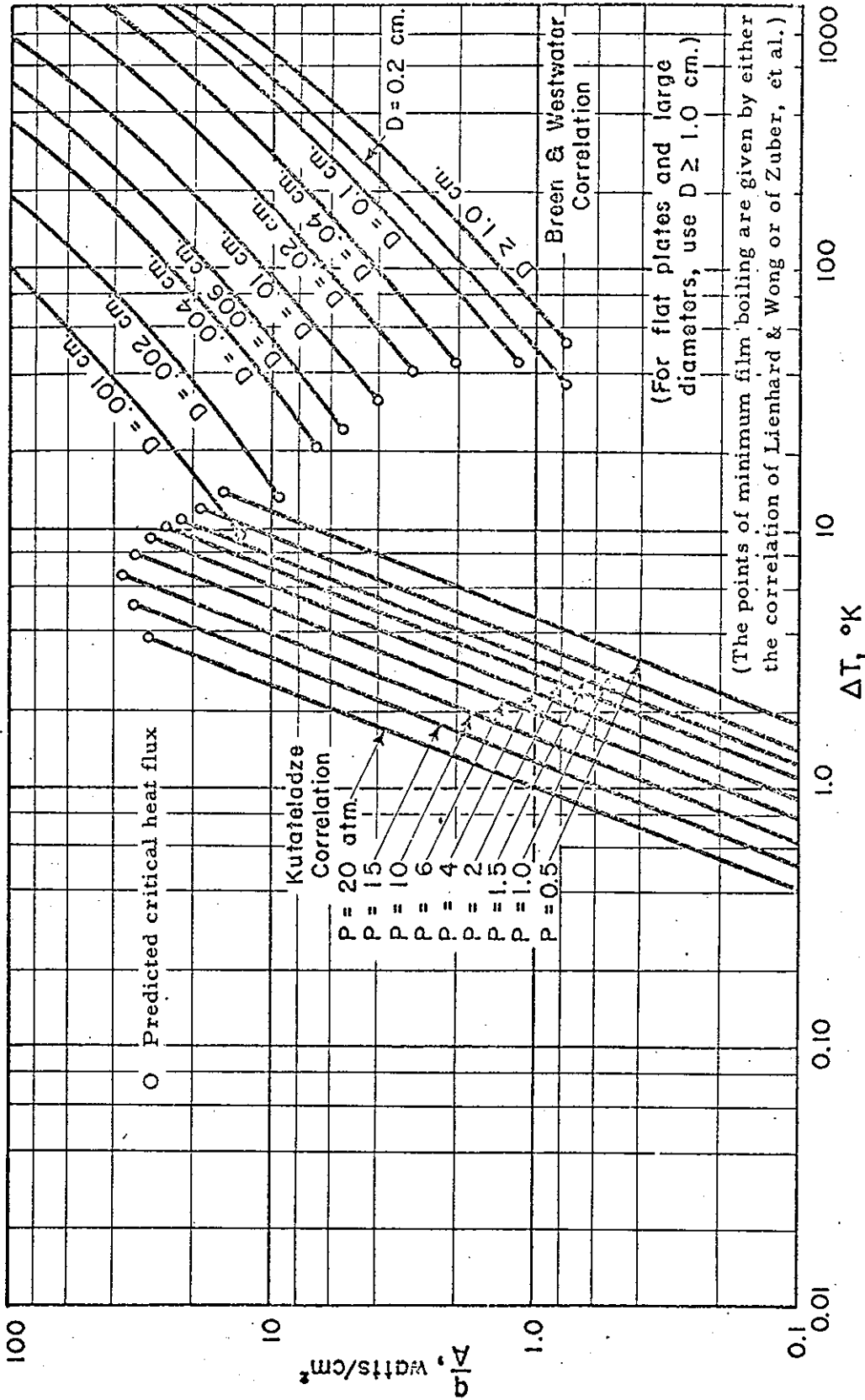


FIGURE 2.6  
 Predictive Nucleate and Film Pool Boiling Correlations for Nitrogen

Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. Note No. 317, Sept. 1965.

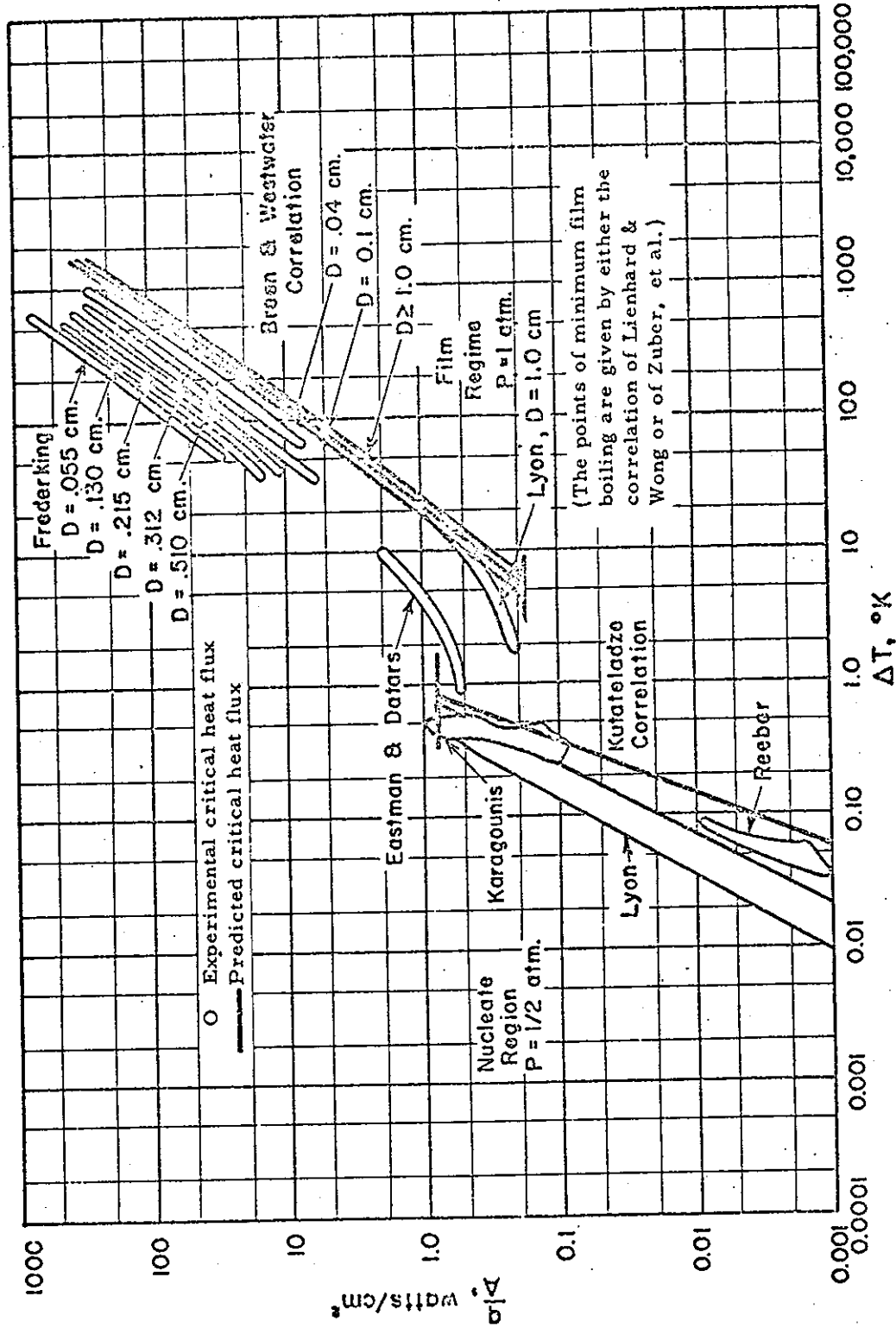


FIGURE 2.4

Experimental Nucleate and Film Pool Boiling of Helium Compared with the Predictive Correlation of Kutateladze and Breen and Westwater

Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. Note No. 317, Sept. 1965.

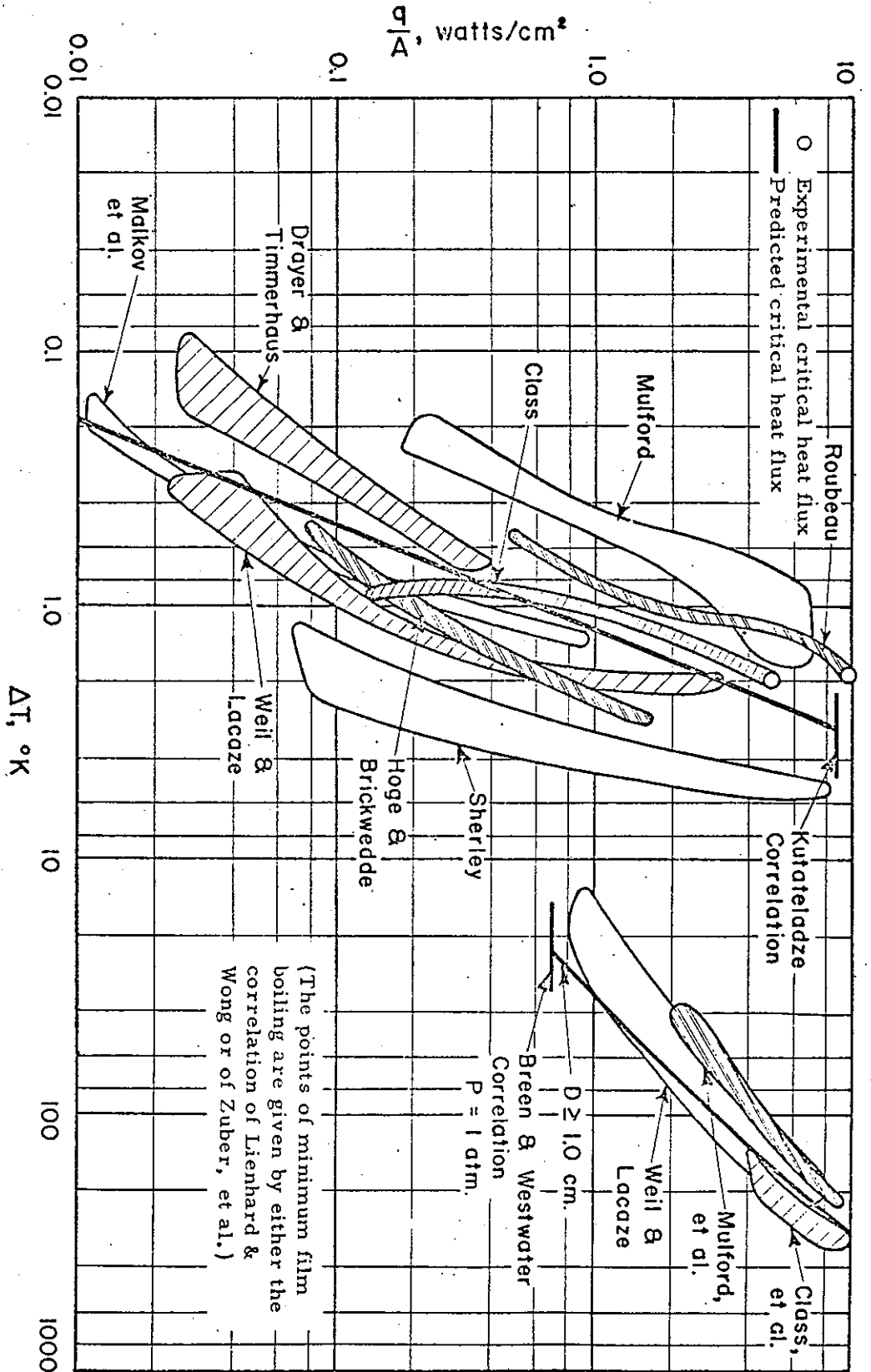


FIGURE 2.3  
 Experimental Nucleate and Film Pool Boiling of Hydrogen at One Atmos-  
 phere Compared with the Predictive Correlations of Kutateladze and  
 Breen and Westwater

Brentari, E.G., et al., Boiling Heat Transfer For  
 Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech.  
 Note No. 317, Sept. 1965.

TEMPERATURE, °R

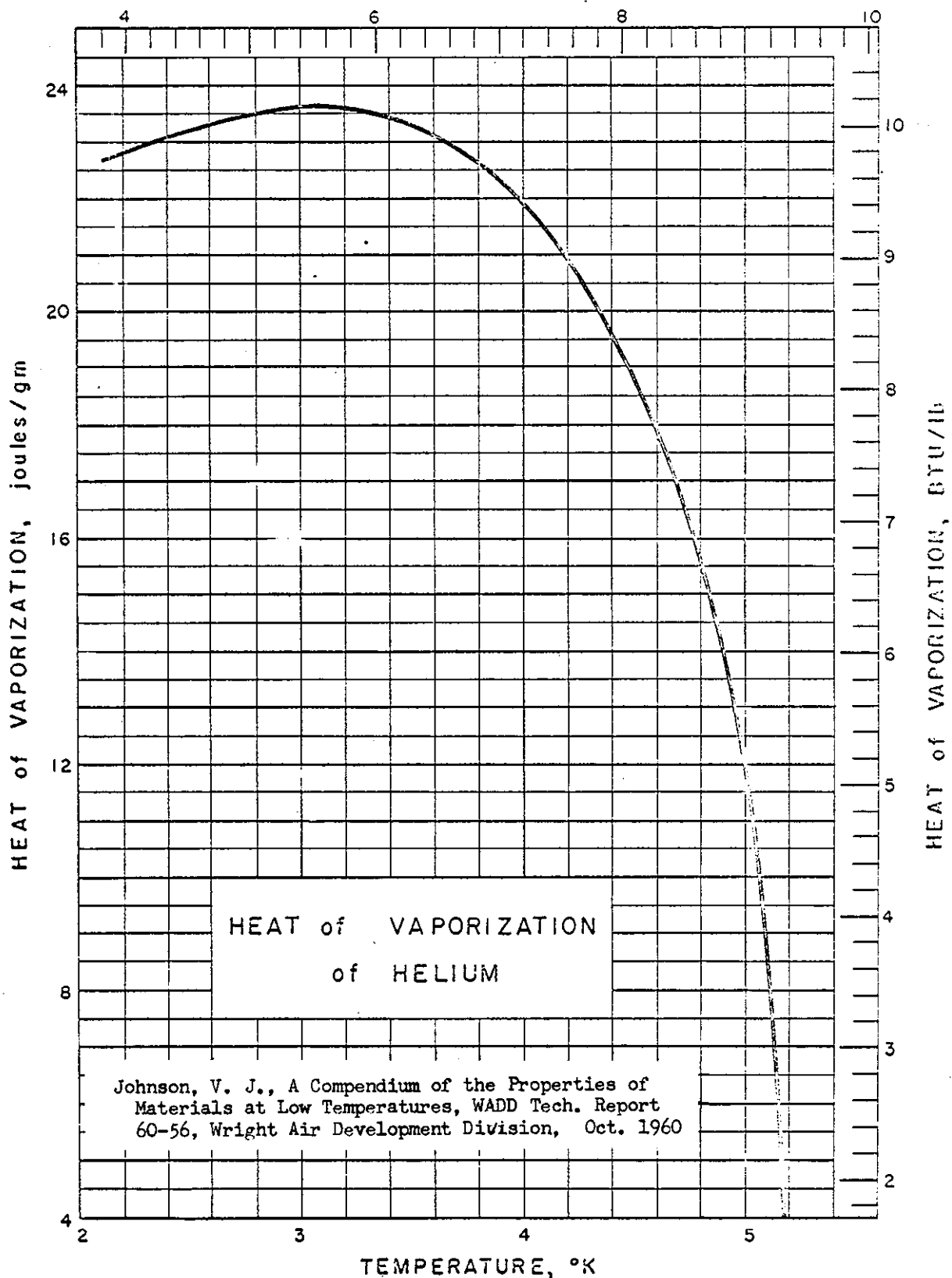
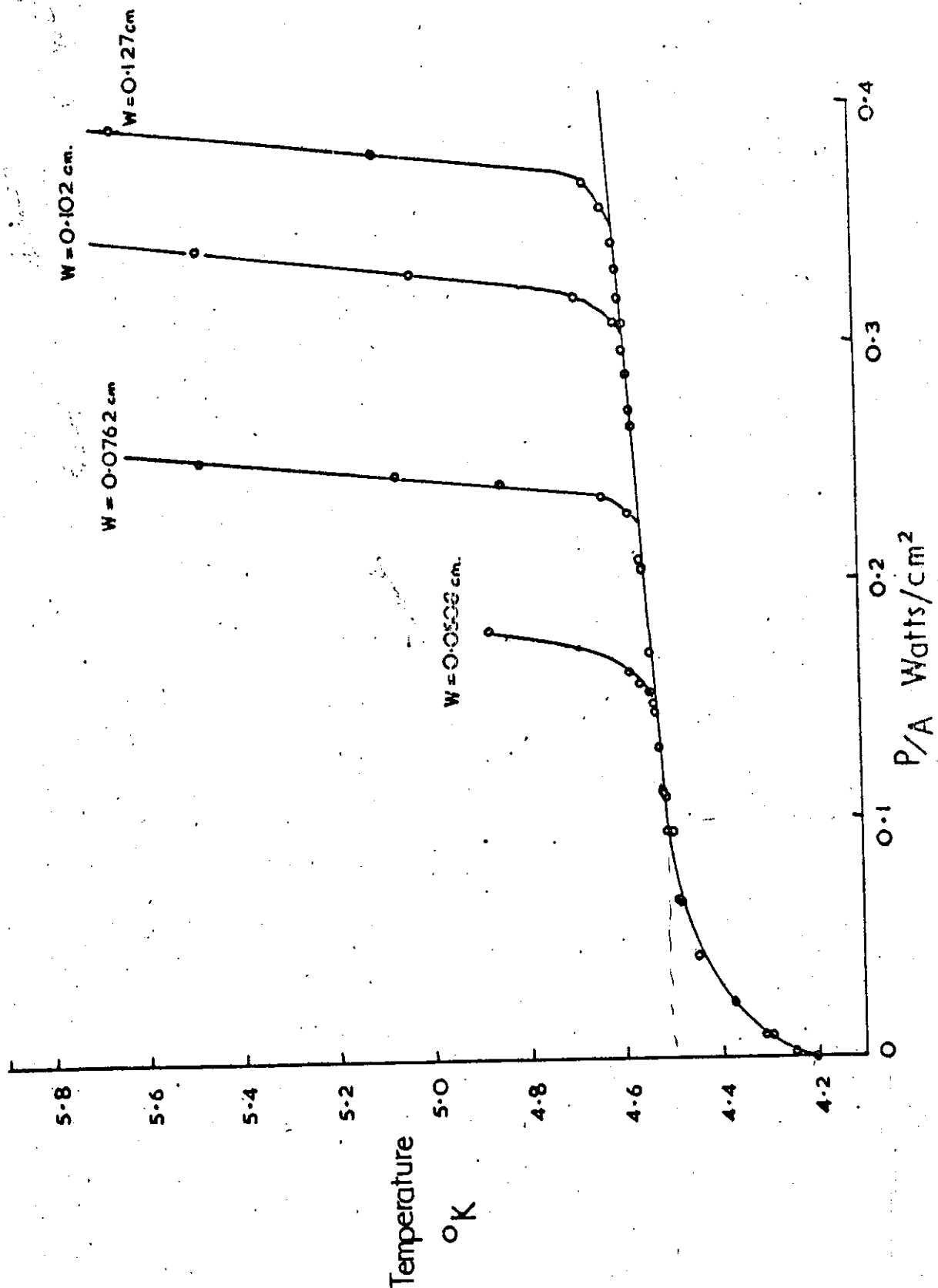


Fig. 2 Power/cm<sup>2</sup> vs. Temperature.

*W = w · D · ρ · C<sub>p</sub>*



Wilson, M. N., Heat Transfer to Boiling Liquid Helium in Narrow Vertical Channels, Rutherford Lab., Berks., England, IIR Commission 1 Meeting, June 64

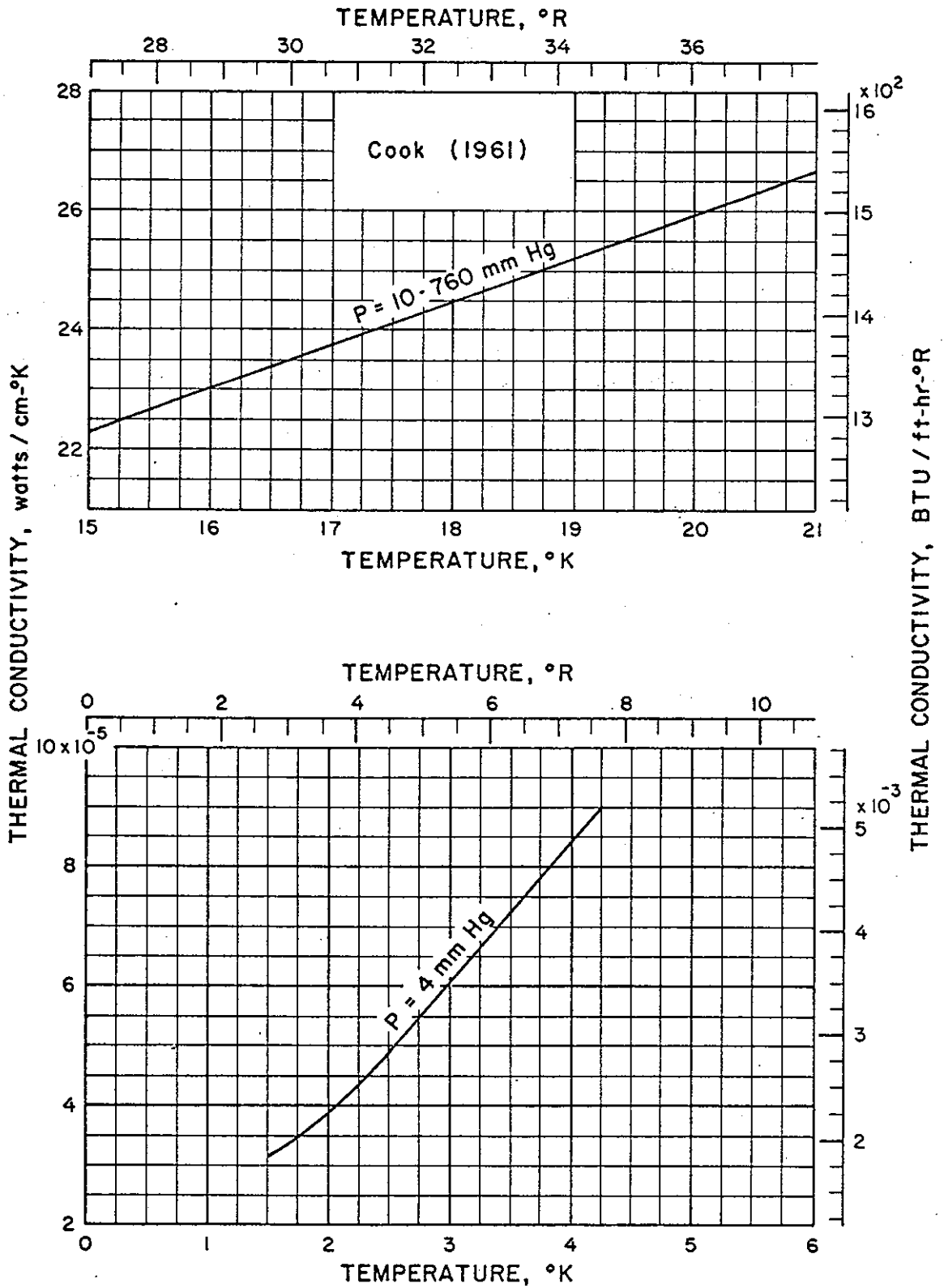


Figure 6.29 Thermal conductivity of gaseous helium from  $1.5^{\circ}\text{K}$  to  $21^{\circ}\text{K}$

Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. Note No. 317, Sept. 1965.

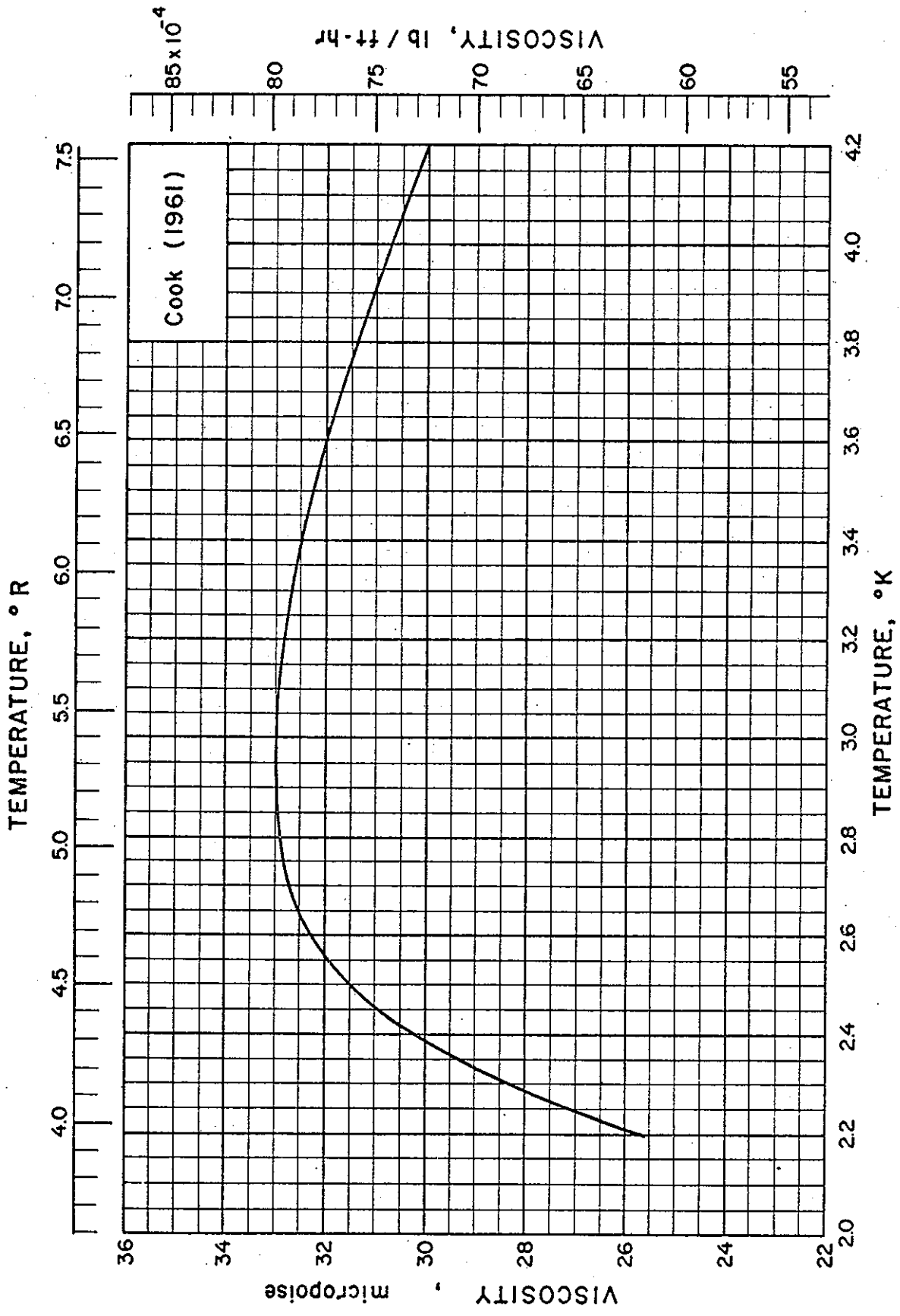
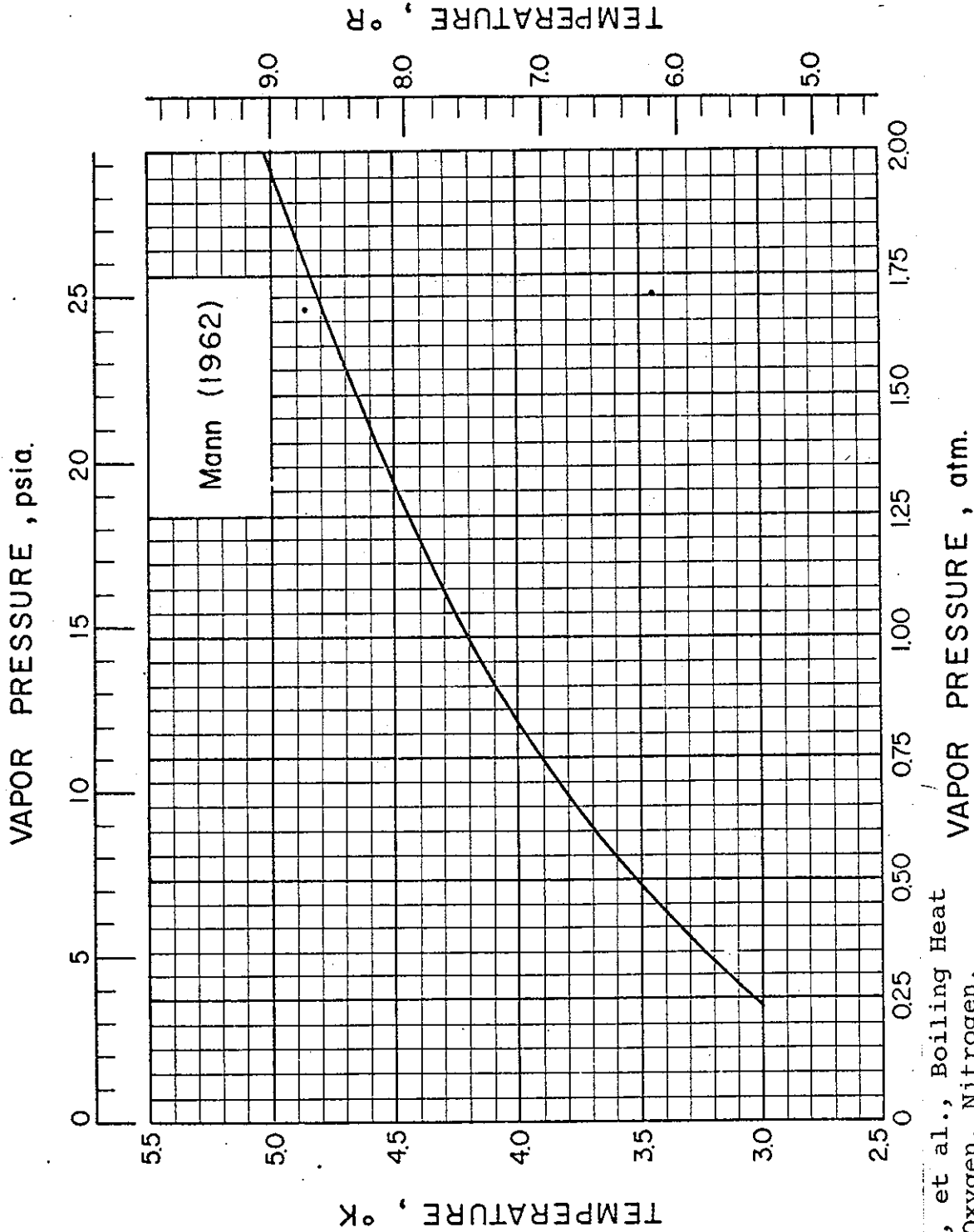


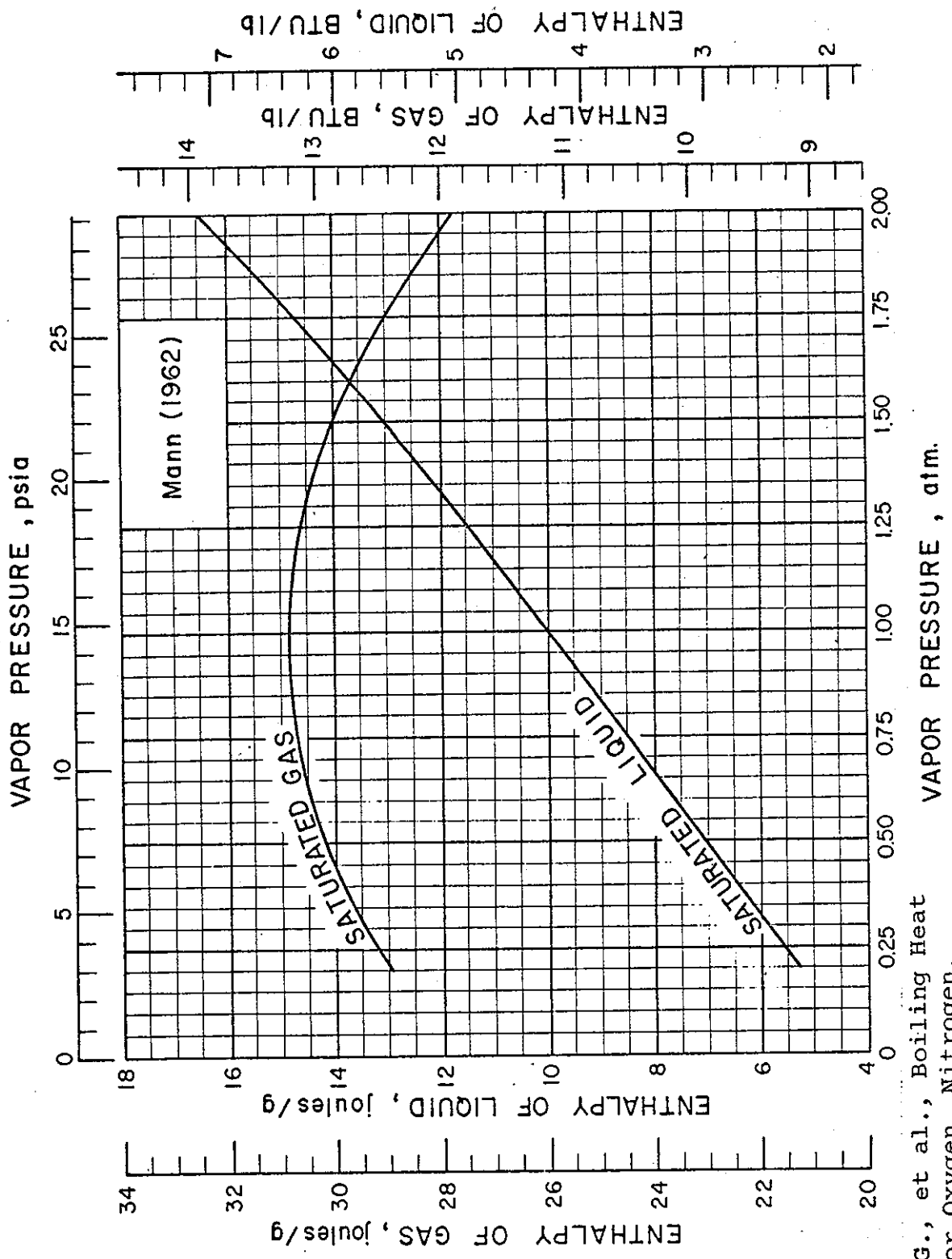
Figure 6.36 Viscosity of saturated liquid helium I  
 Brentari, E.G., et al., Boiling Heat Transfer For  
 Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech.  
 Note No. 317, Sept. 1965





Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen and Helium, NBS Tech. Note No. 317, Sept. 1965.

Vapor Pressure of helium-4



Brentari, E.G., et al., Boiling Heat Transfer for Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. No. 317, Sept. 1965. Figure 6.1 Enthalpy of saturated gaseous and saturated liquid helium

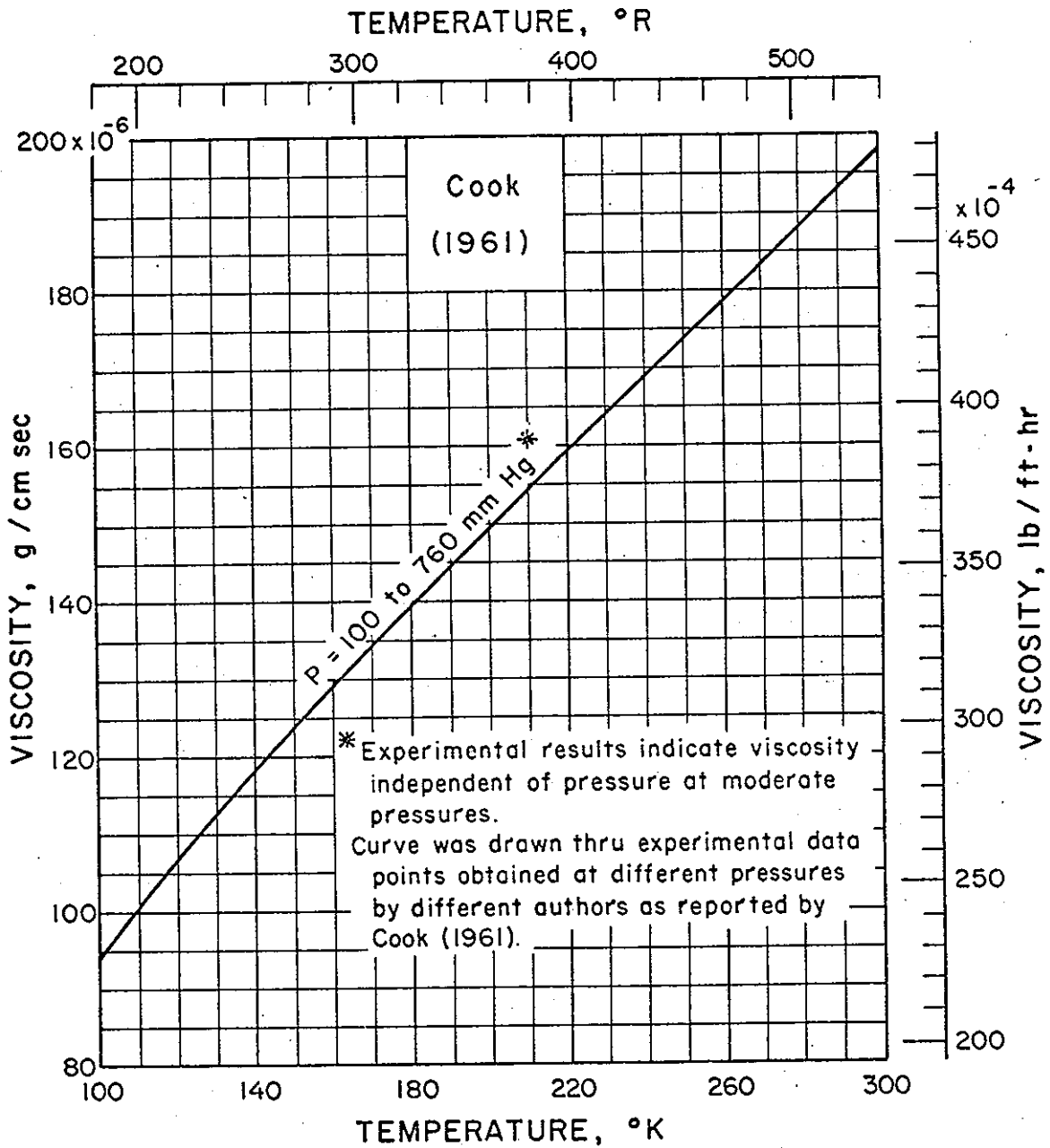


Figure 6.38 Viscosity of gaseous helium from 100° to 300°K

Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. Note No. 317, Sept. 1965.

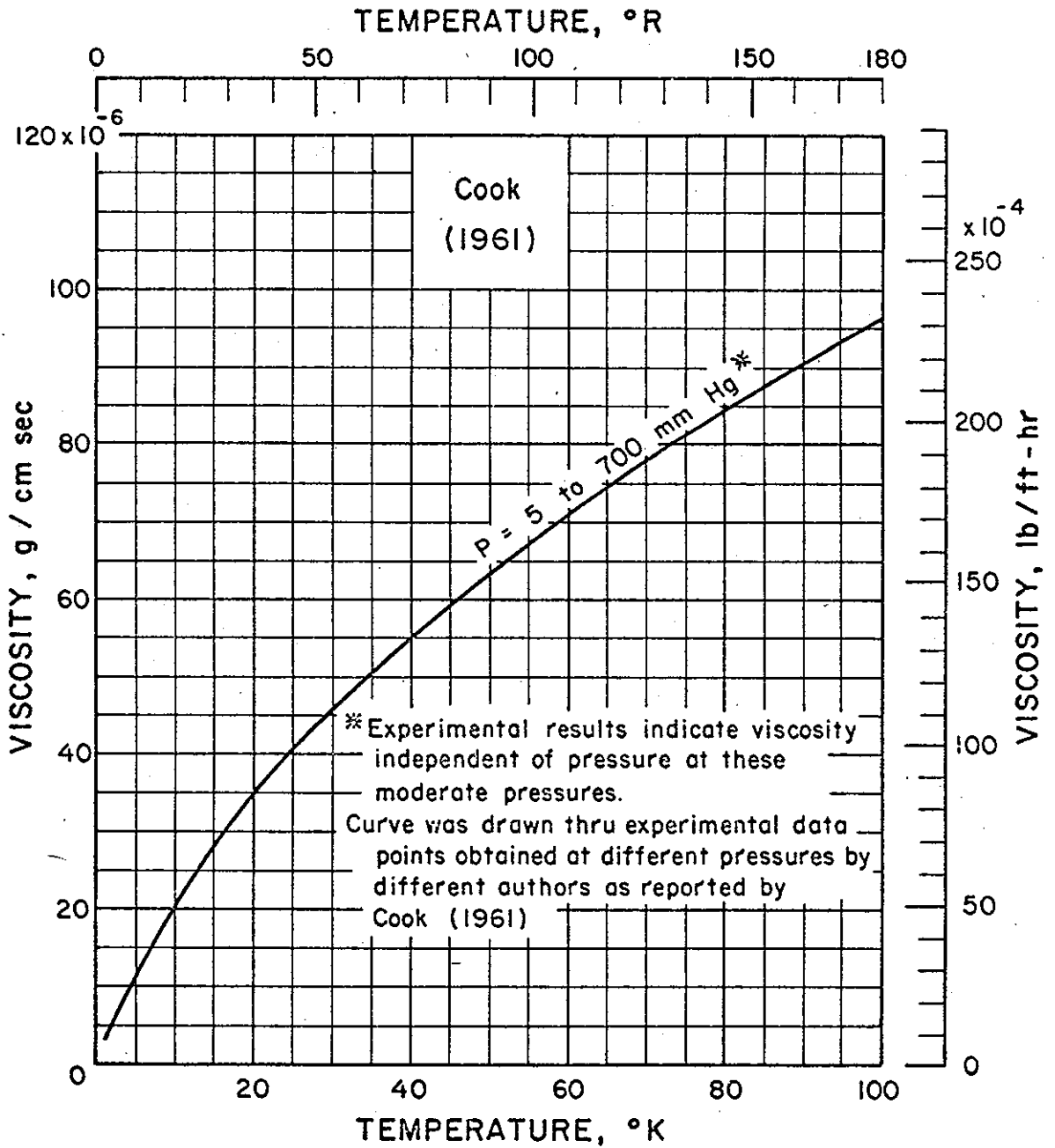


Figure 6.37 Viscosity of gaseous helium from 0° to 100°K .

Brentari, E.G., et al., Boiling Heat Transfer For Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech. Note No. 317, Sept. 1965.

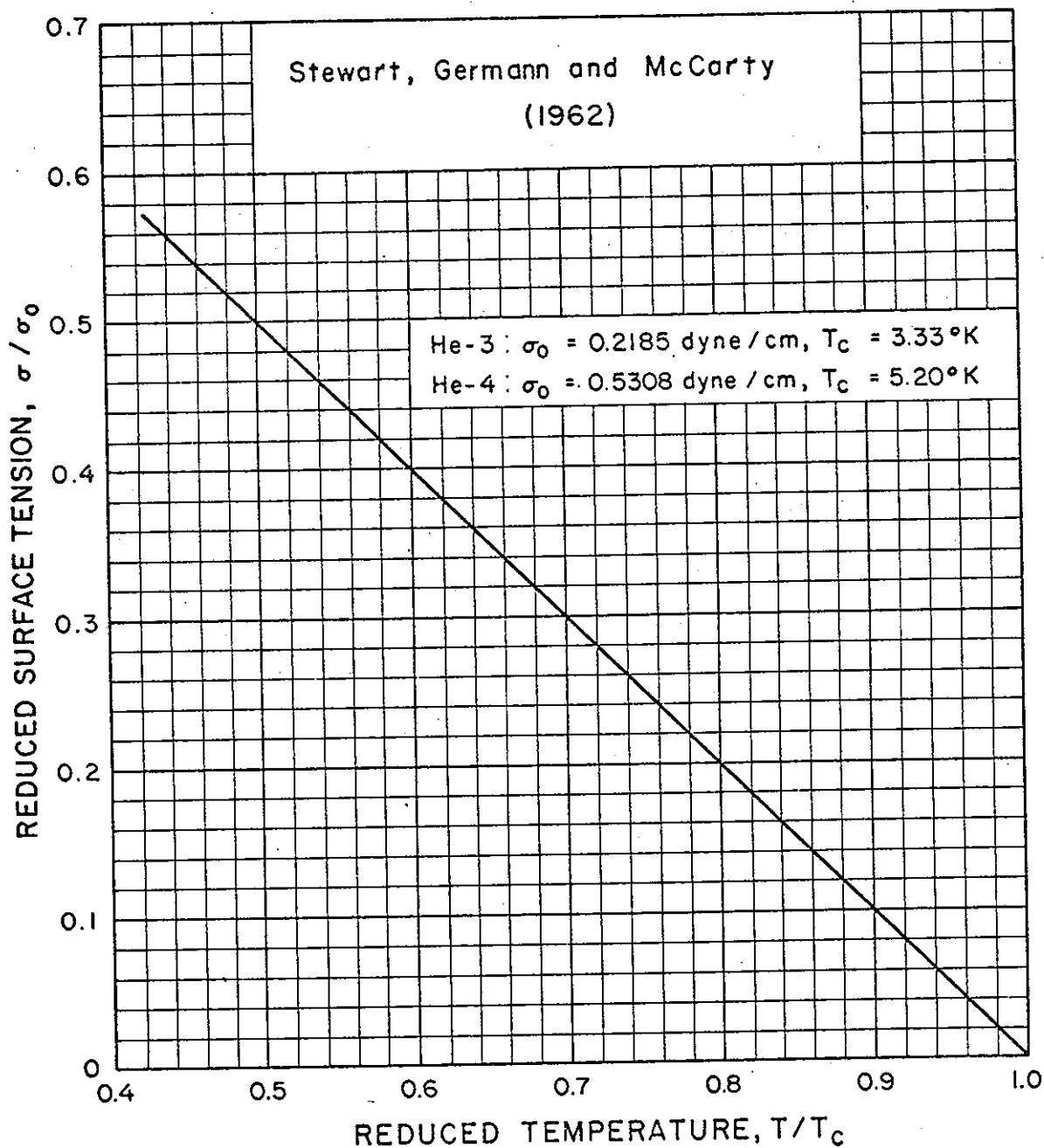
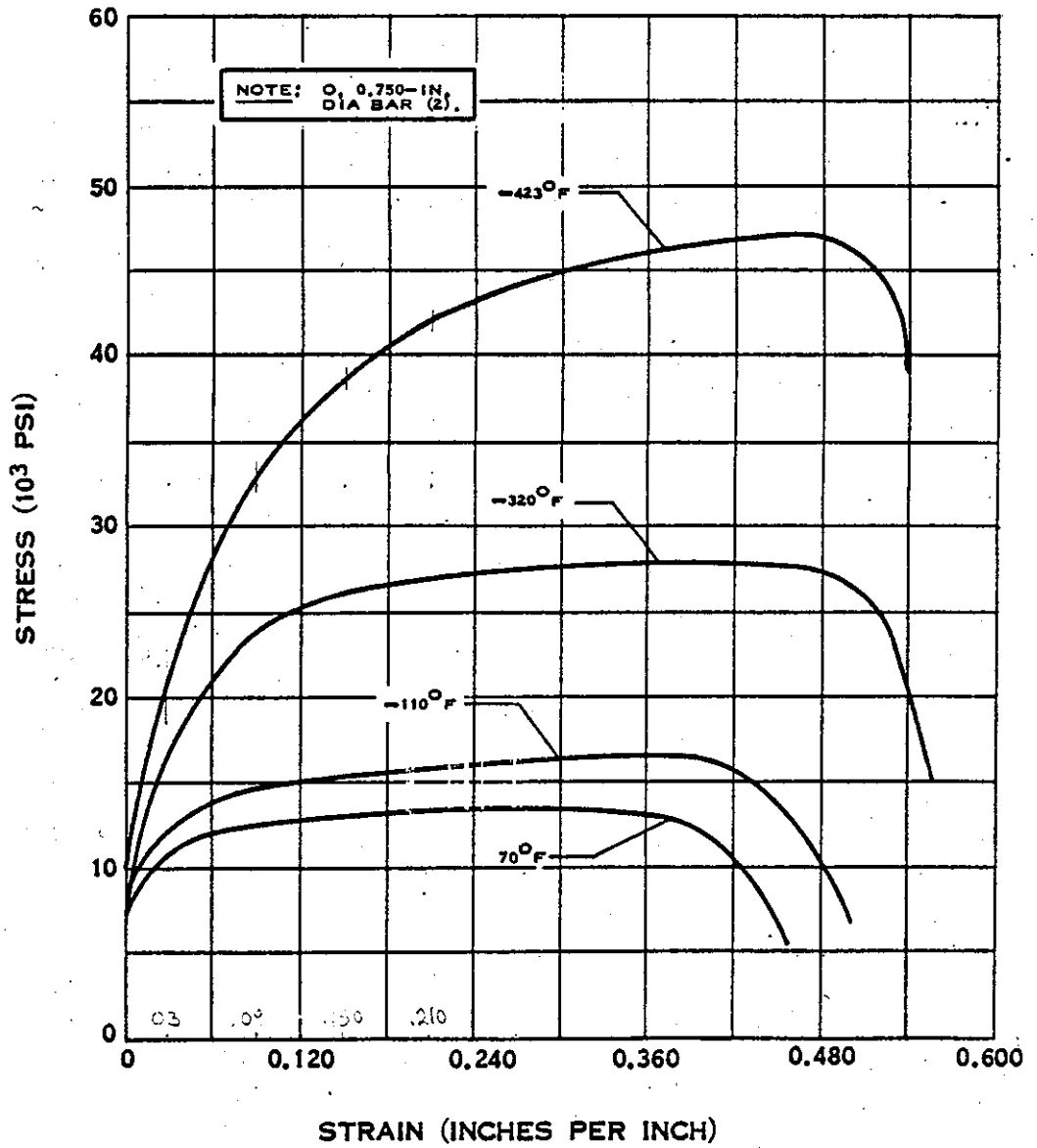


Figure 6.24 Surface tension for helium-3, and helium 4

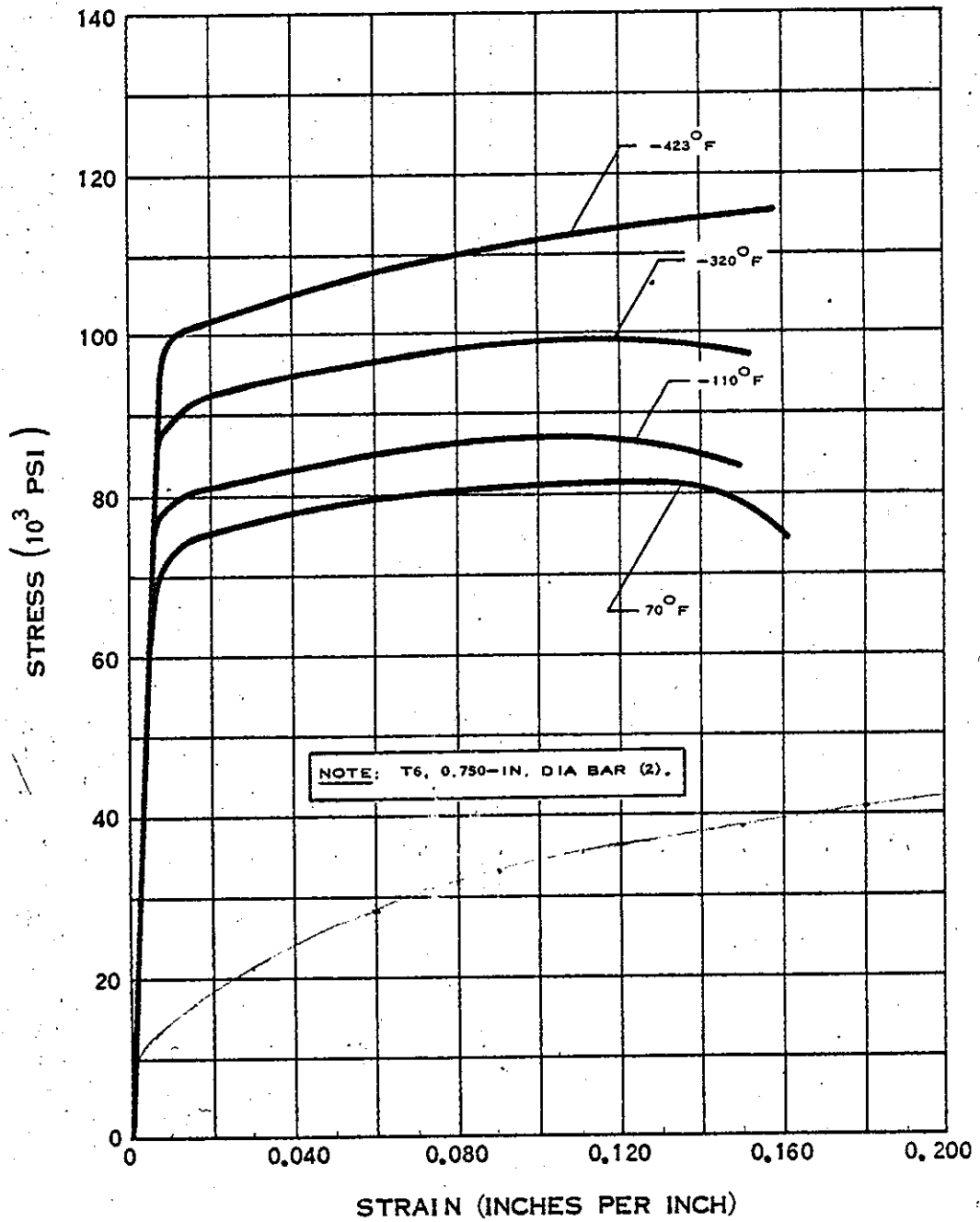
Brentari, E.G., et al., Boiling Heat Transfer For  
Oxygen, Nitrogen, Hydrogen, and Helium, NBS Tech.  
Note No. 317, Sept. 1965.

A.1.h



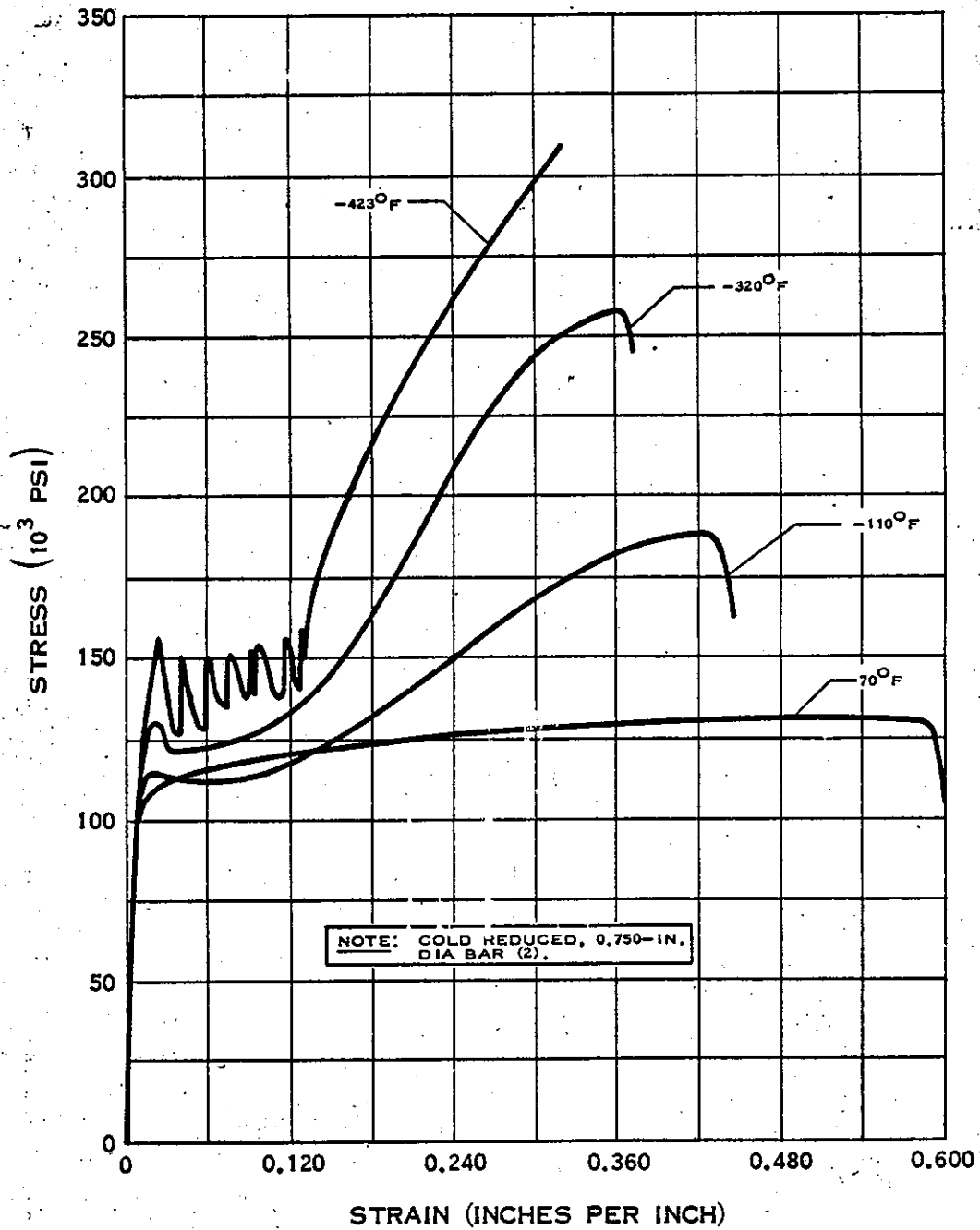
STRESS-STRAIN DIAGRAM FOR 1100 ALUMINUM

# A.16.h



**STRESS-STRAIN DIAGRAM FOR 7075 ALUMINUM**

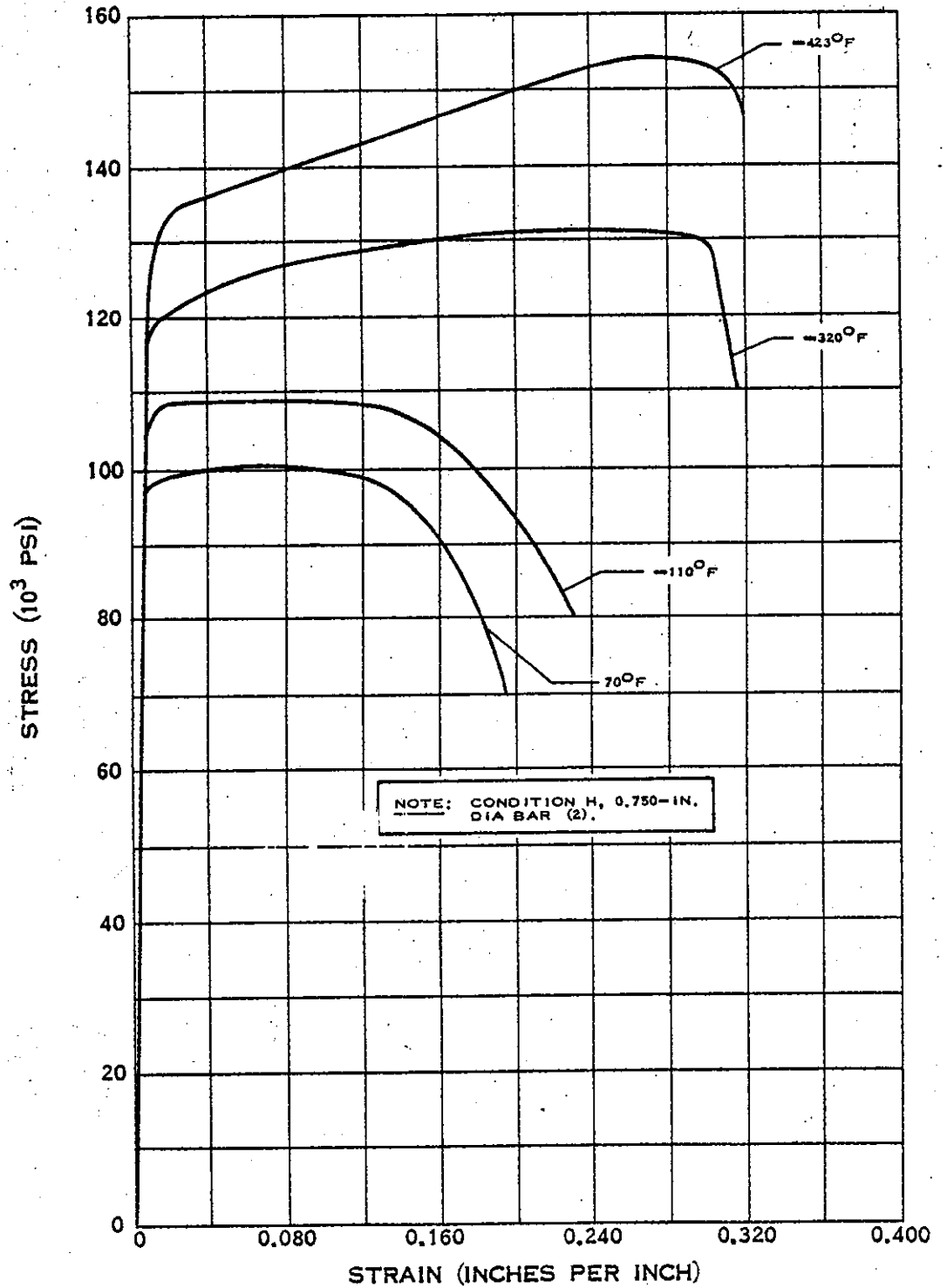
B.2.h



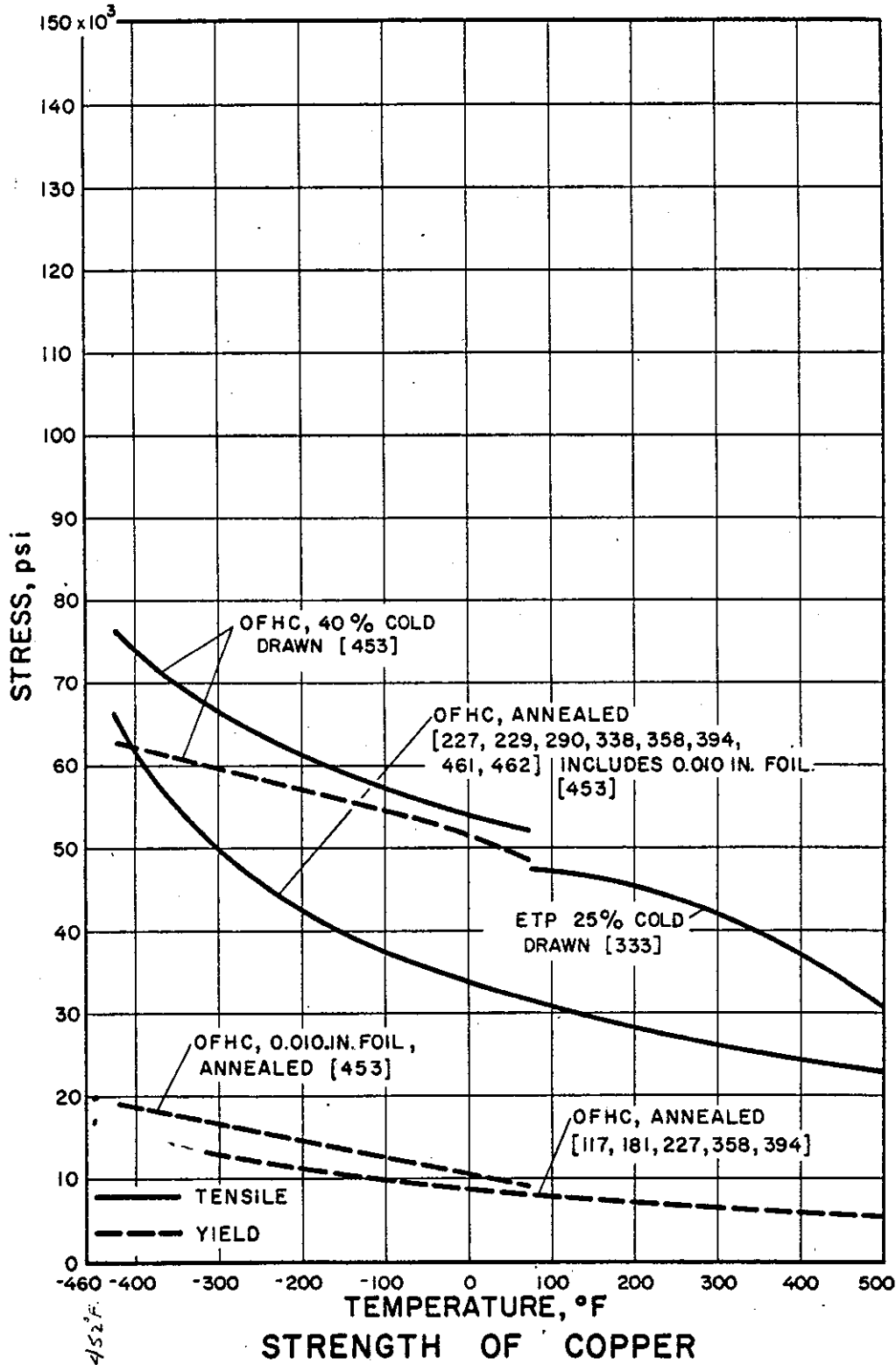
**STRESS-STRAIN DIAGRAM FOR 302 STAINLESS STEEL**



F.2.h-1

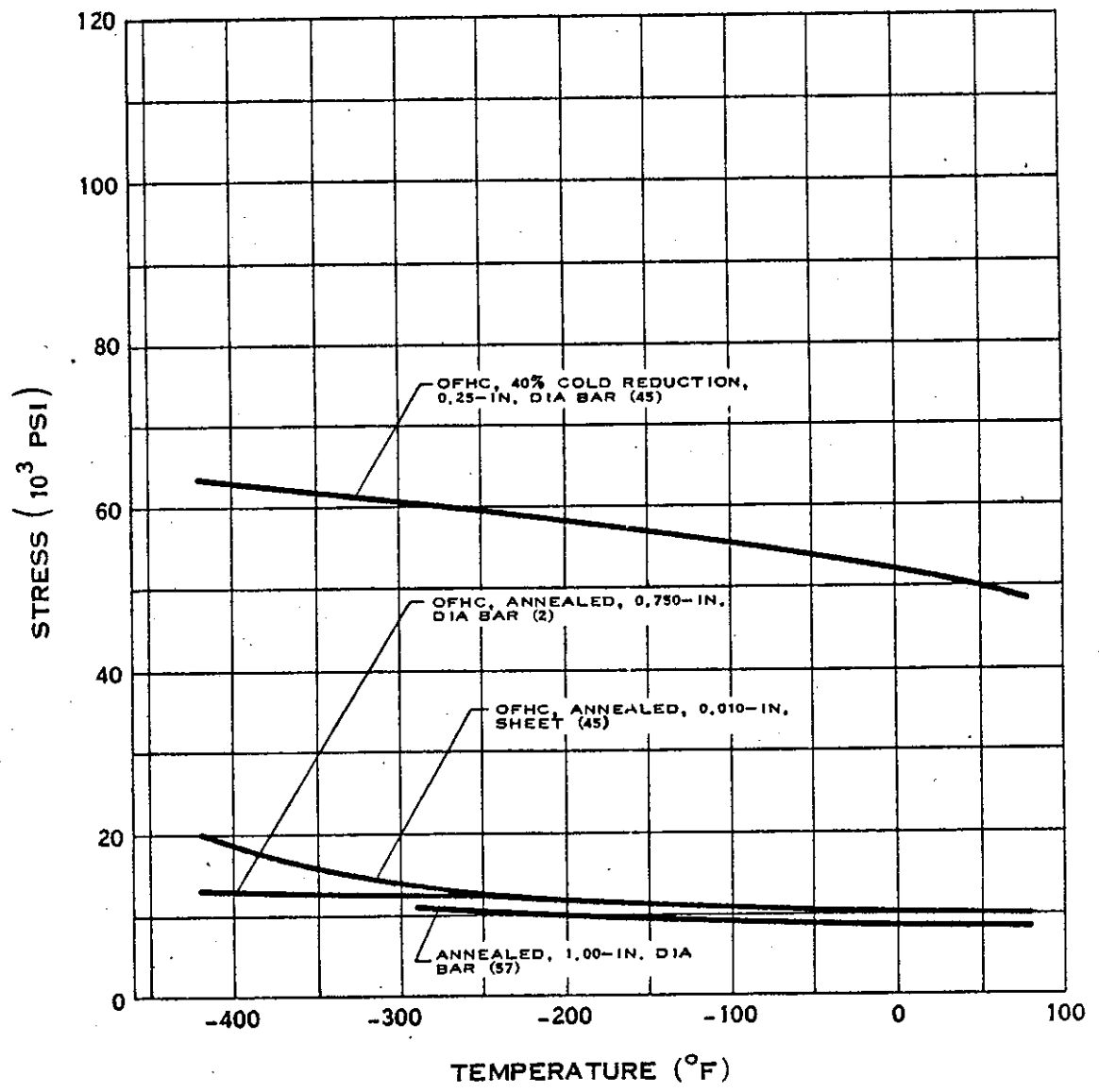


STRESS-STRAIN DIAGRAM FOR BERYLLIUM COPPER



4.2%  
452°F

F.1.a

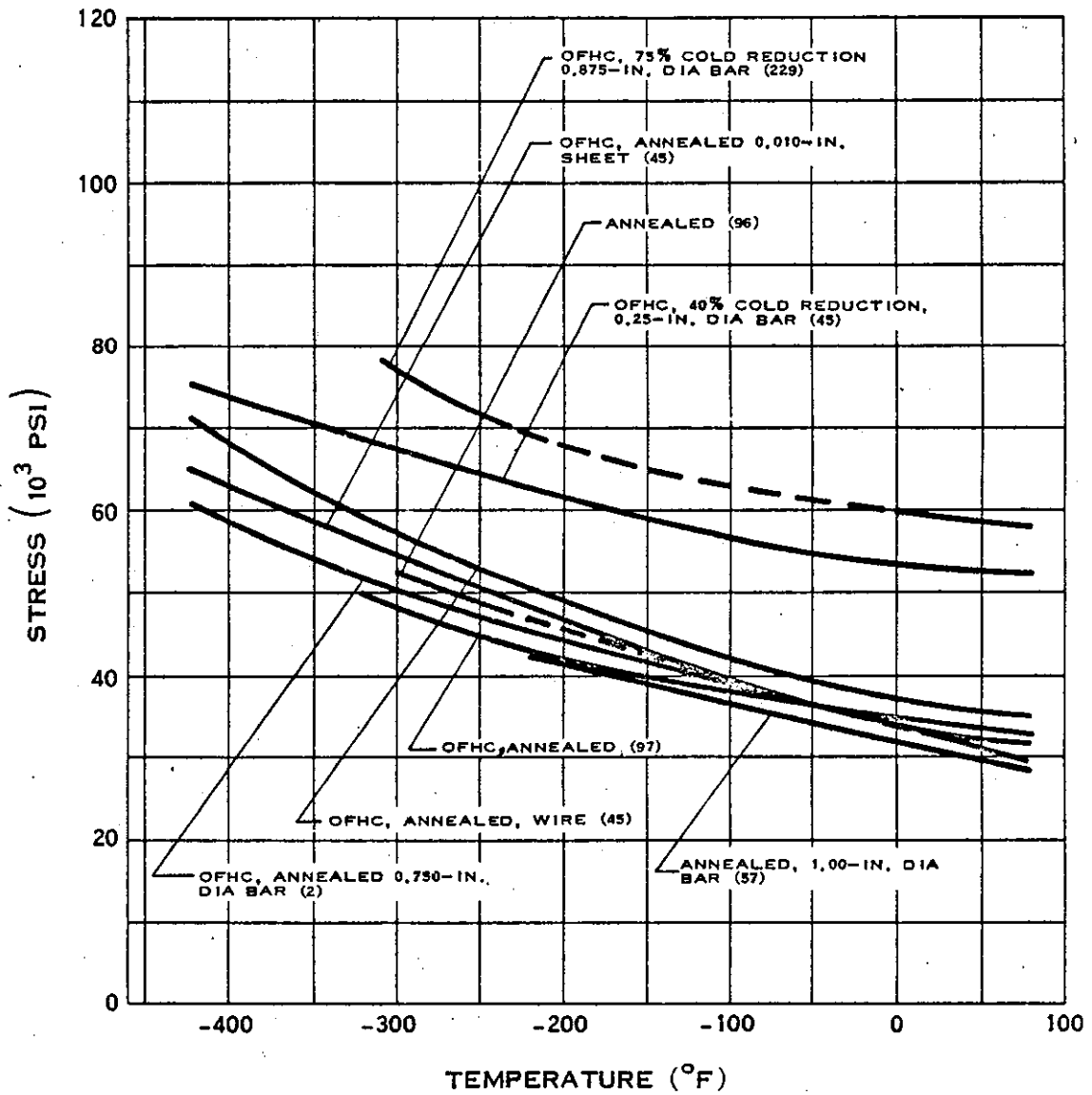


YIELD STRENGTH OF COPPER

(7-64)

Schwartzberg, et al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-2810  
AD609562, Air Force Materials Lab., Aug. 1964

# F.1.b

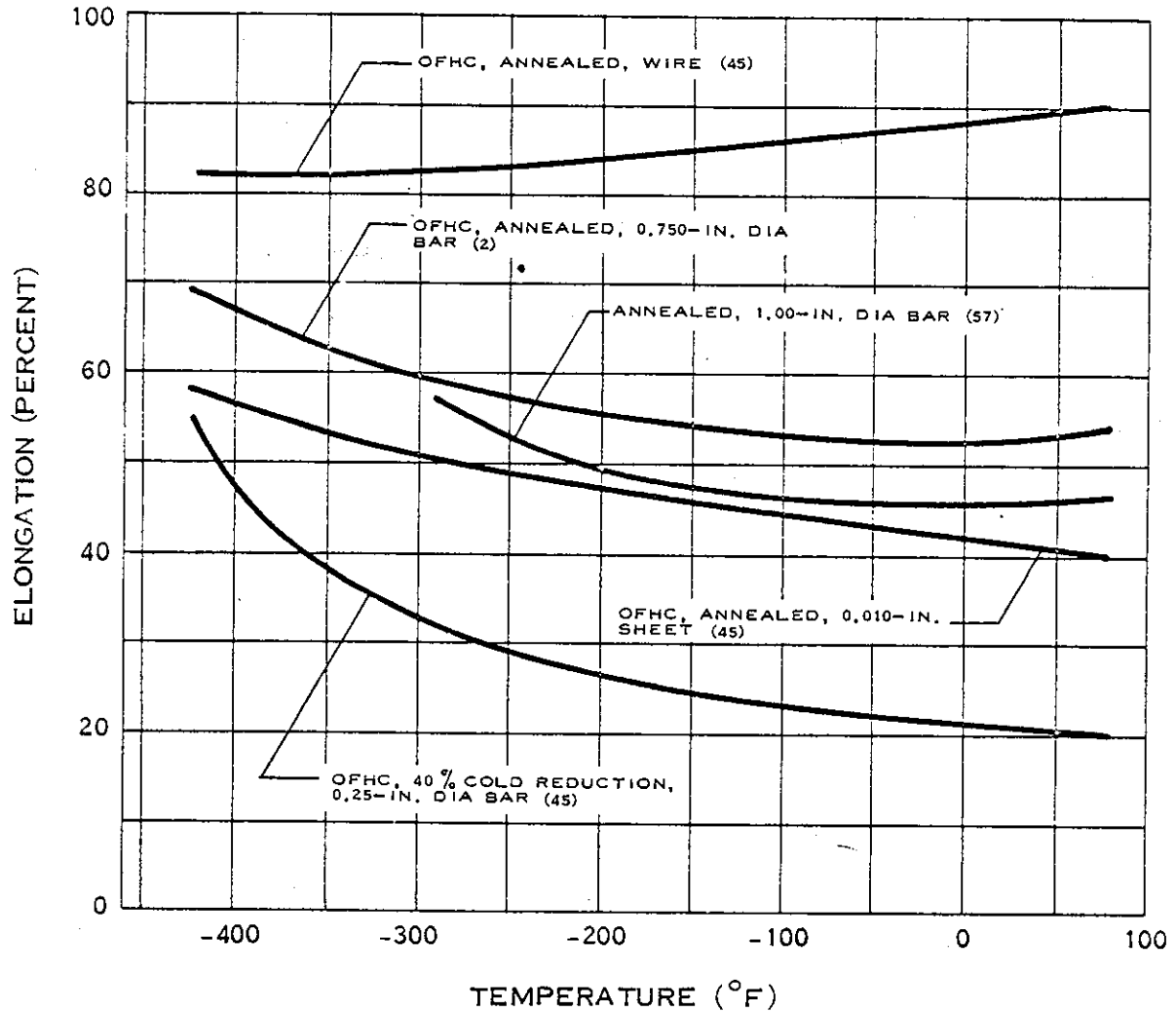


## TENSILE STRENGTH OF COPPER

(7-64)

Schwartzberg, et al., Cryogenic Material Data  
 Handbook, Technical Documentary Report ML-TDR-64-280  
 AD609562, Air Force Materials Lab., Aug. 1964

F.1.c

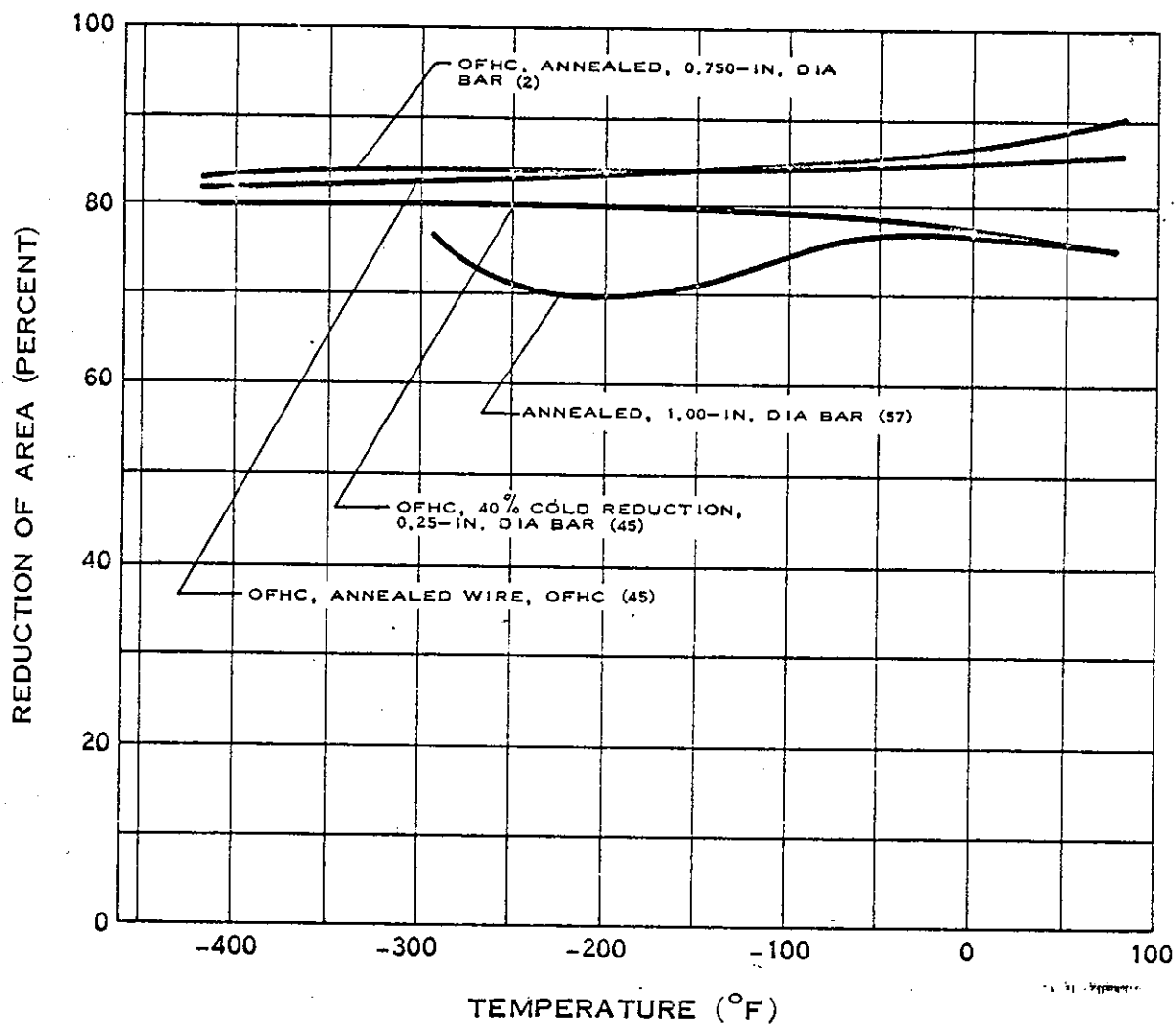


### ELONGATION OF COPPER

(7-64)

Schwartzberg, et al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964

F.1.d

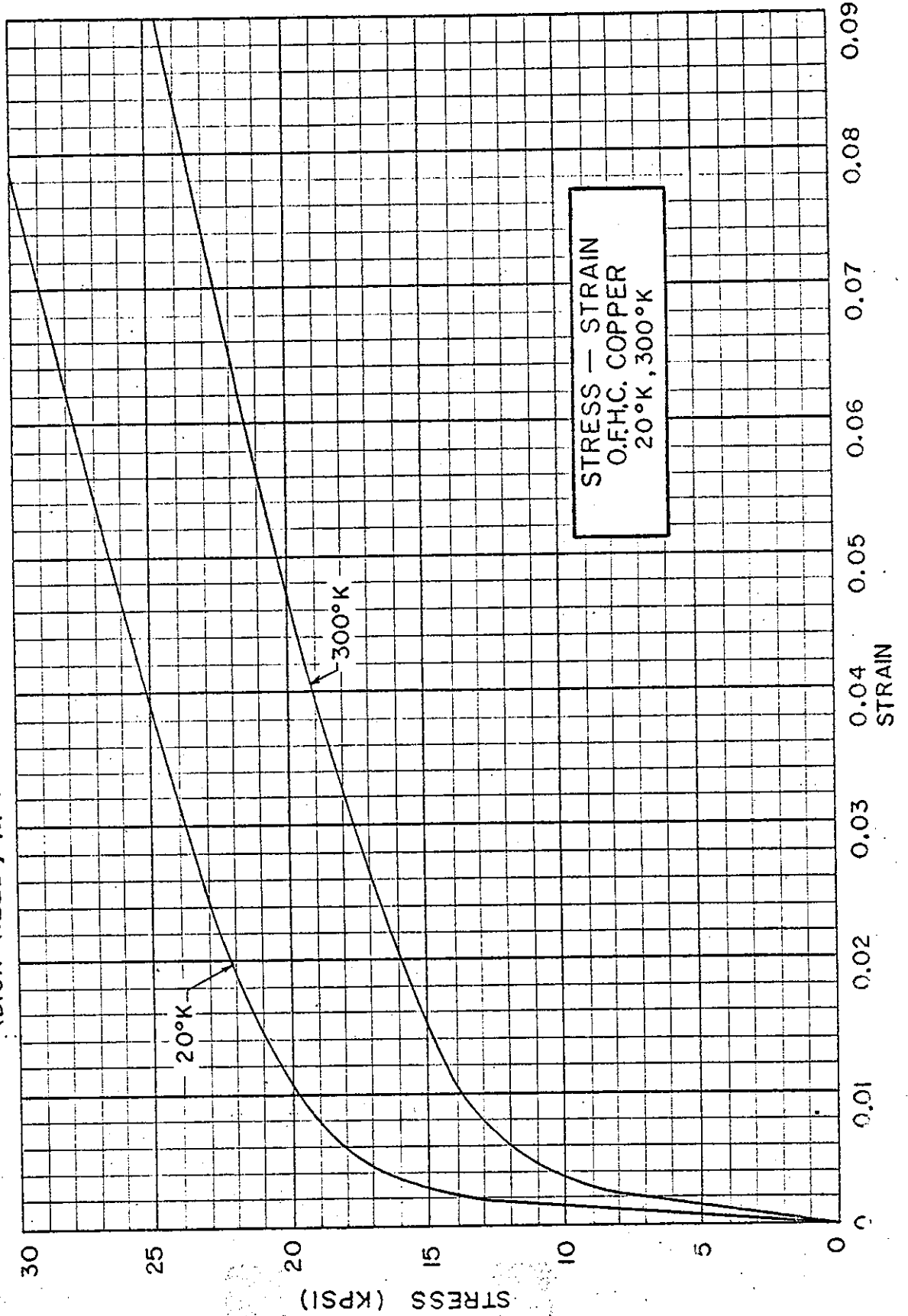


### REDUCTION OF AREA OF COPPER

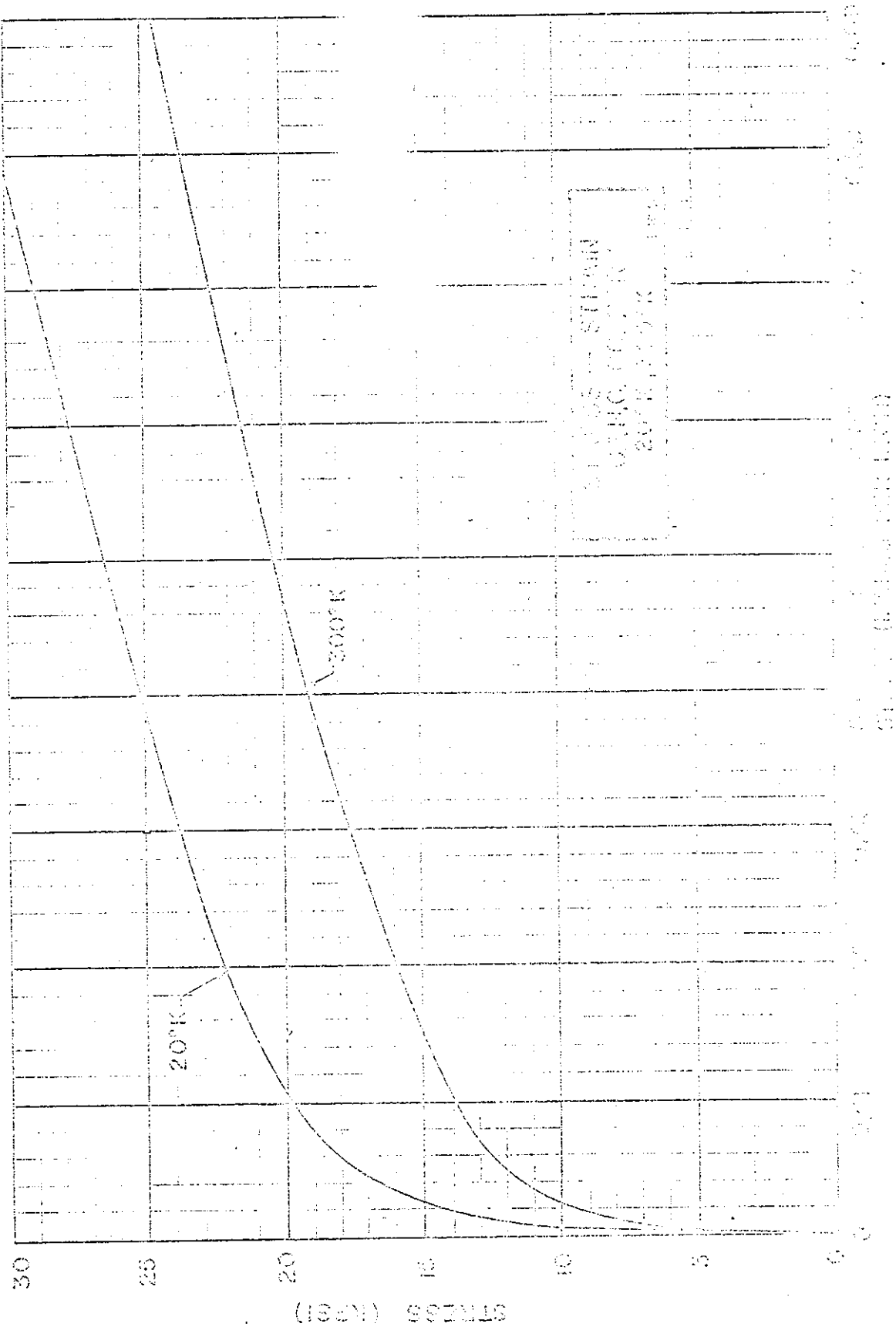
(7-64)

Schwartzberg, et al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-28C  
AD609562, Air Force Materials Lab., Aug. 1964

(DICK REED, N.B.S. BOULDER 1961)

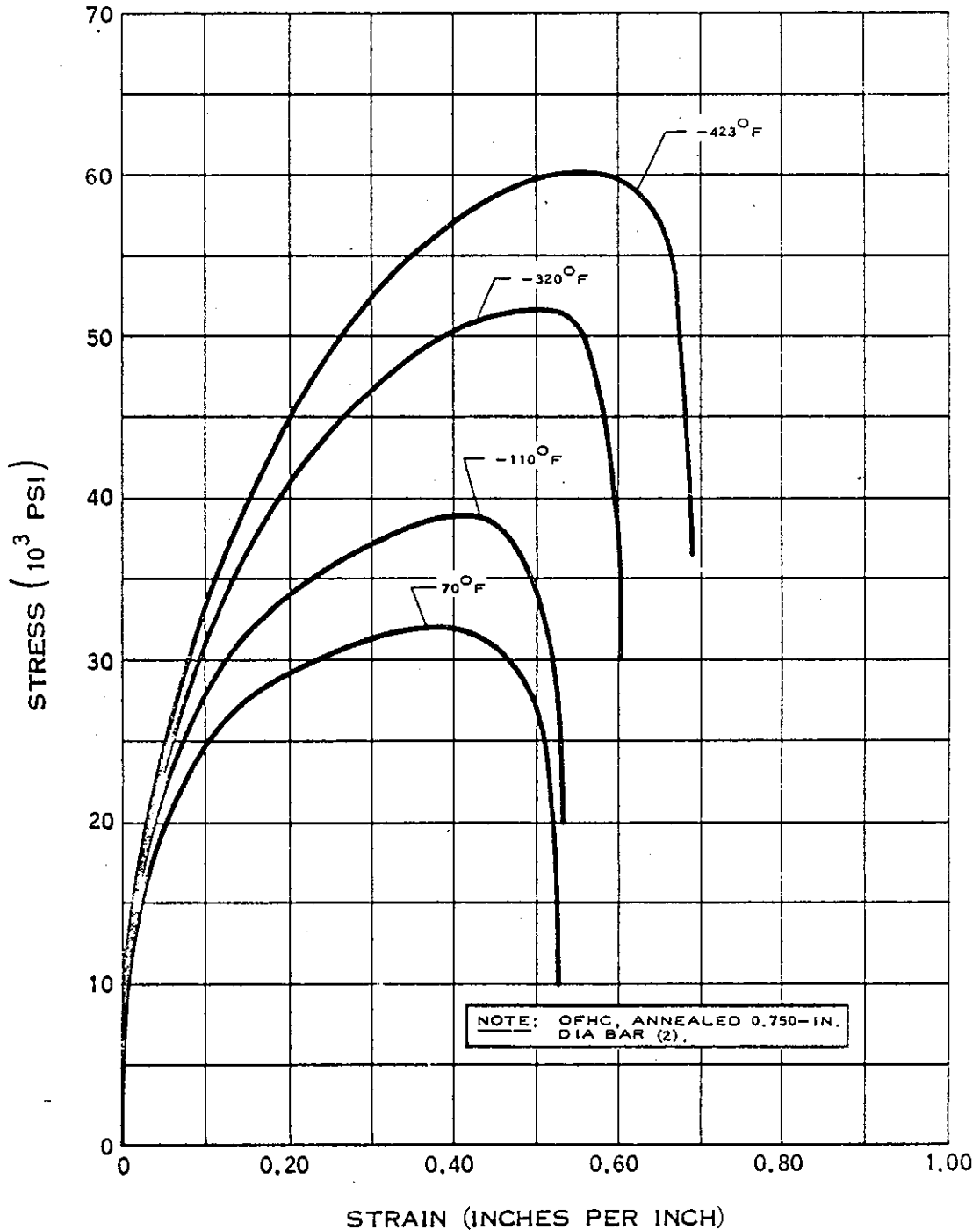


(DISK HEAD, U.S. HOLDER B61)





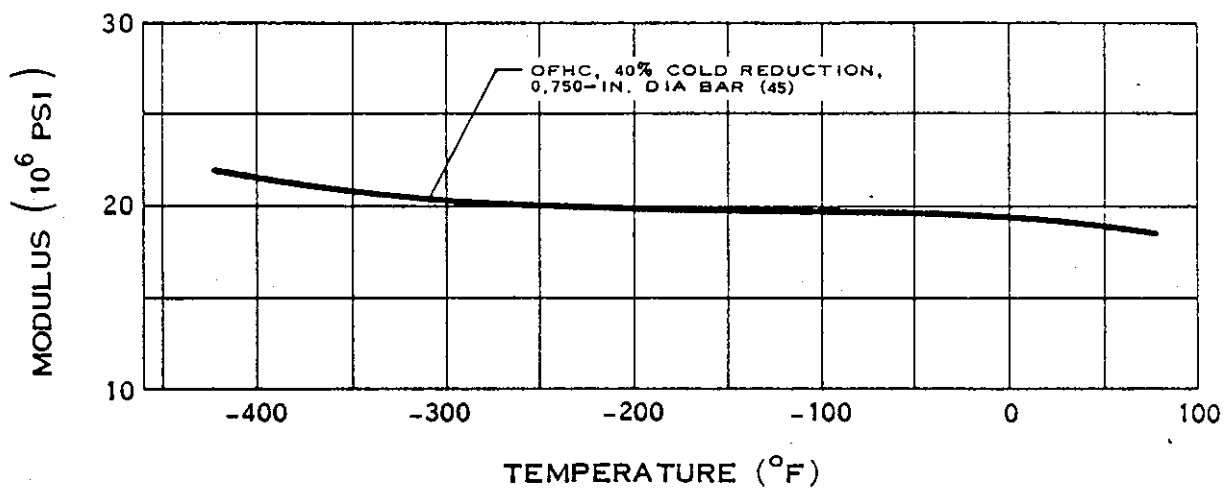
# F.1.h



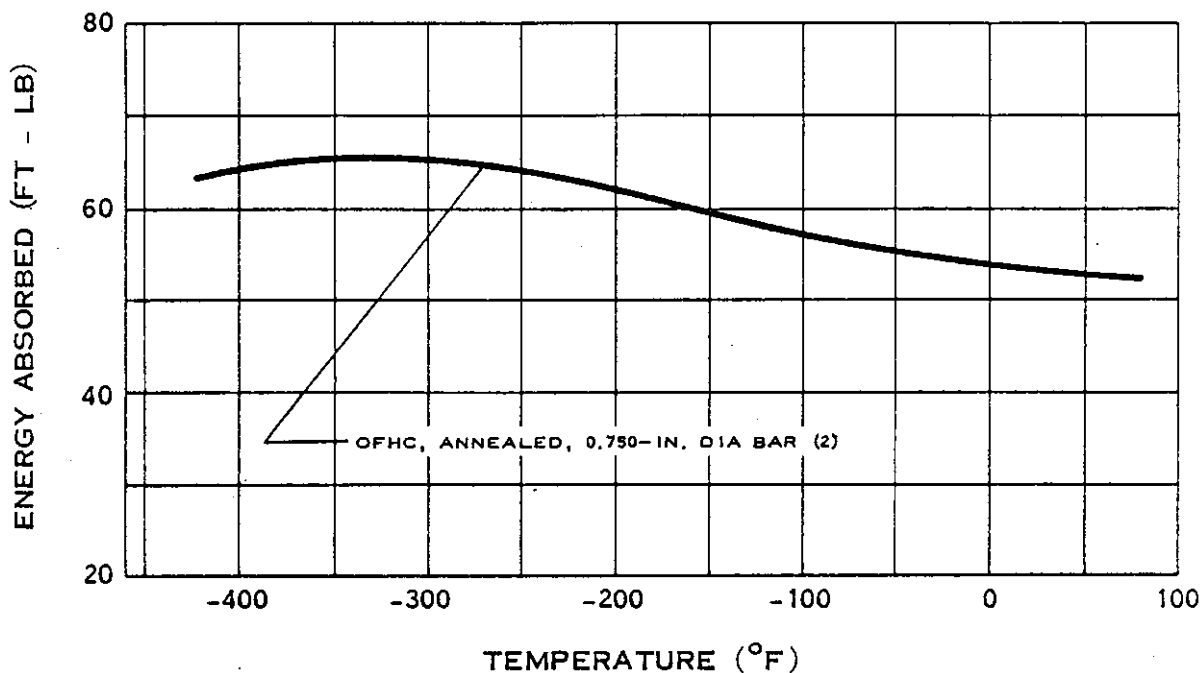
## STRESS-STRAIN DIAGRAM FOR COPPER

Schwartzberg, et al., Cryogenic Material  
Data Handbook, Technical Documentary  
Report ML-TDR-64-280 AD609562, Air  
Force Materials Lab., Aug. 1964.

### F.1.ij



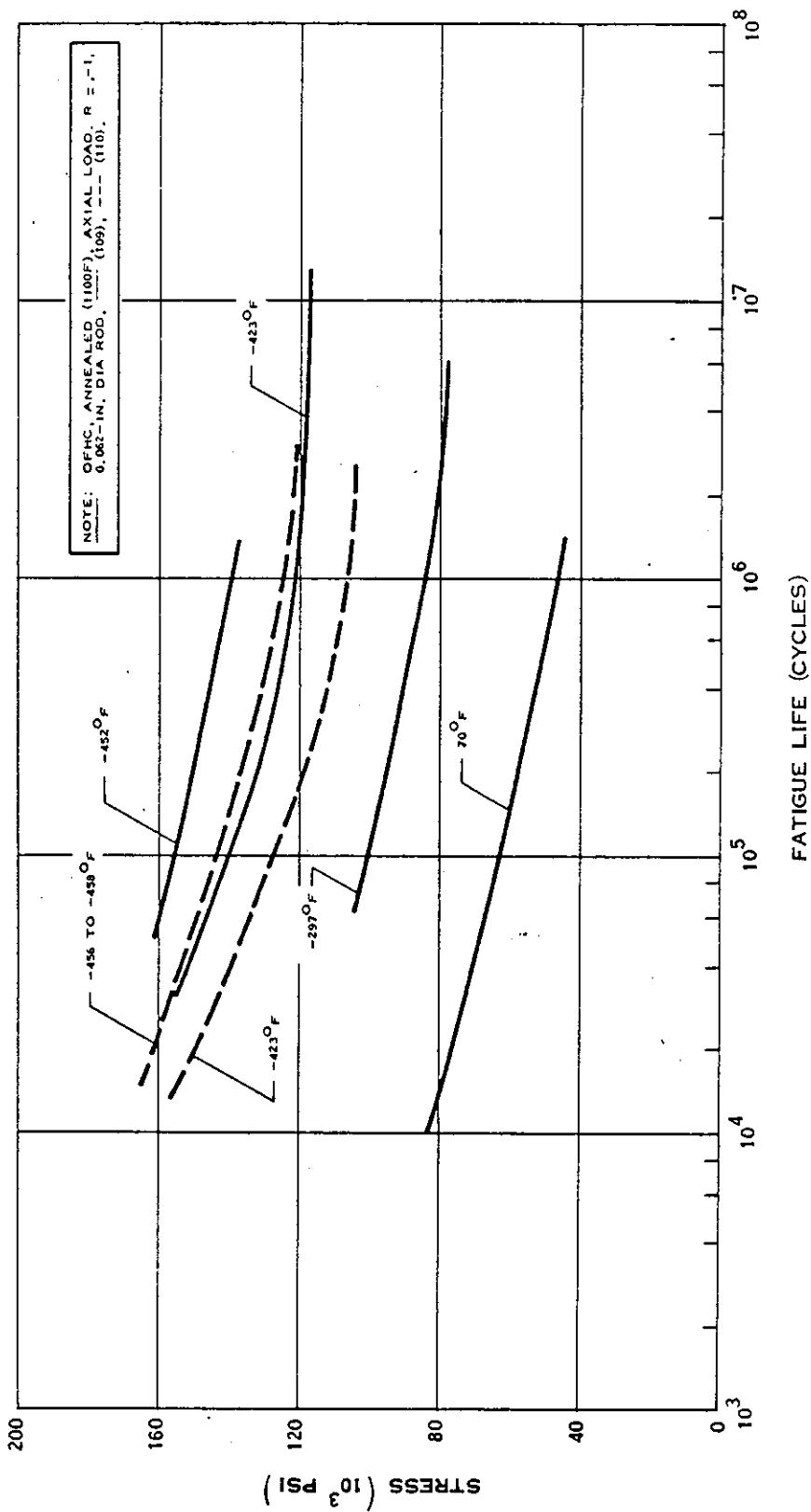
### MODULUS OF ELASTICITY OF COPPER



### IMPACT STRENGTH OF COPPER

Schwartzberg, et al., Cryogenic Material Data Handbook, Technical Documentary Report ML-TDR-64-280 AD609562, Air Force Materials Lab., Aug. 1964.

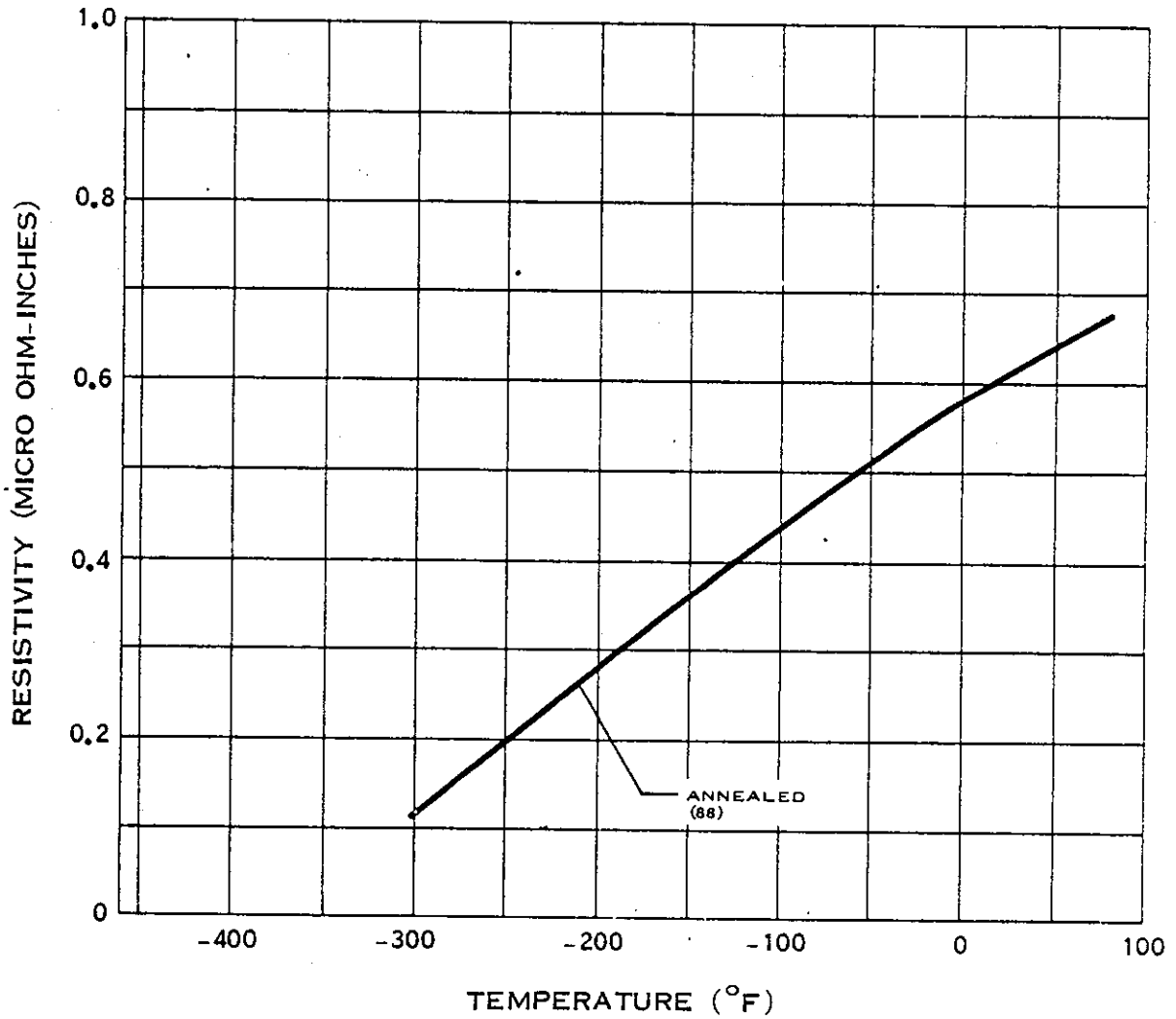
F.1.0



### FATIGUE STRENGTH OF COPPER

Schwartzberg, et al., Cryogenic Material Data Handbook, Technical Documentary Report ML-TDR-64-280 AD609562, Air Force Materials Lab., August 1964.

F.1.w



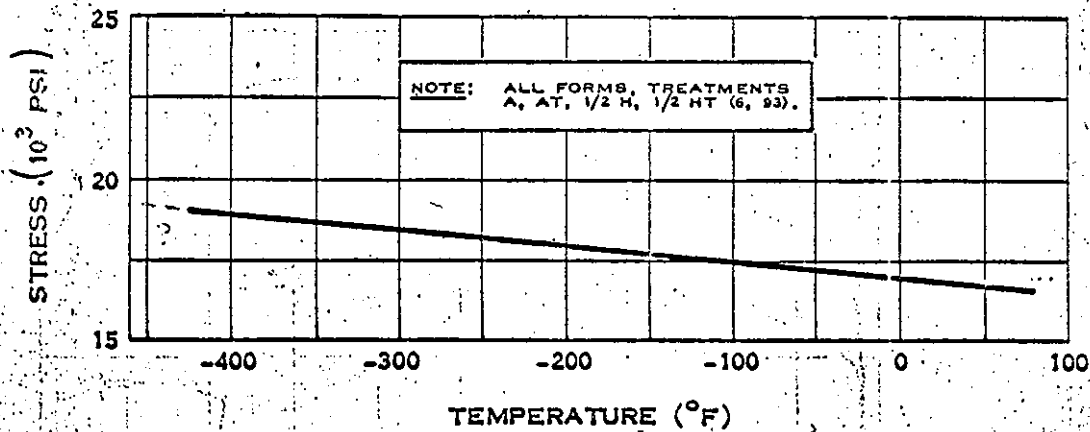
## ELECTRICAL RESISTIVITY OF COPPER

Schwartzberg, et al., Cryogenic Material Data Handbook, Technical Documentary Report ML-TDR-64-280 AD609562, Air Force Materials Lab., Aug. 1964.

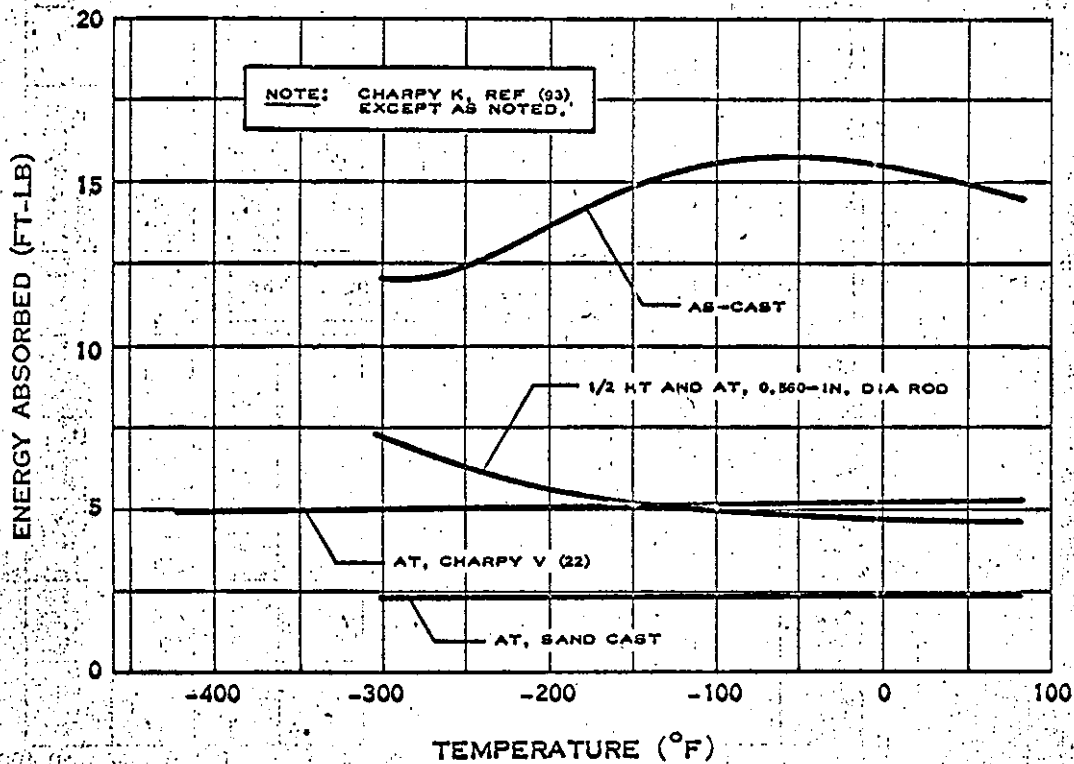
SECTION 11

STIFFENERS

F.2.1



MODULUS OF ELASTICITY OF BERYLLIUM COPPER

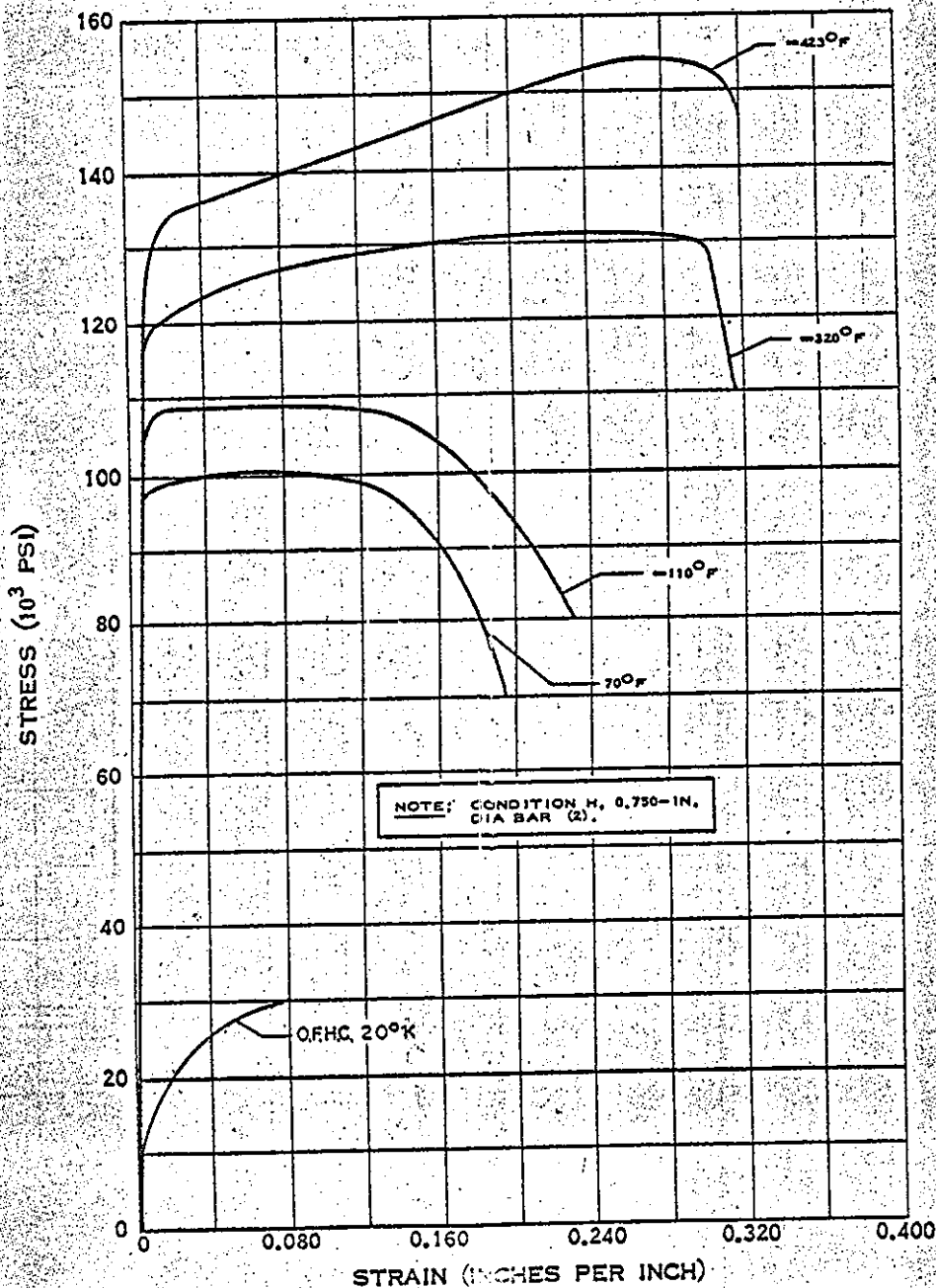


IMPACT STRENGTH OF BERYLLIUM COPPER

(7-64)

Schwartzberg, et al., Cryogenic Material Data Handbook,  
 Tech. Doc. Report ML-TDR-64-280 AD609562,  
 Air Force Materials Lab., Aug. 1964

F.2.h-1

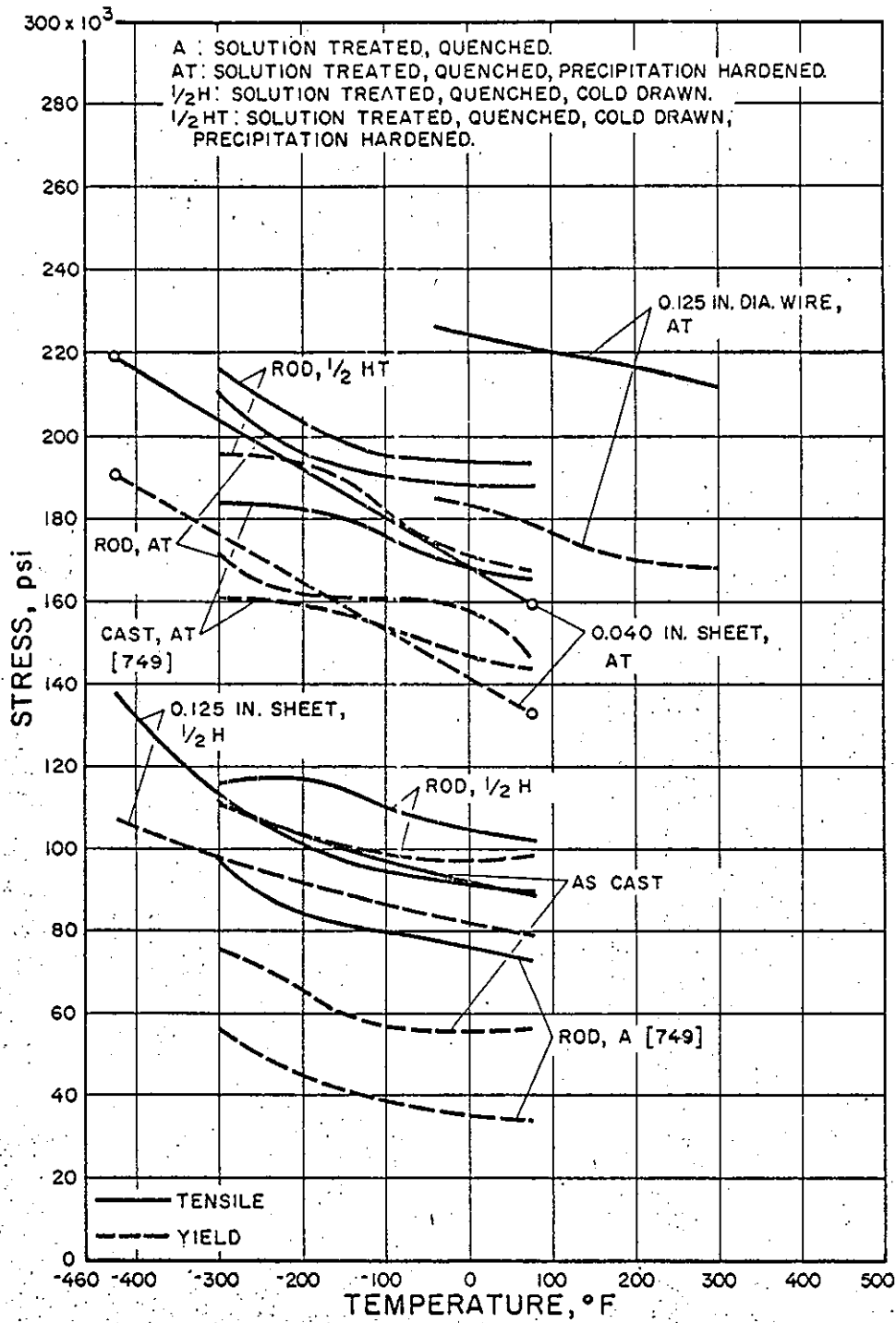


**STRESS-STRAIN DIAGRAM FOR BERYLLIUM COPPER**

(7-64)

Schwartzberg, et al., Cryogenic Material Data Handbook,  
Tech. Doc. Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964.

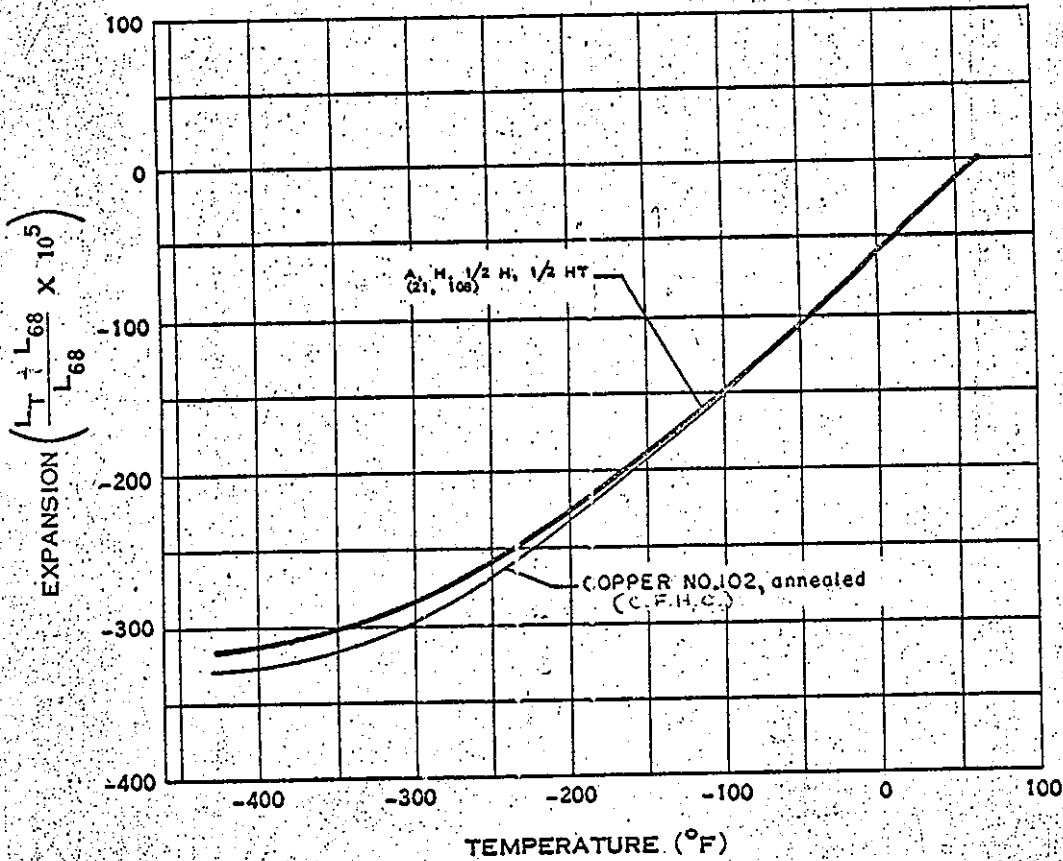
# STRENGTH OF BERYLCO 25



BERYLCO TECH. BUL. 1040A



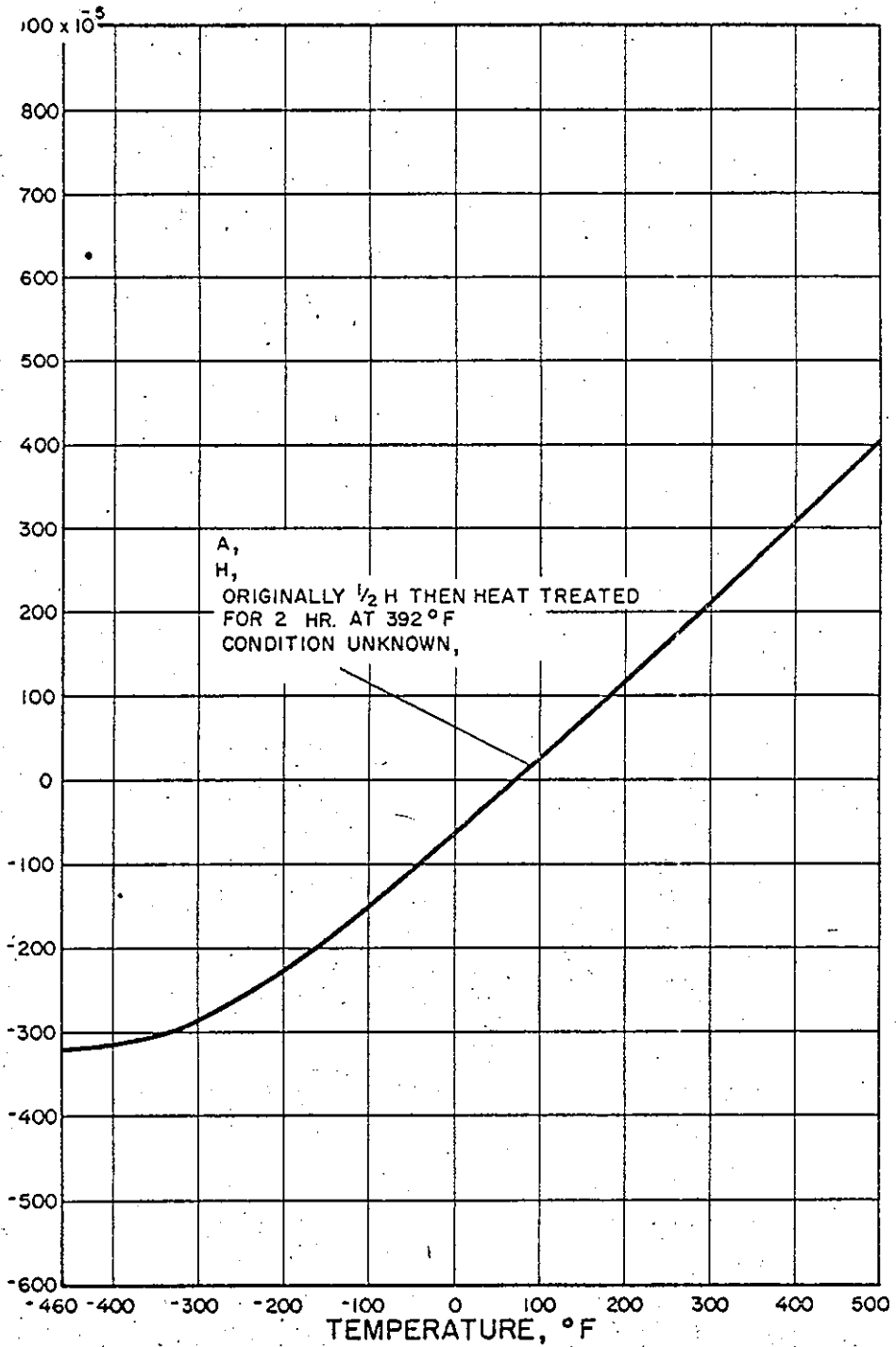
F.2.t



### THERMAL EXPANSION OF BERYLLIUM COPPER

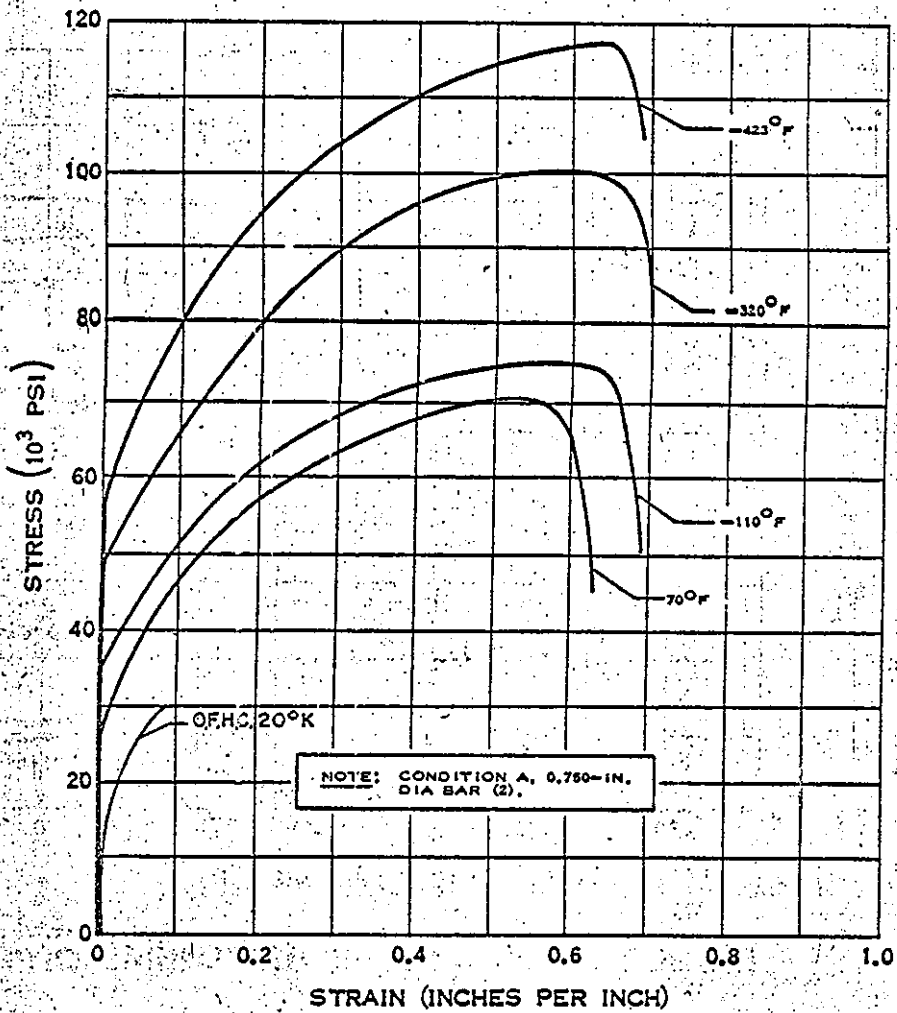
Schwartzberg, et al., Cryogenic Material Data Handbook,  
Tech. Doc. Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

# THERMAL EXPANSION OF BERYLCO 25



Berylco Tech, Bul. 1040A.

F.2-h



### STRESS-STRAIN DIAGRAM FOR BERYLLIUM COPPER

Schwartzberg, et al., Cryogenic Material Data Handbook,

Tech. Doc. Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

(7-64)

KE 10 X 10 TO THE CENTIMETER 46 1513  
MADE IN U.S.A.  
KEUFFEL & ESSER CO.

STRAIN, IN. per IN.

0.0025  
0.0024  
0.0023  
0.0022  
0.0021  
0.0020  
0.0019  
0.0018  
0.0017  
0.0016  
0.0015  
0.0014  
0.0013  
0.0012  
0.0011  
0.0010  
0.0009  
0.0008  
0.0007  
0.0006  
0.0005  
0.0004  
0.0003  
0.0002  
0.0001

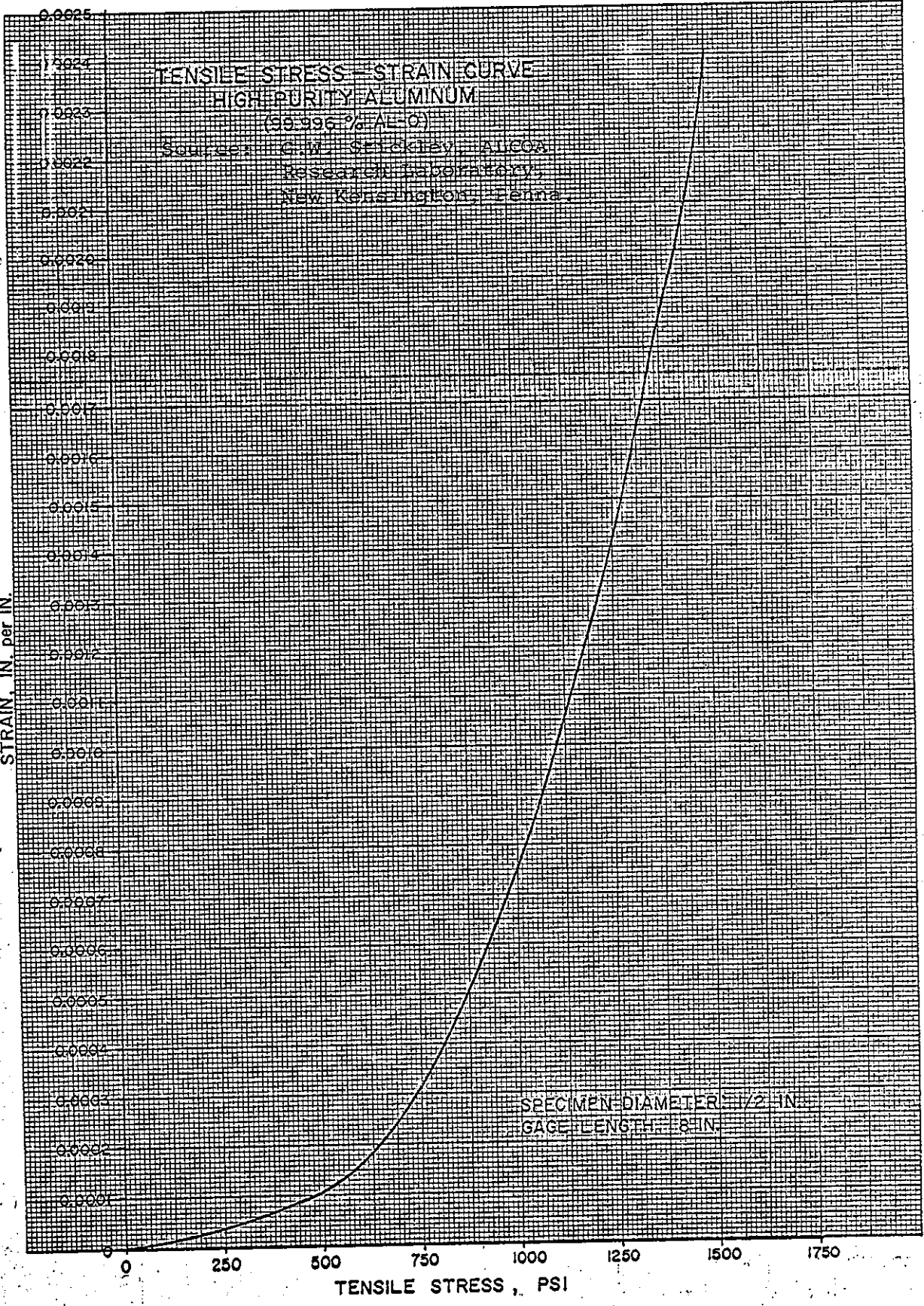
### TENSILE STRESS-STRAIN CURVE HIGH PURITY ALUMINUM

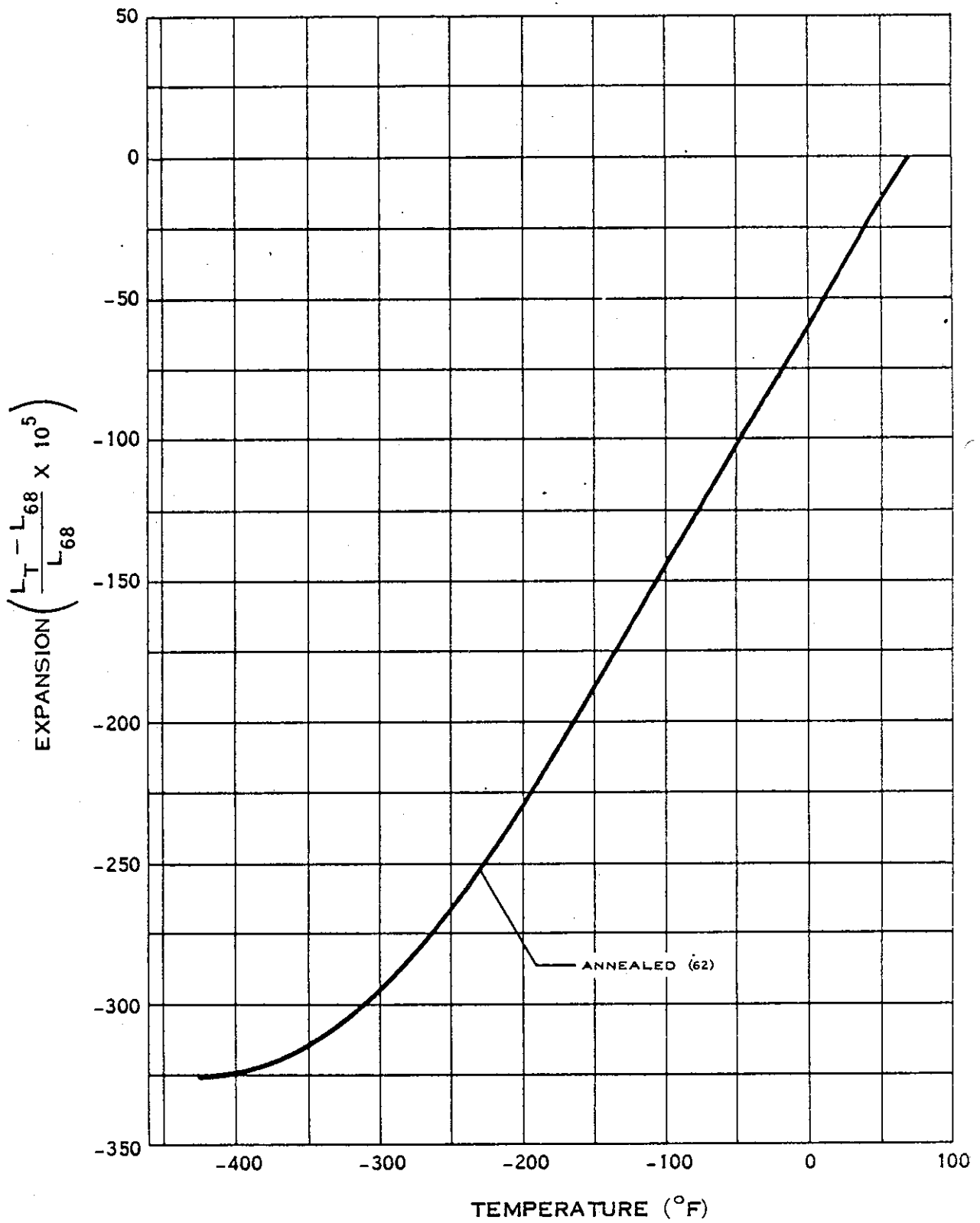
(99.996% AL-O)

Source: G.W. Stickley, ALCOA  
Research Laboratory,  
New Kensington, Penna.

SPECIMEN DIAMETER: 1/2 IN.  
GAGE LENGTH: 8 IN.

0 250 500 750 1000 1250 1500 1750  
TENSILE STRESS, PSI

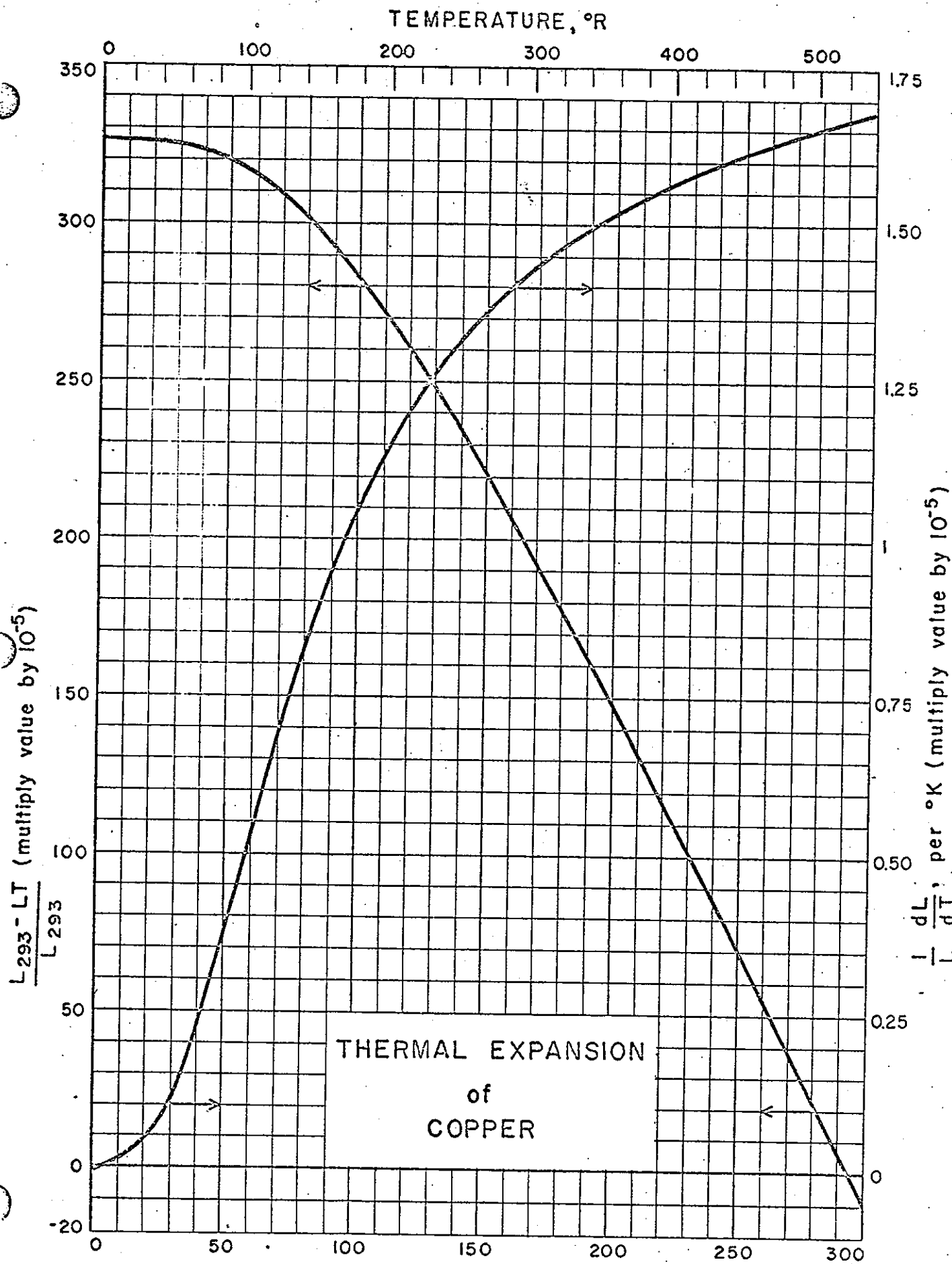




### THERMAL EXPANSION OF COPPER

(7-64)

Schwartzberg, et al., Cryogenic Material Data Handbook, Technical Documentary Report ML-TDR-64-280 AD609562, Air Force Materials Lab., Aug. 1964.



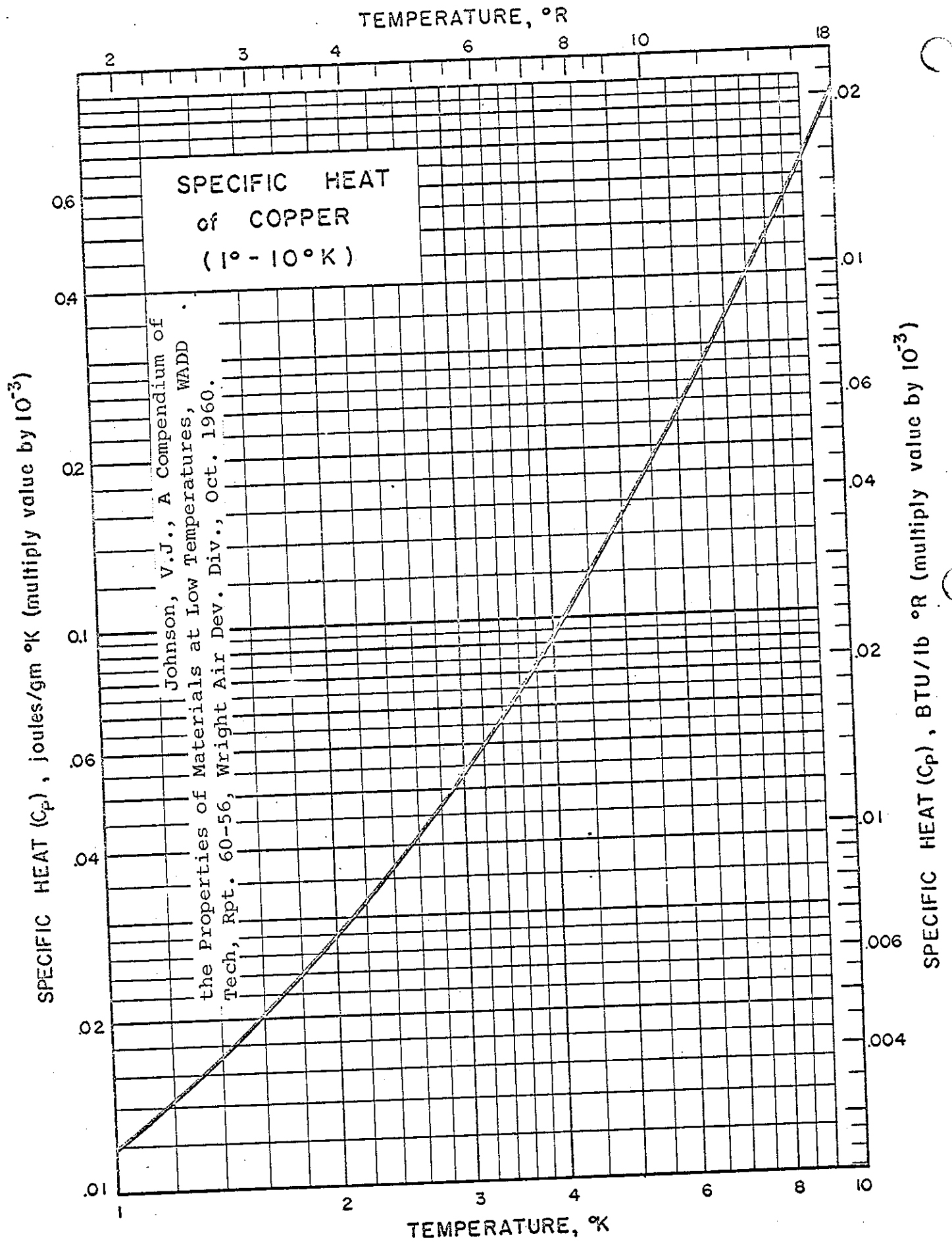
$\frac{L_{293} - L_T}{L_{293}}$  (multiply value by  $10^{-5}$ )

$\frac{1}{L} \frac{dL}{dT}$ , per °K (multiply value by  $10^{-5}$ )

TABLE 2.2. Linear thermal contraction and coefficients of linear thermal expansion—Continued

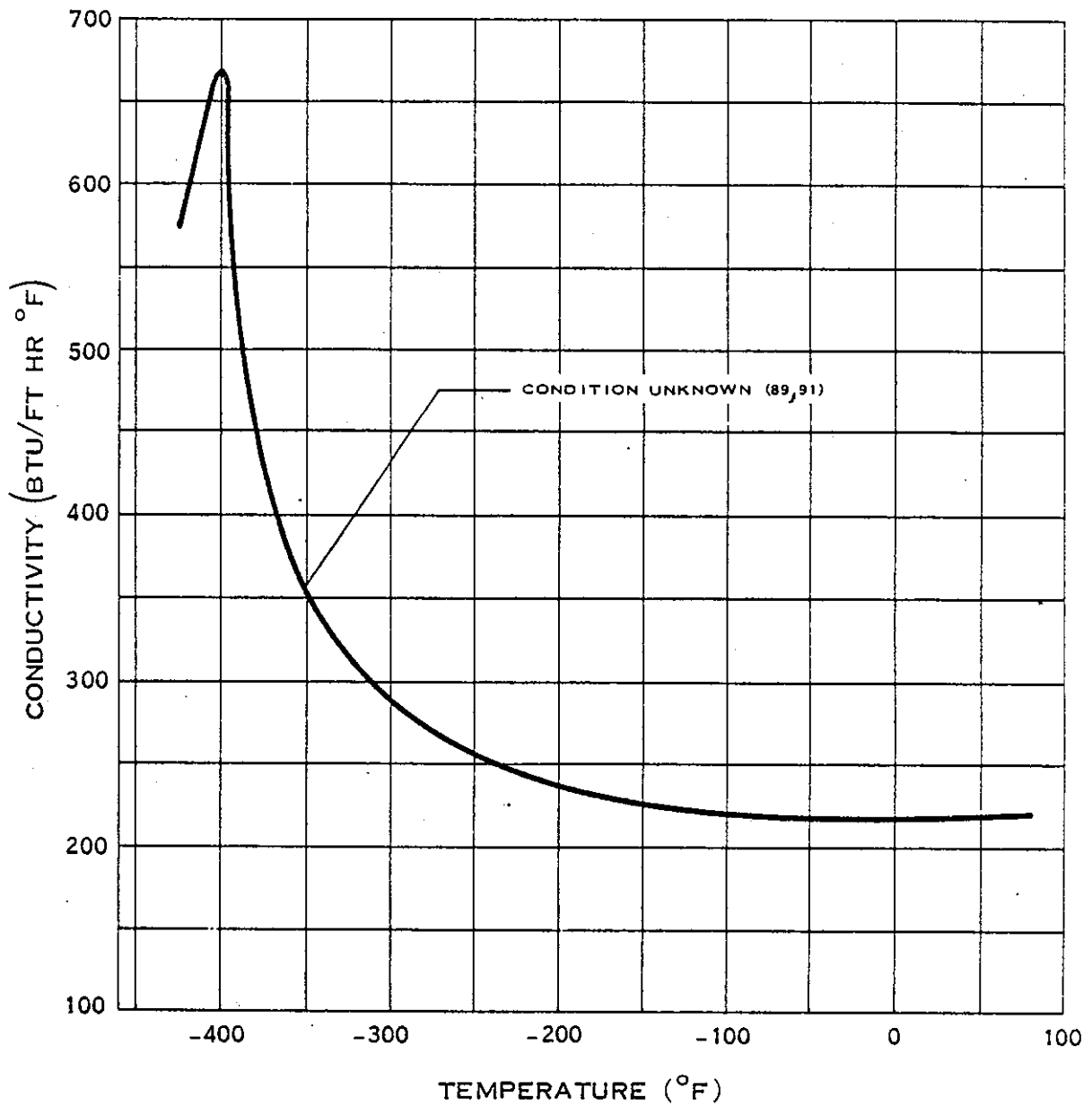
T	Alloys					Sources of above data	Other refs.		
	Steel, SAE 52160 <sup>b</sup>	Steel, AISI 304 <sup>c</sup>	Steel, AISI 302 <sup>d</sup>	Steel, AISI 301 <sup>e</sup>	Steel, AISI 303 <sup>f</sup>			Steel, AISI 316 <sup>g</sup>	Steel, AISI 310 <sup>h</sup>
$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$	$\frac{10^6}{L_{23}} \frac{dL}{dT}$
$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$	$\frac{d\alpha}{dT} \text{ } ^\circ\text{K}^{-1}$
0	• 316	• 0.01	0	• 295	• 0.01	• 297	• 415	• 295	• 295
10	246	• 0.01	0	246	• 0.01	247	• 415	• 295	• 295
20	395	• 0.03	• 0.03	247	• 0.03	247	415	• 295	• 295
30	316	• 0.03	• 0.03	247	• 0.03	247	415	• 295	• 295
40	315	1.4	1.4	246	1.4	246	413	• 291	• 291
50	313	2.7	2.7	244	2.7	244	410	• 291	• 291
60	309	4.3	4.3	240	4.3	240	403	• 216	• 216
70	304	5.5	5.5	235	5.5	235	389	• 216	• 216
80	296	6.5	6.5	227	6.5	227	384	• 216	• 216
90	287	8.3	8.3	223	8.3	223	381	• 216	• 216
100	287	9.6	9.6	221	9.6	221	381	• 216	• 216
110	277	10.1	10.1	221	10.1	221	379	• 216	• 216
120	265	11.0	11.0	218	11.0	218	373	• 216	• 216
130	243	11.8	11.8	213	11.8	213	373	• 216	• 216
140	195	7.6	7.6	188	7.6	188	277	• 216	• 216
150	140	8.2	8.2	163	8.2	163	277	• 216	• 216
160	133	8.0	8.0	163	8.0	163	210	• 216	• 216
180	105	9.7	9.7	136	9.7	136	201	• 216	• 216
200	81.4	10.5	10.5	111	10.5	111	193	• 216	• 216
220	62.4	11.5	11.5	85.6	11.5	85.6	118	• 216	• 216
240	39.4	11.7	11.7	57.0	11.7	57.0	75	• 216	• 216
253	21.0	11.9	11.9	35.0	11.9	35.0	45	• 216	• 216
250	15.6	12.0	12.0	29.2	12.0	29.2	30	• 216	• 216
253	0.0	12.1	12.1	0.0	12.1	0.0	0	• 216	• 216
253	0.0	12.1	12.1	0.0	12.1	0.0	0	• 216	• 216
300	-8.5	12.1	12.1	-11.0	12.1	-11.0	-11	• 216	• 216

<sup>a</sup> Estimated.  
<sup>b</sup> The measurements were made on a German steel that approximated the specifications of SAE 52160. Its composition was 0.61 C, 0.27 Si, 0.31 Mn, 0.93 Cr, bal. Fe. The above data adequately represent two hard (800-1000 HB) and one of which was "heat treated at 750 °C", the other "quenched from 800 °C in oil and tempered at 630 °C". The expansion coefficients in the "soak" condition were also measured and were about 1/2 to 1/3 lower than the above values at 40 °K, about the same from 125 to 240 °K, and about 10% lower at 300 °K.  
<sup>c</sup> 0.13 C, 0.50 Mn, 0.51 Si, 16.0 Cr, 7.25 Ni, bal. Fe. Annealed 20 min at 1650 °F and water quenched. After cooling to 27 °K and returning to room temperature a small permanent expansion was found to have occurred due to irreversible formation of ferrite.  
<sup>d</sup> Composition and heat treatment of sample not stated. Composition limits for this alloy are: 0.65-0.30 C, 2 (max.) Mn, 1 (max.) Si, 17-10 Cr, 8-10 Ni, bal. Fe.  
<sup>e</sup> Composition limits for this alloy are: 0.05 (max.) C, 2 (max.) Mn, 1 (max.) Si, 16-15 Cr, 10-14 Ni, 2-3 Mo, bal. Fe.  
<sup>f</sup> 0.11 C, 1.41 Mn, 0.22 Si, 0.01 S, 0.02 P, 27.2 Cr, 21.6 Ni, bal. Fe. Annealed 30 min at 1650 °F and water quenched.  
<sup>g</sup> Composition and heat treatment of sample not stated. Composition limits for this alloy are: 0.10 (max.) C, 2 (max.) Mn, 1 (max.) Si, 16-15 Cr, 10-14 Ni, 2-3 Mo, bal. Fe.  
<sup>h</sup> Composition limits for this alloy are: 0.05 (max.) C, 2 (max.) Mn, 1 (max.) Si, 16-15 Cr, 10-14 Ni, 2-3 Mo, bal. Fe.



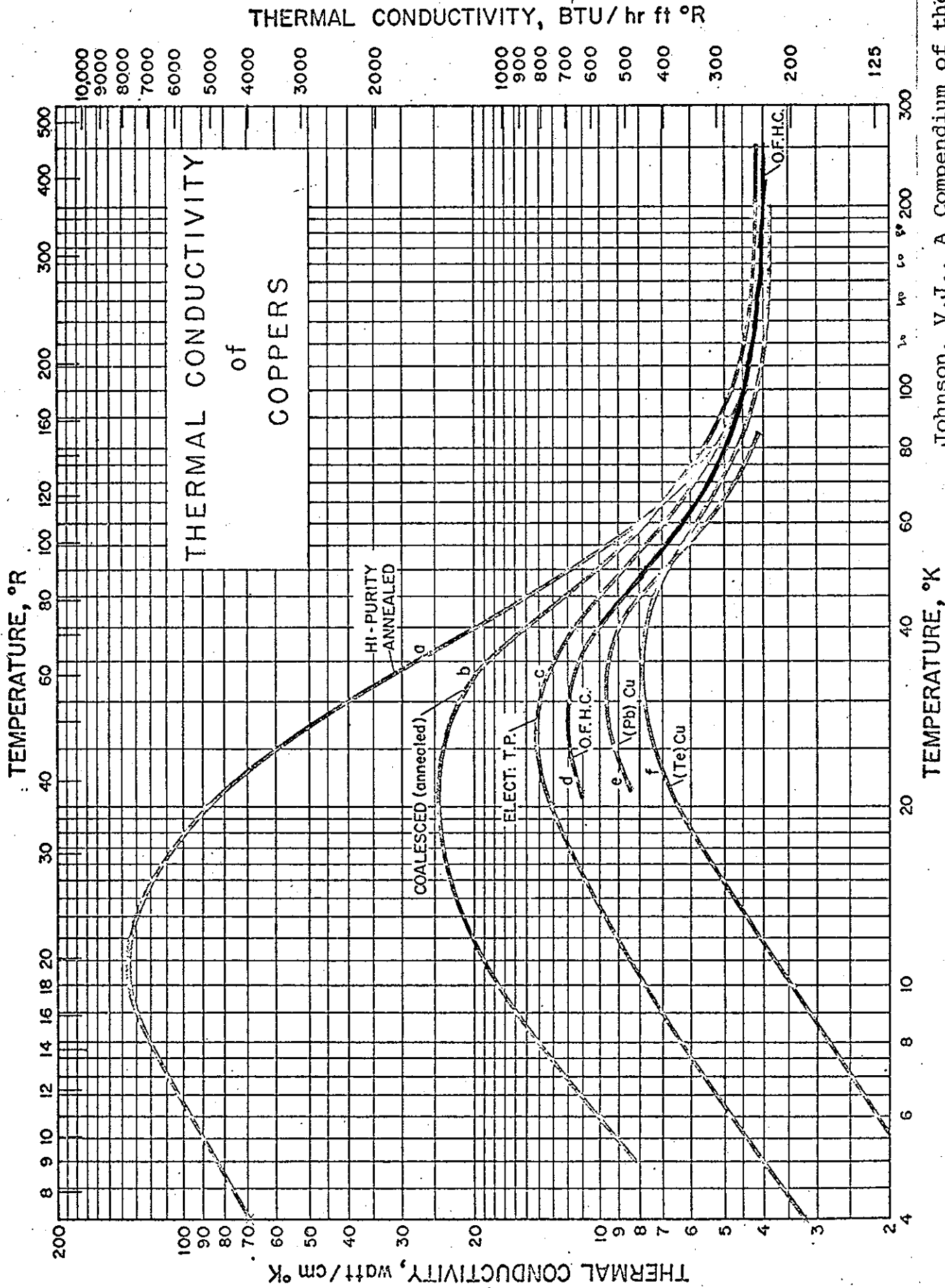


F.1.v



### THERMAL CONDUCTIVITY OF COPPER

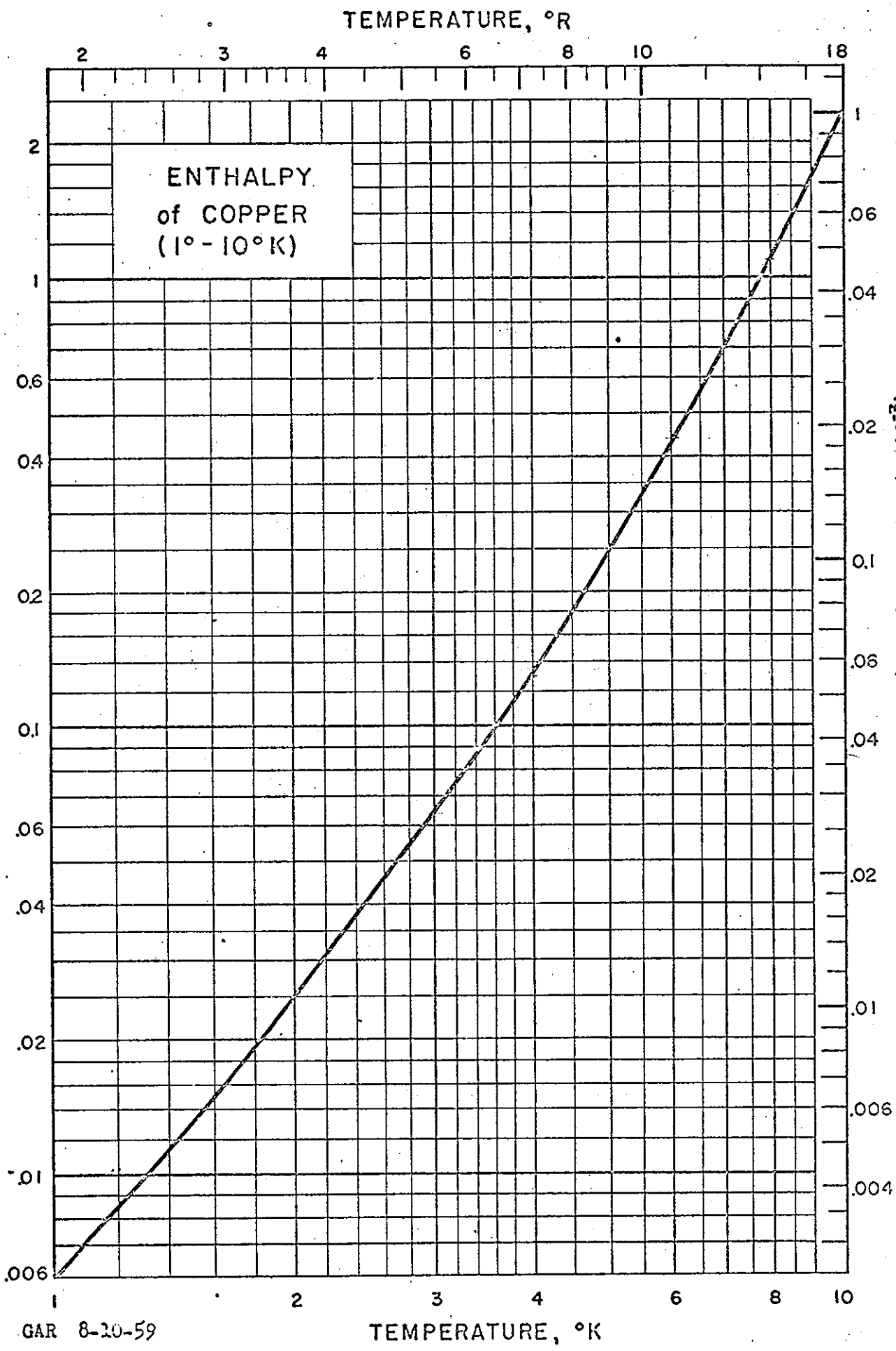
Schwartzberg, et al., Cryogenic Material Data Handbook, Technical Documentary Report ML-TDR-64-280 AD609562; Air Force Materials Lab., Aug. 1964.



Johnson, V.J., A Compendium of the Properties of Materials at Low Temperatures, WADD Tech.

SPECIFIC HEAT ( $C_p$ ), BTU/lb °R (multiply value by  $10^{-3}$ )

ENTHALPY, joules/gm (multiply value by  $10^{-3}$ )



GAR 8-10-59

TEMPERATURE, °K

Johnson, V.J., A Compendium of  
Materials at Low Temperatures. WADD

# HEAT LOSS FOR MULTISTAGE COOLED LEADS \*

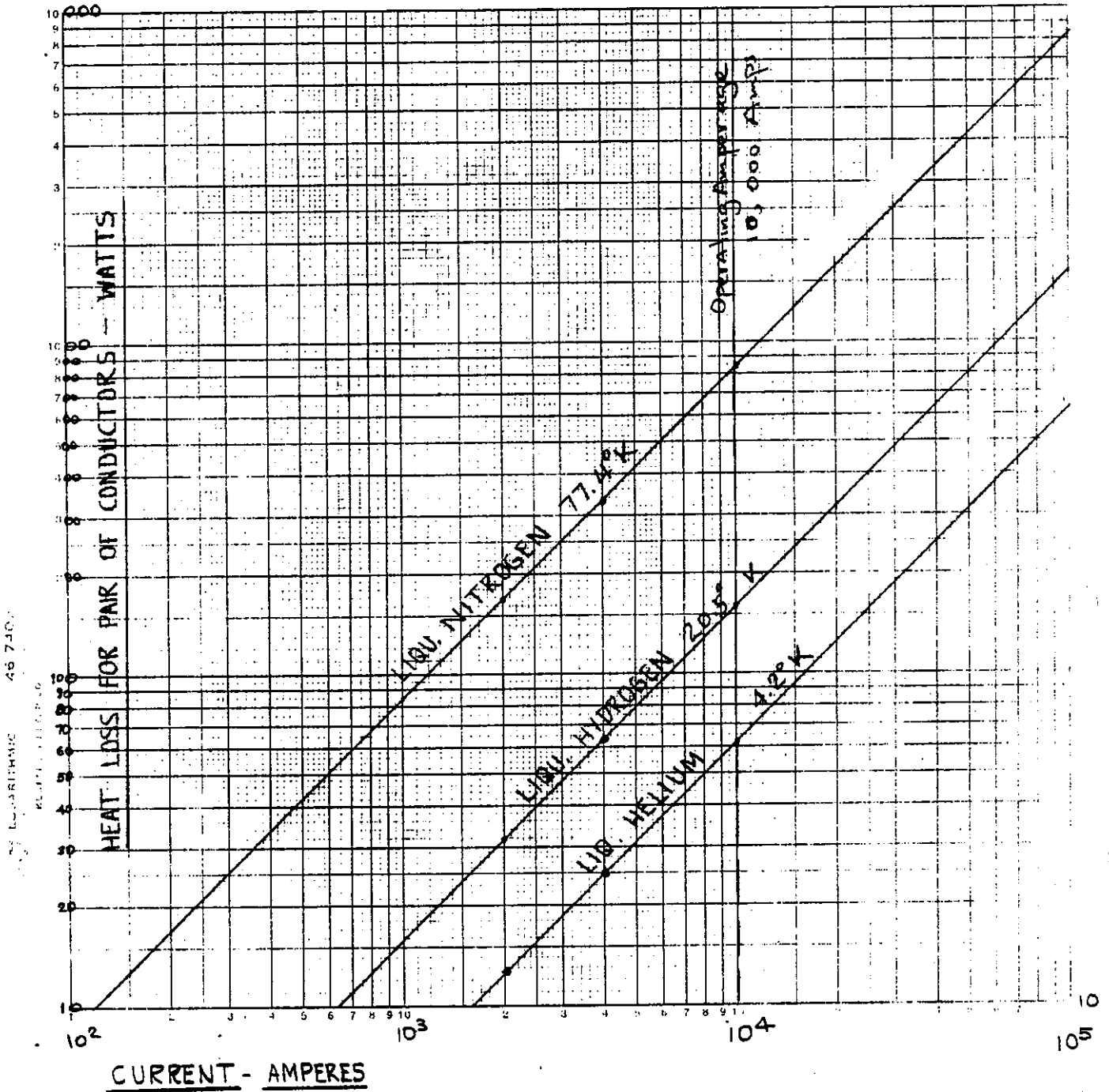
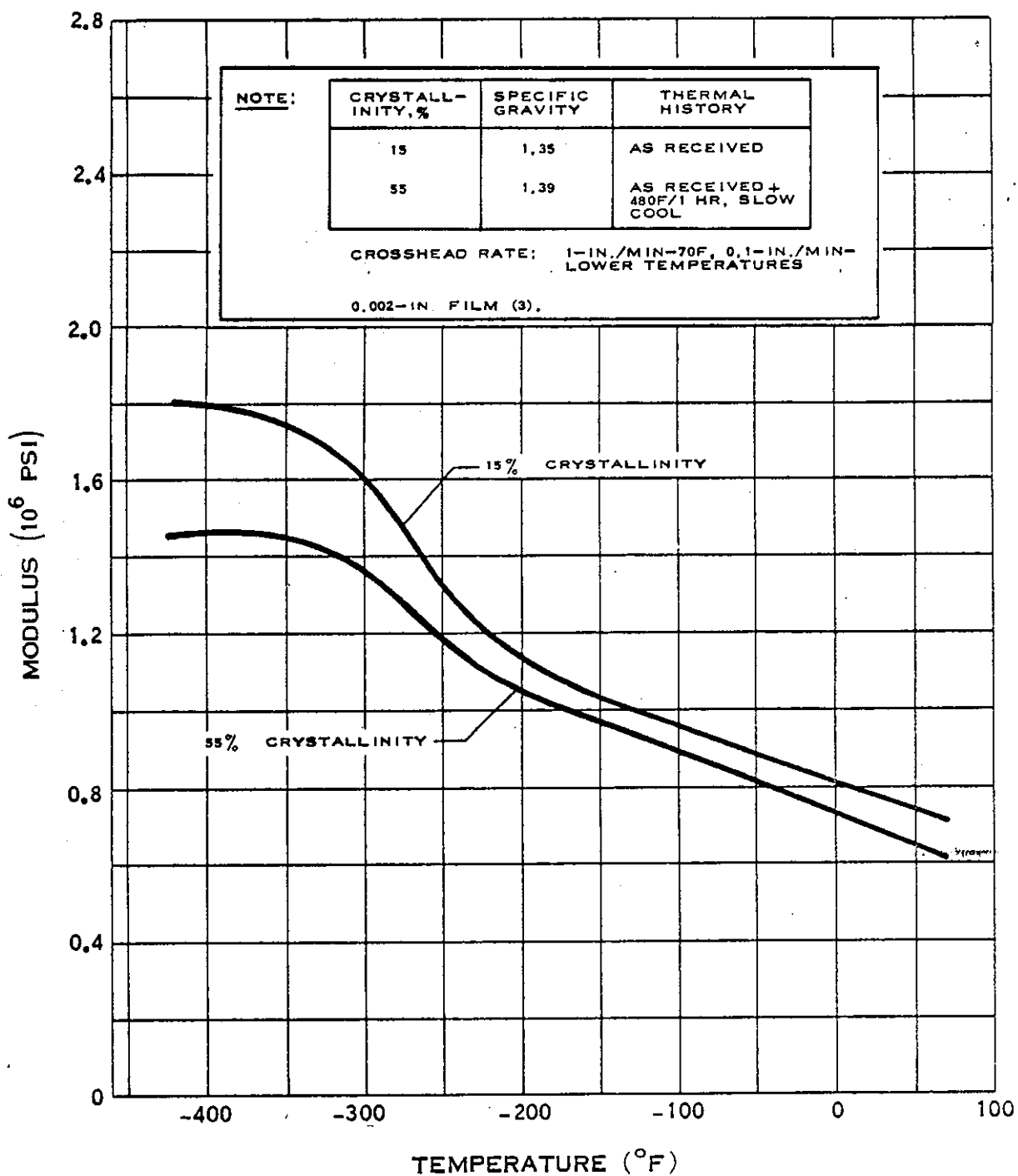


FIG. 2

\* Per McFee - Review of Scienc Int. Vol. 30 No.2

# G.2.i



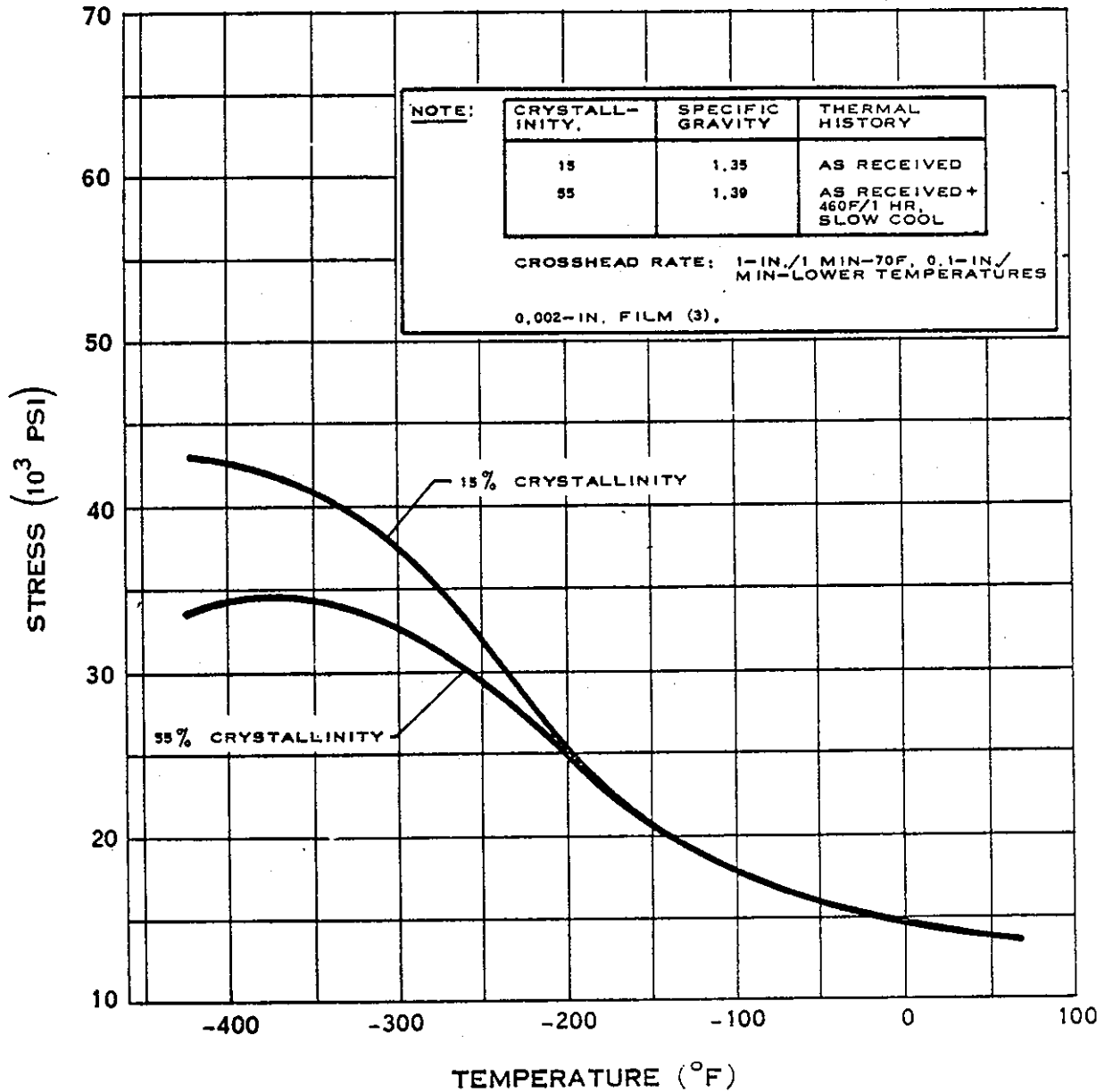
## MODULUS OF ELASTICITY OF MYLAR\*

\* T.M. E. 1; DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et. al., Cryogenic Material Data Handbook,  
 Technical Documentary Report ML-TDR-64-280 AD609562,  
 Air Force Materials Lab., Aug, 1964

# G.2.a



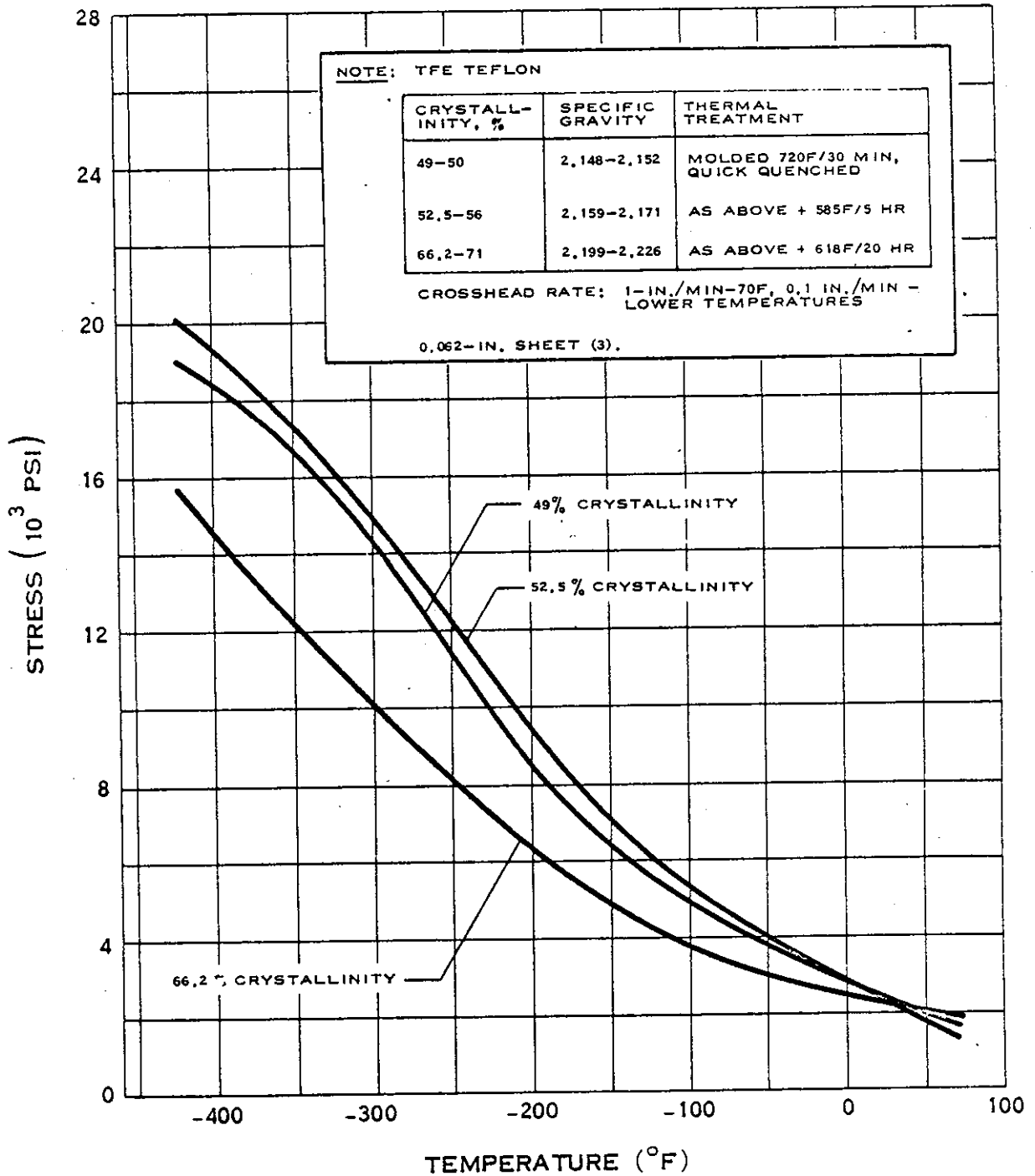
## YIELD STRENGTH OF MYLAR\*

\* T.M.  
E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schurtsberg, et. al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964

### G.3.a

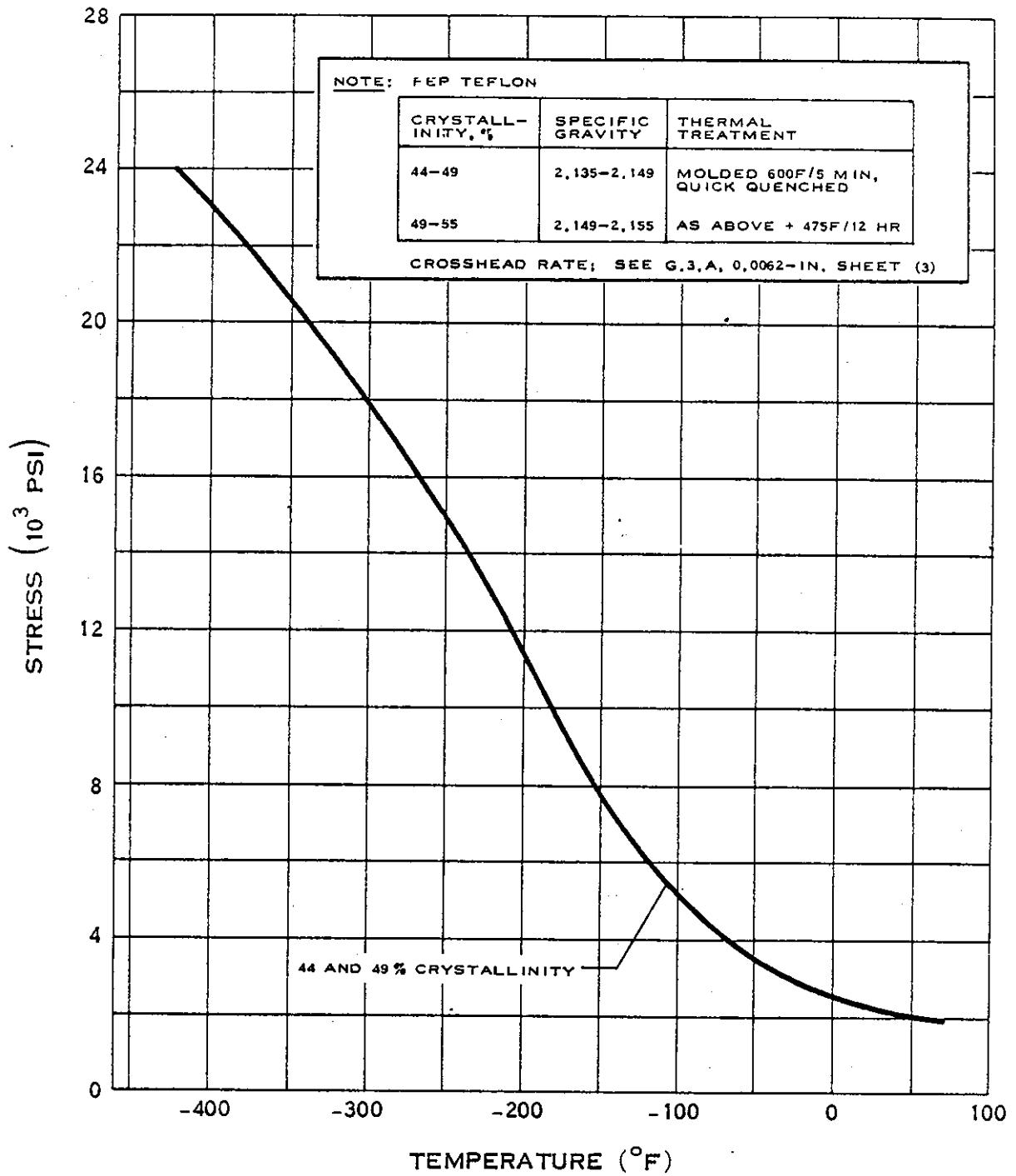


## YIELD STRENGTH OF TEFLON\* (TFE)

\* T.M. DUPONT DE NEMOURS AND CO.  
E. I.

Schwartzberg, et al., Cryogenic Material Data Handbook,  
Technical Documentary Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

# G.3.a-1



## YIELD STRENGTH OF TEFLON\* (FEF)

\* T.M. DUPONT DE NEMOURS AND CO.

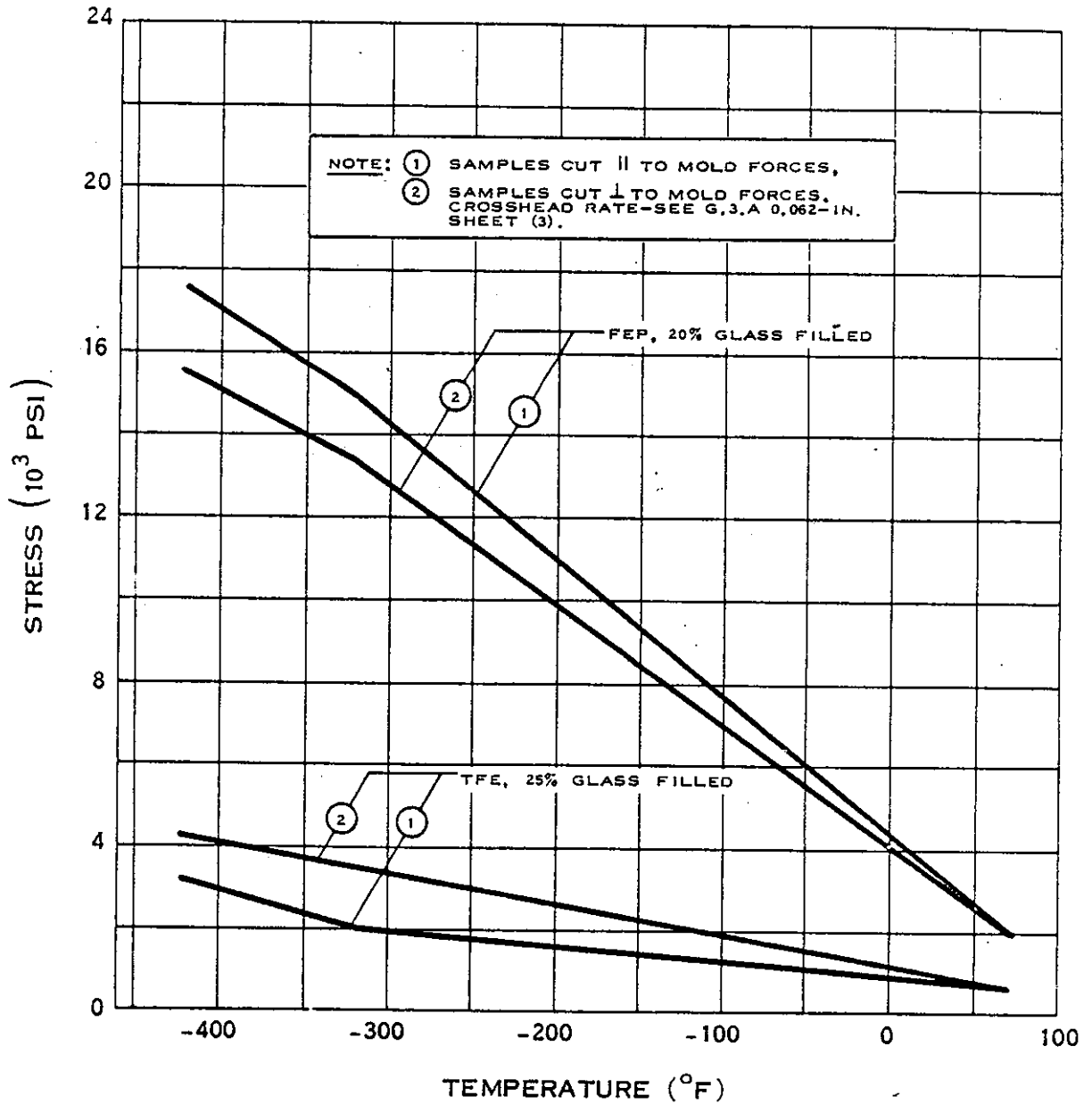
Schwartzberg, et al., Cryogenic Material Data Handbook,

Technical Documentary Report ML-TDR-64-280 AD609562,

Air Force Materials Lab., Aug. 1964



### G.3.a-3



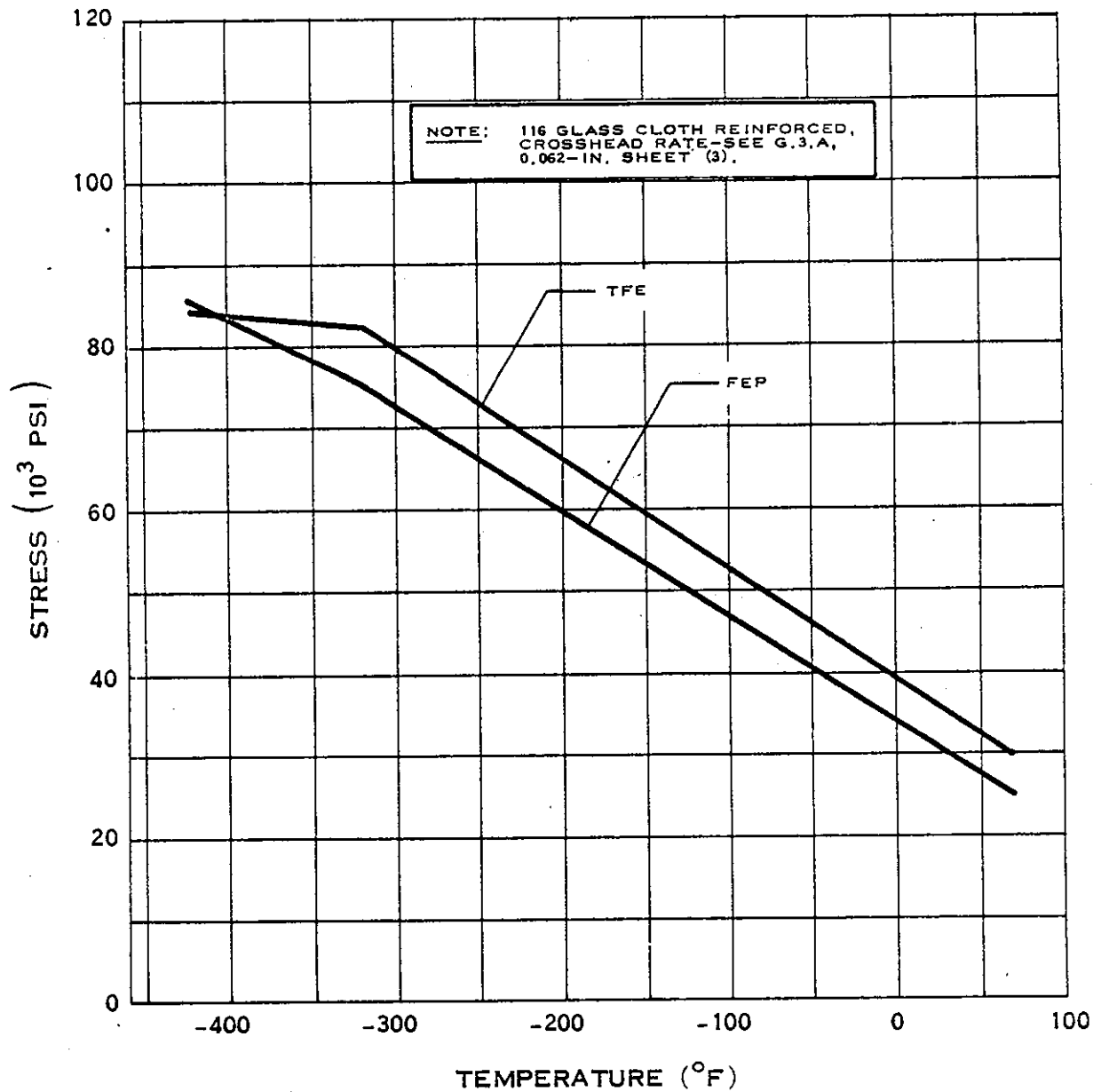
### YIELD STRENGTH OF TEFLON\*

Schwartzberg., et al., Cryogenic Material Data Handbook,

Tech. Doc. Rpt. ML-TDR-64-208 AD609562,

\* T.M.  
E. I. DUPONT DE NEMOURS AND CO. Air Force Materials Lab., Aug. 1964

### G.3.a-4



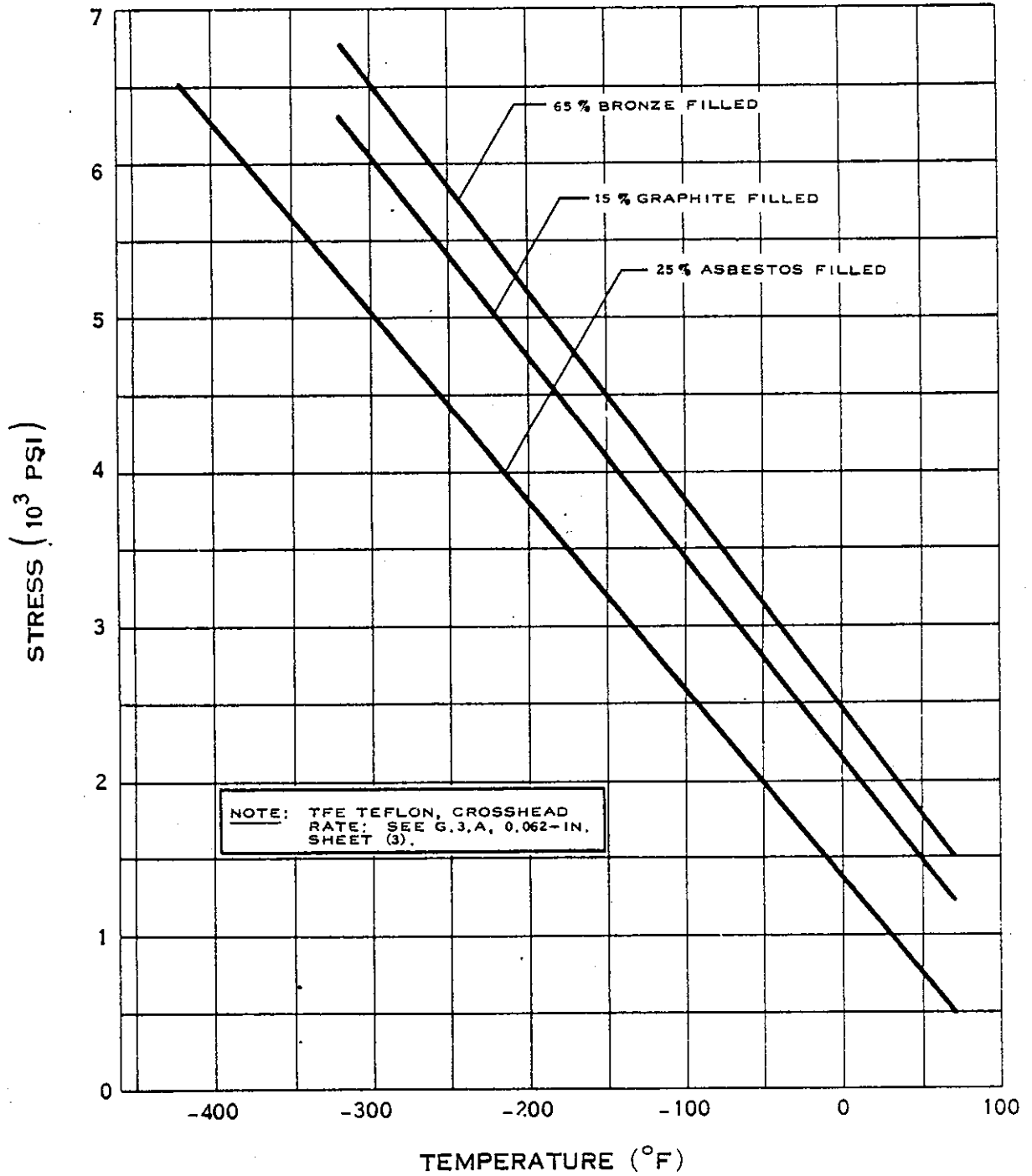
## YIELD STRENGTH OF TEFLON\*

\* T.M. E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schmartsberg, et al., Cryogenic Materials Data Handbook,  
Tech. Documentary Report ML-TDR-64-280 · AD609562,  
Air Force Materials Lab., Aug. 1964

G.3.a-2

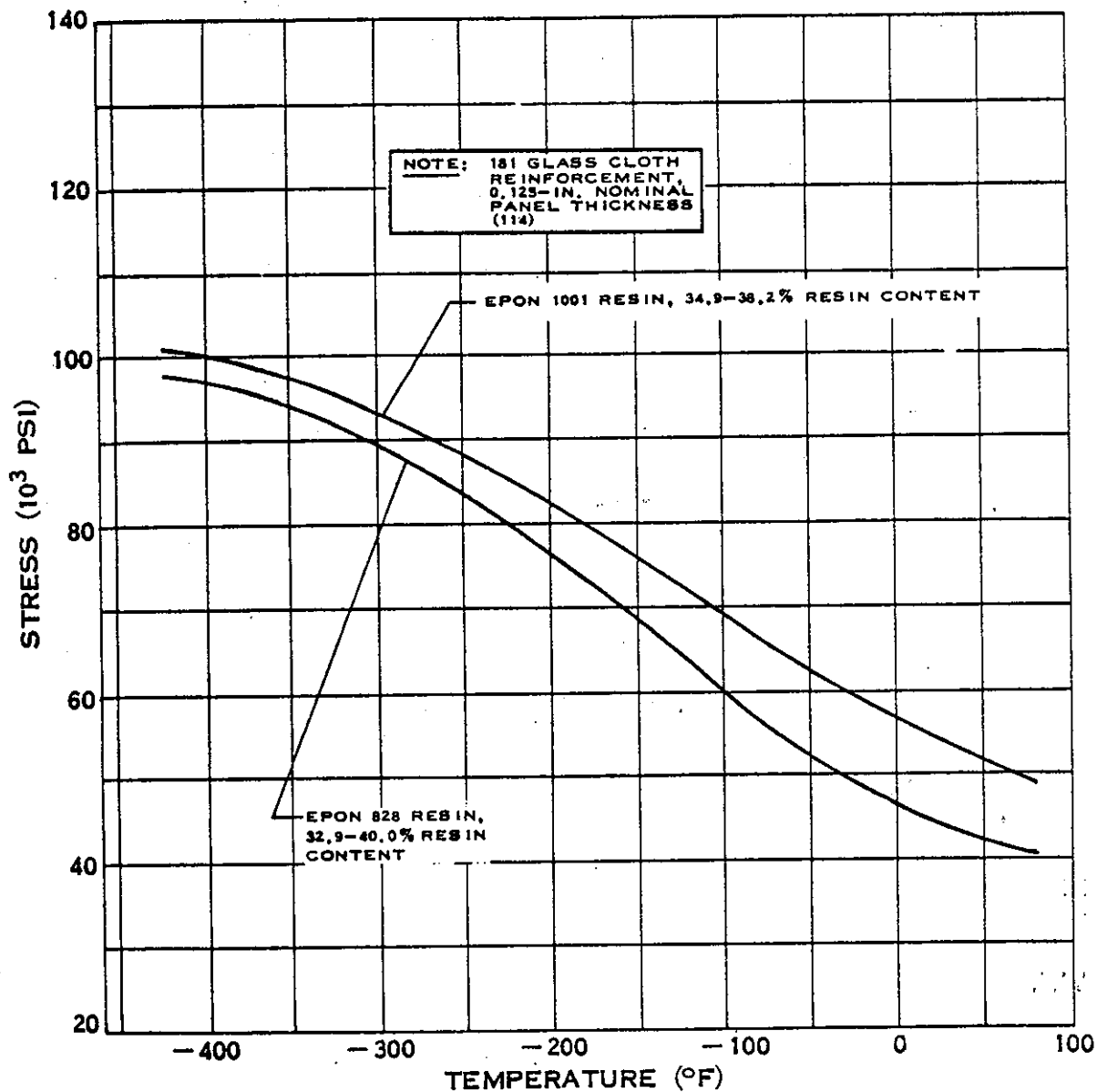


**YIELD STRENGTH OF TEFLON\* (TFE)**

Tech. Doc. Rpt. ML-TDR-54-208 XD609562,

\* T.M. E. I. DUPONT DE NEMOURS AND CO. Air Force Materials Lab., Aug. 1964

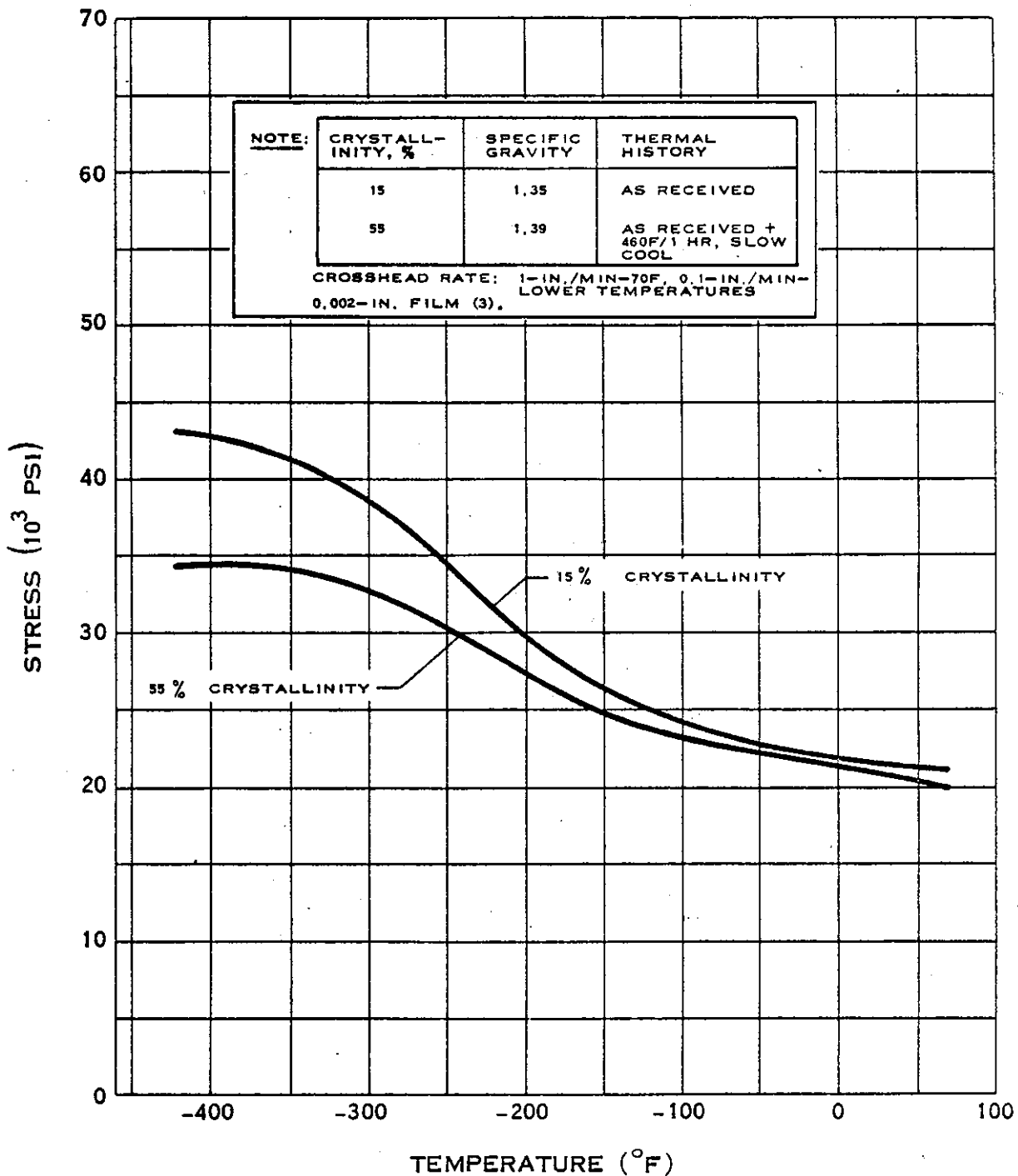
## H.1.b

**TENSILE STRENGTH OF EPOXY -  
FIBERGLASS LAMINATE**

(7-64)

Schmartzberg, et. al., Cryogenic Material Data Handbook,  
Technical Documentary Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

# G.2.b



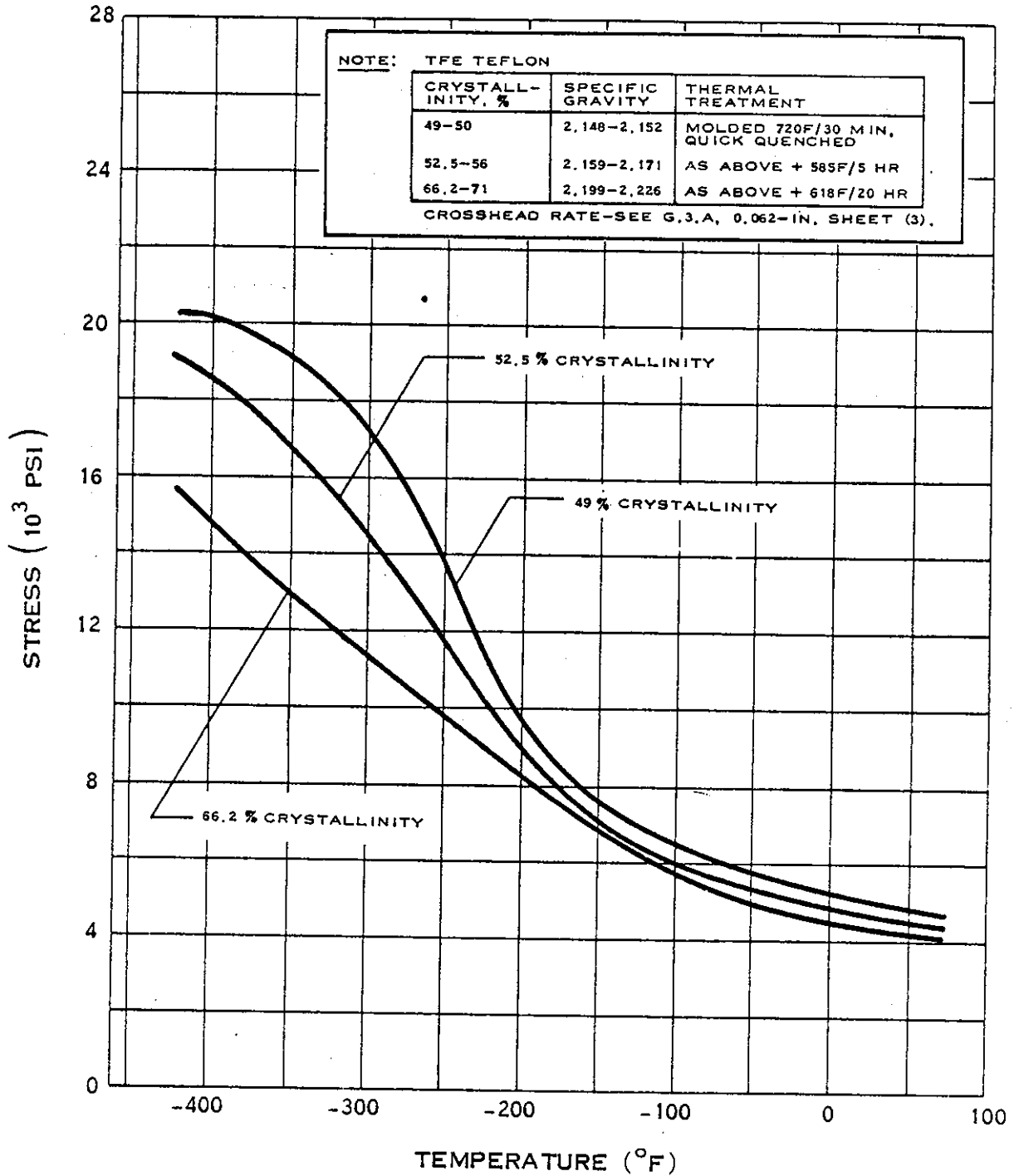
## TENSILE STRENGTH OF MYLAR\*

\* T.M. E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et. al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964

### G.3.b



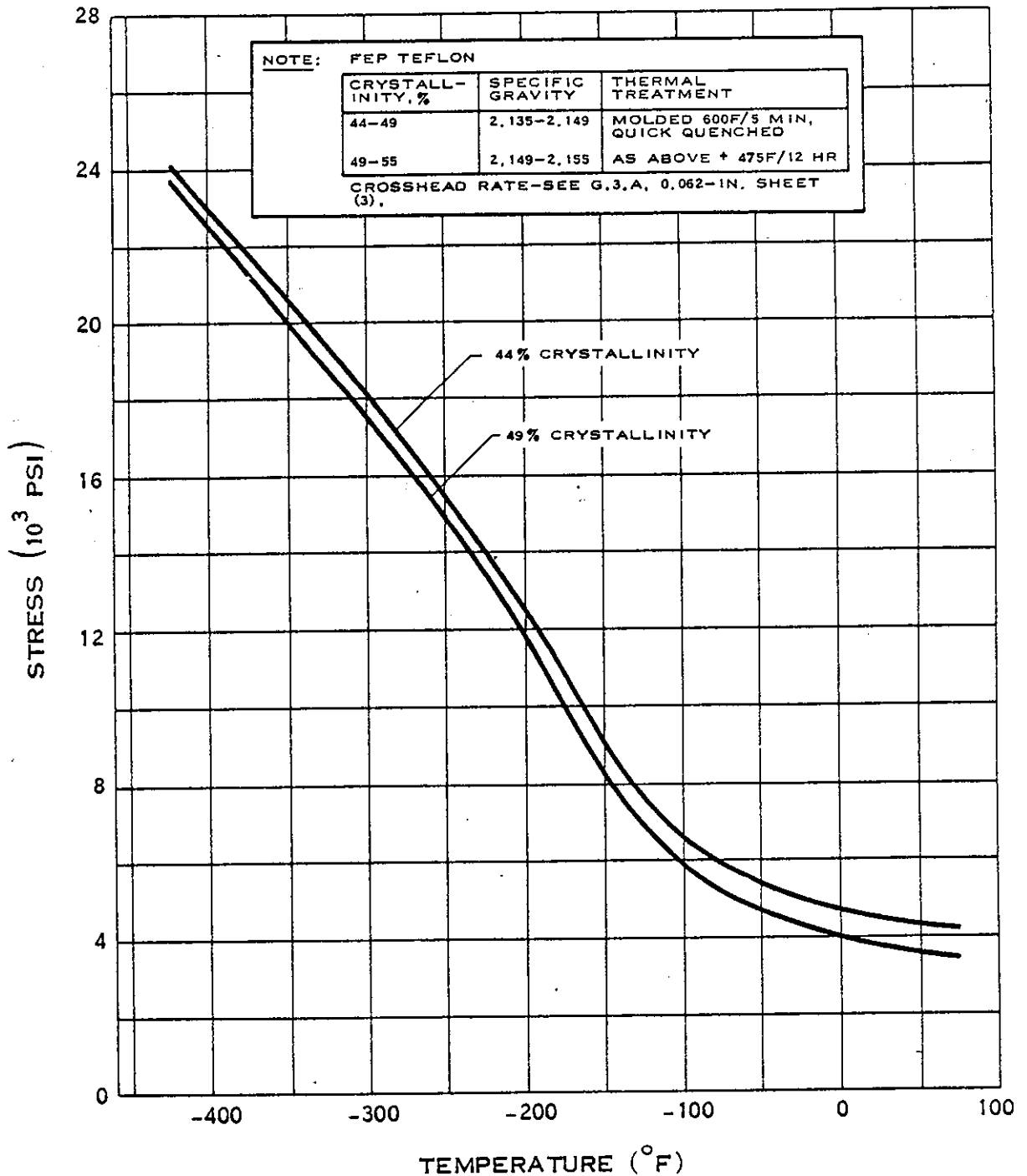
## TENSILE STRENGTH OF TEFLON\* (TFE)

\*T.M.  
E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et al., Cryogenic Materials Data Handbook,  
/ Tech. Documentary Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

# G.3.b-1



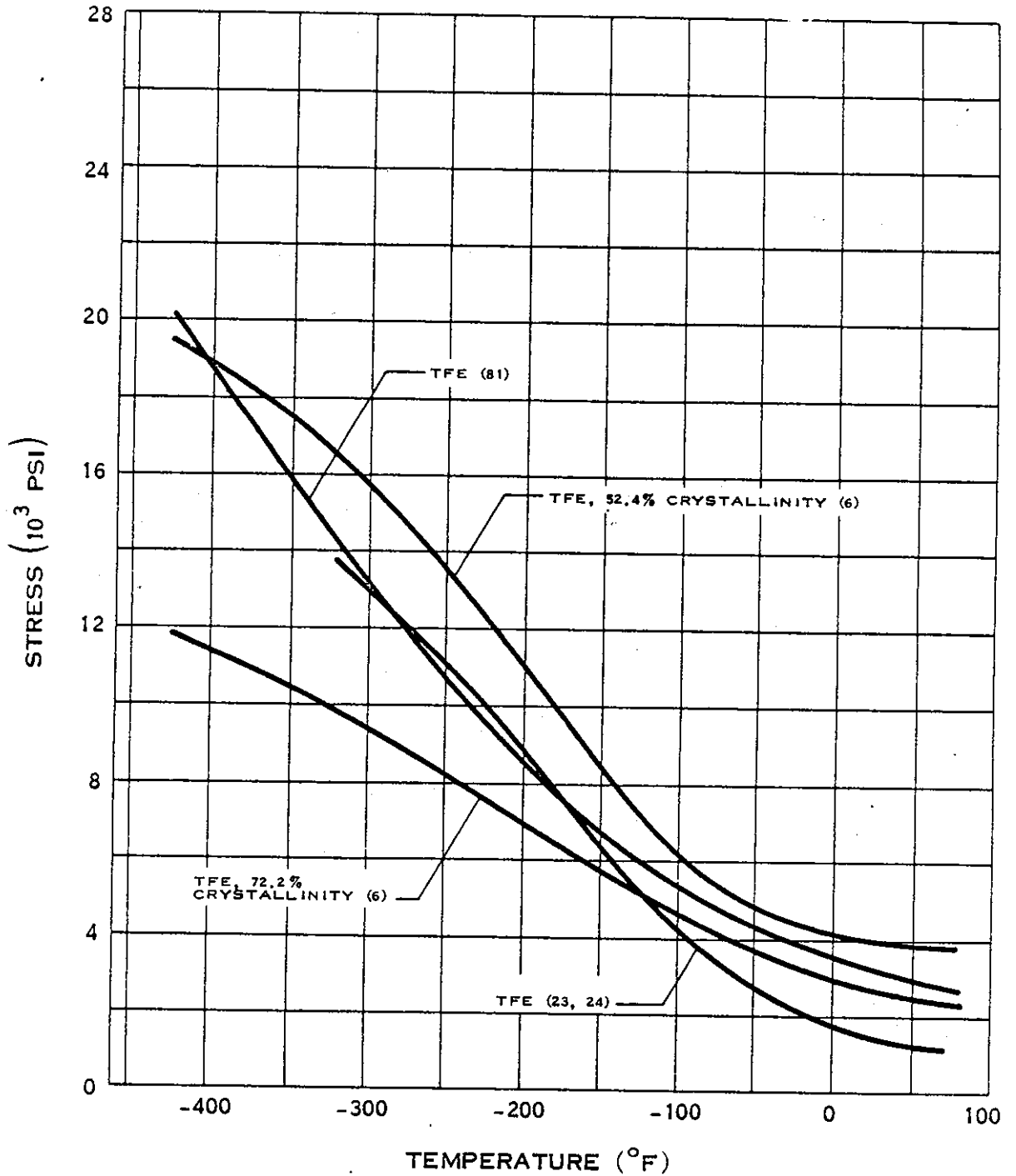
## TENSILE STRENGTH OF TEFLON\* (FEP)

\*T.M. E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg., et al., Cryogenic Material Data Handbook,  
 Tech, Doc. Rpt. ML-TDR-64-280 AD609562,  
 Air Force Materials Lab., Aug. 1964

# G.3.b-2



## TENSILE STRENGTH OF TEFLON\* (TFE)

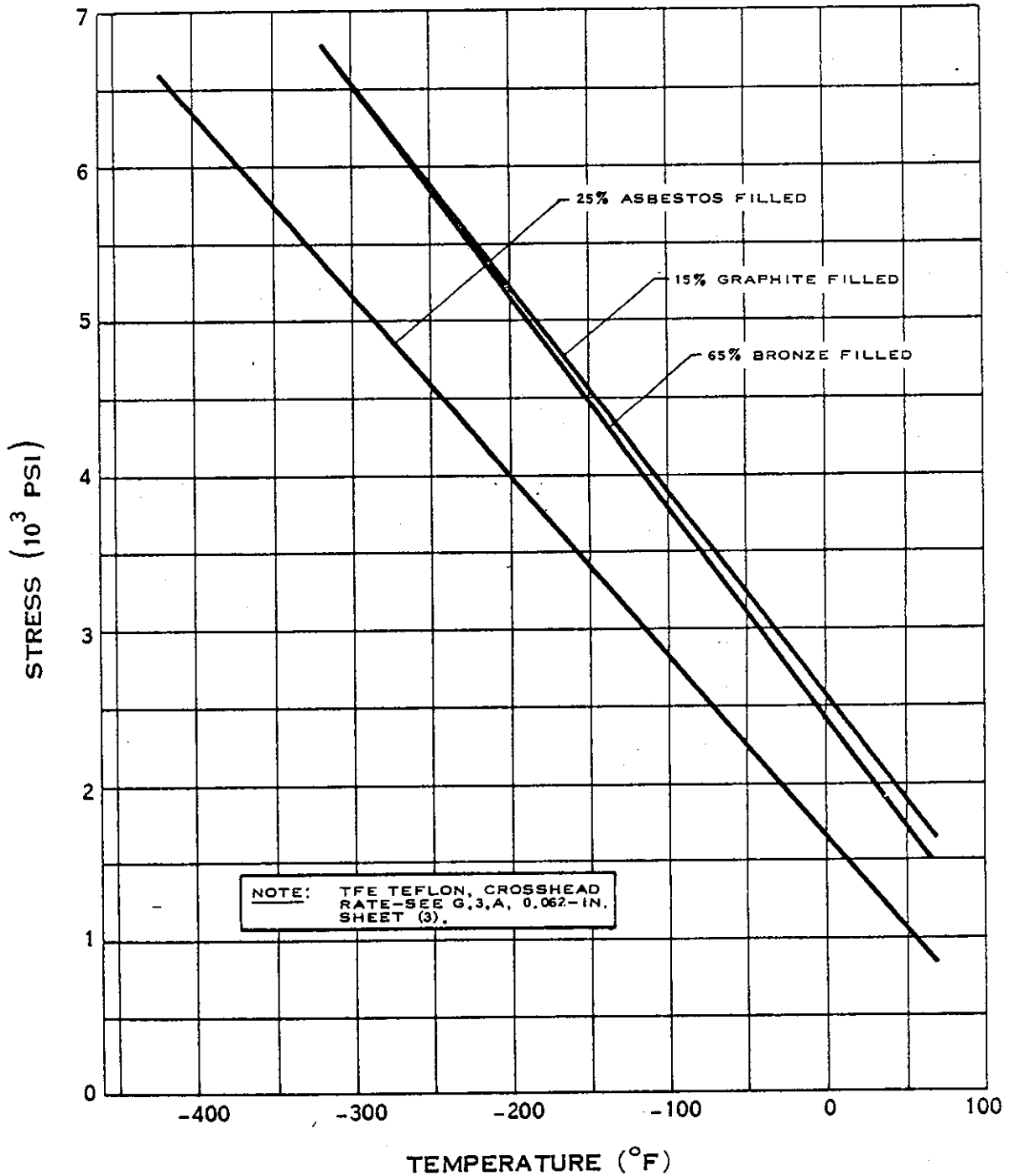
\* T.M.  
E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg., et al., Cryogenic Material Data Handbook,  
Tech. Doc. Rpt. ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964.



# G.3.b-3



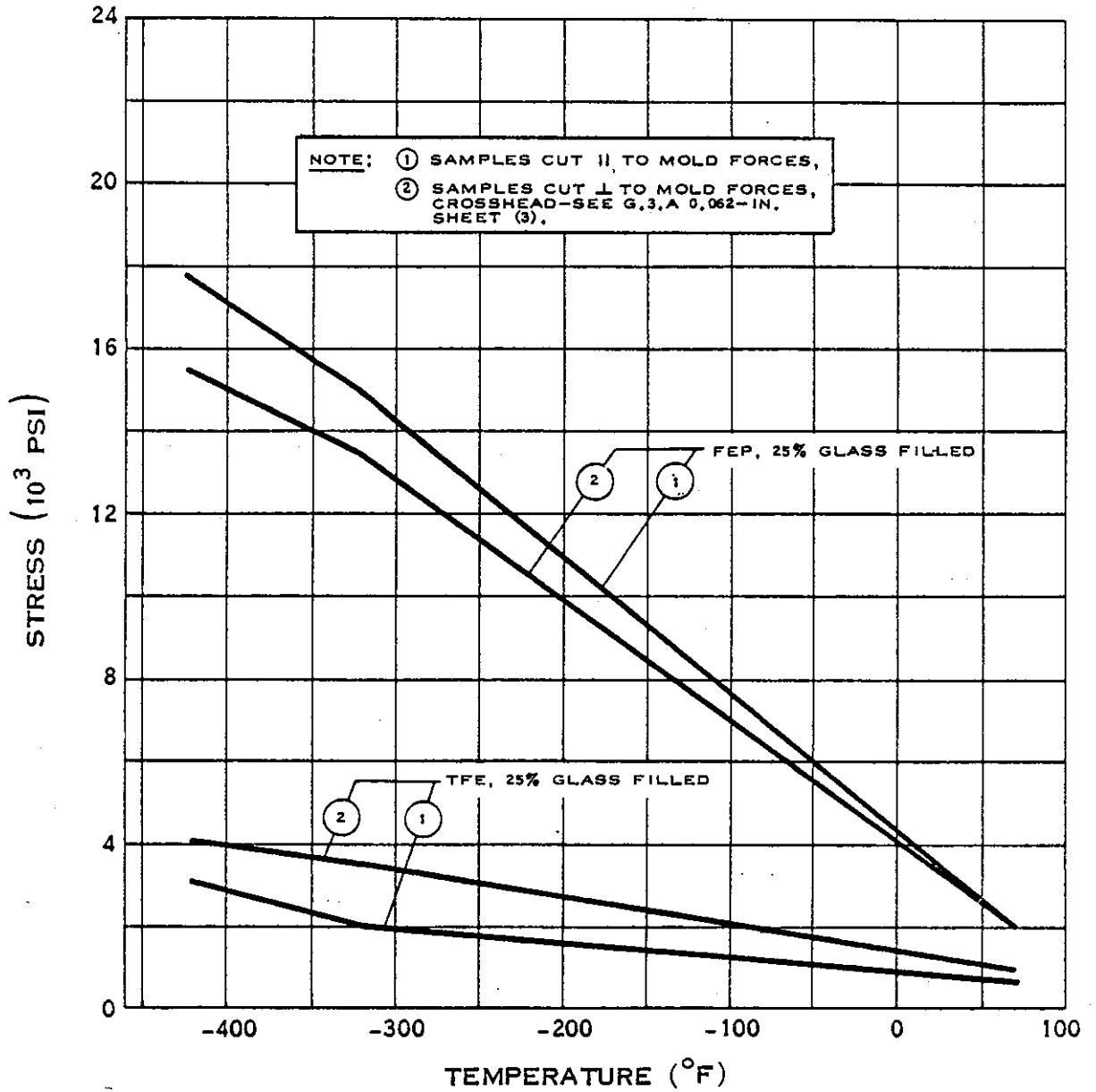
## TENSILE STRENGTH OF TEFLON\* (TFE)

\* T.M. E.I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et al., Cryogenic Material Data Handbook,  
Tech. Doc. Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

### G.3.b-4



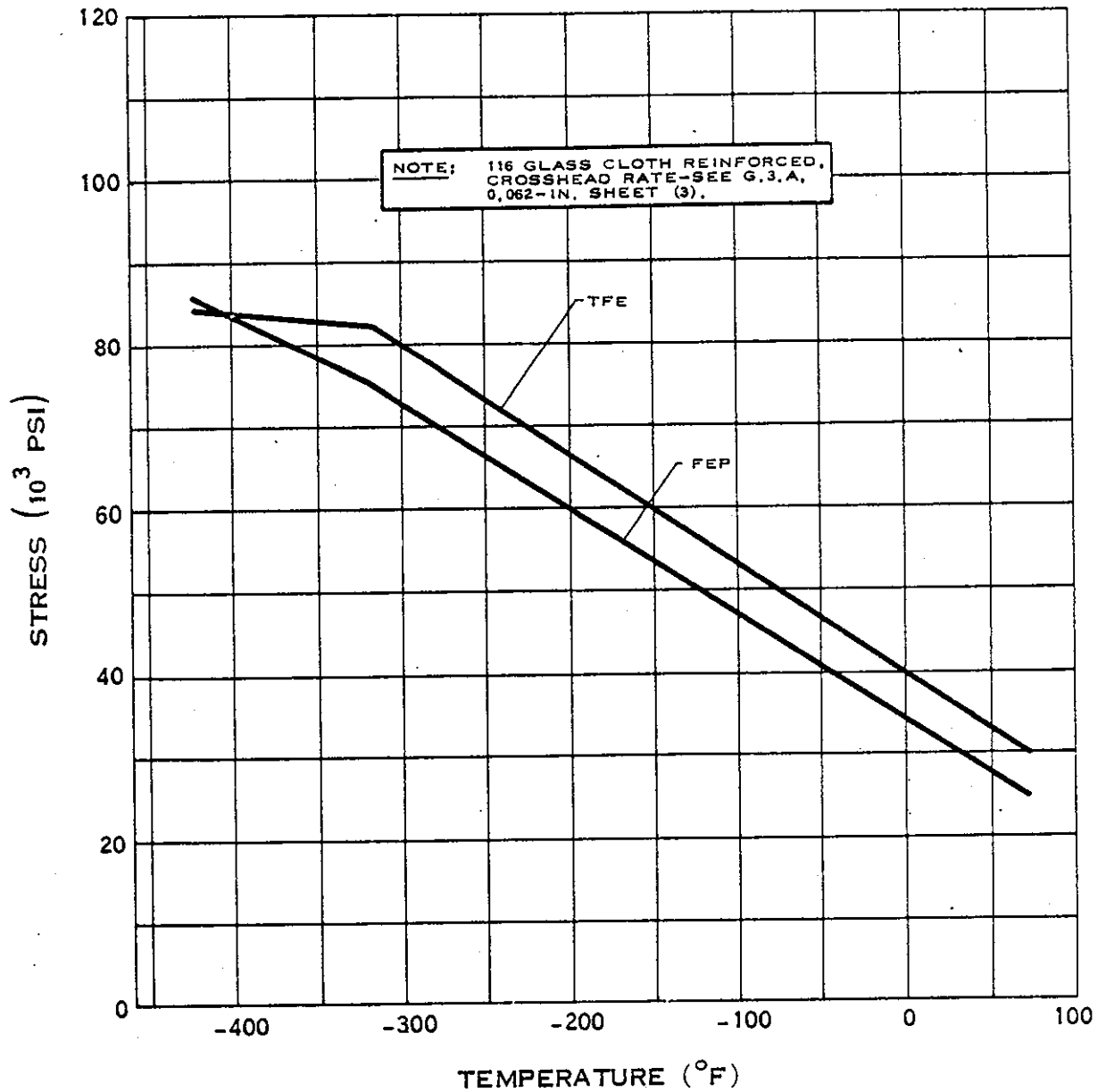
## TENSILE STRENGTH OF TEFLON\*

\* T.M.  
 E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et al., Cryogenic Material Data Handbook,  
 Tech. Doc. Report ML-TDR-64-280 AD609562,  
 Air Force Materials Lab., Aug. 1964

# G.3.b-5



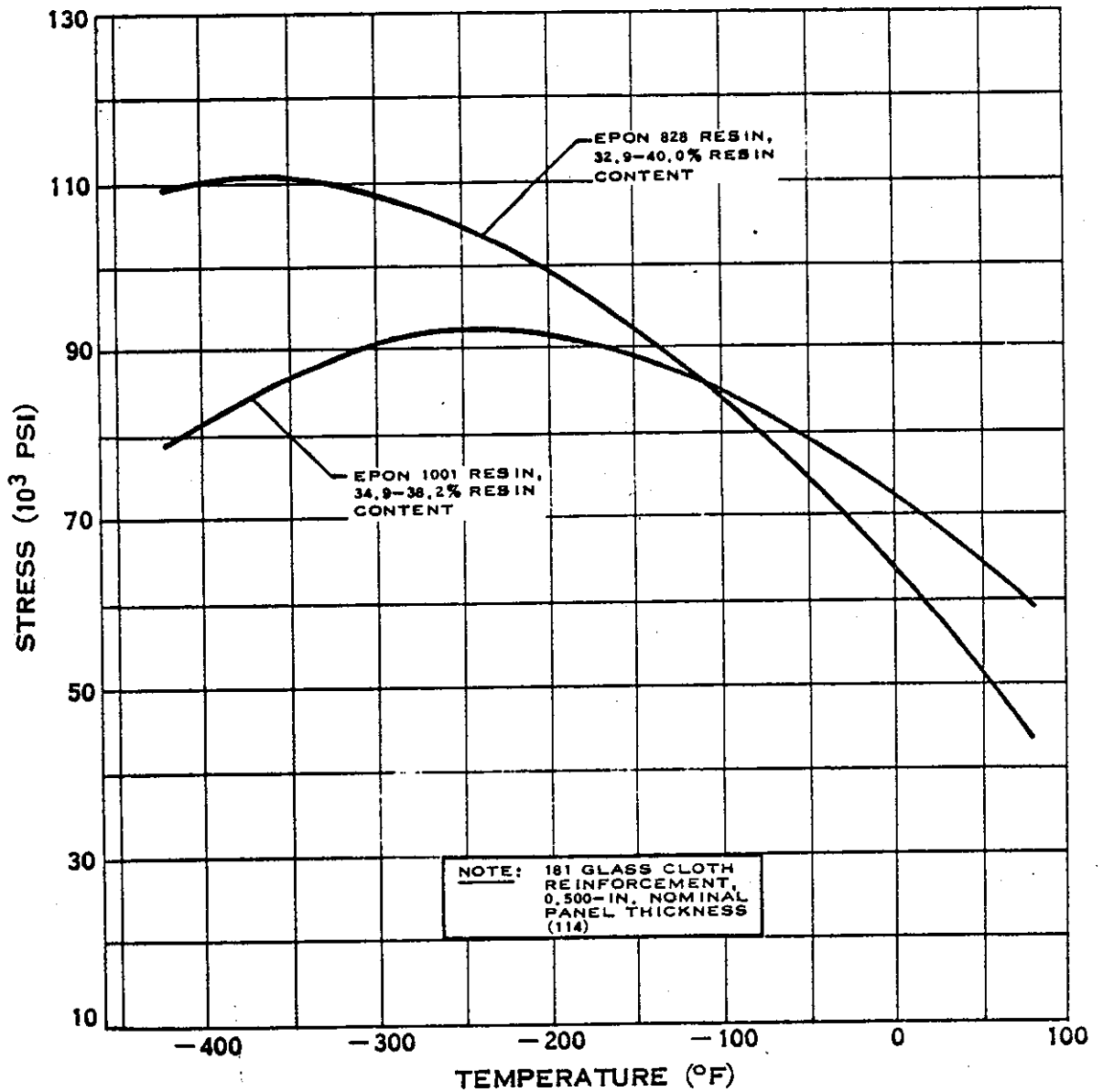
## TENSILE STRENGTH OF TEFLON\*

\* T.M. DUPONT DE NEMOURS AND CO.

Schwartzberg, et al., Cryogenic Materials Data Handbook,  
Technical Documentary Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug., 1964

(7-64)

H.I.m



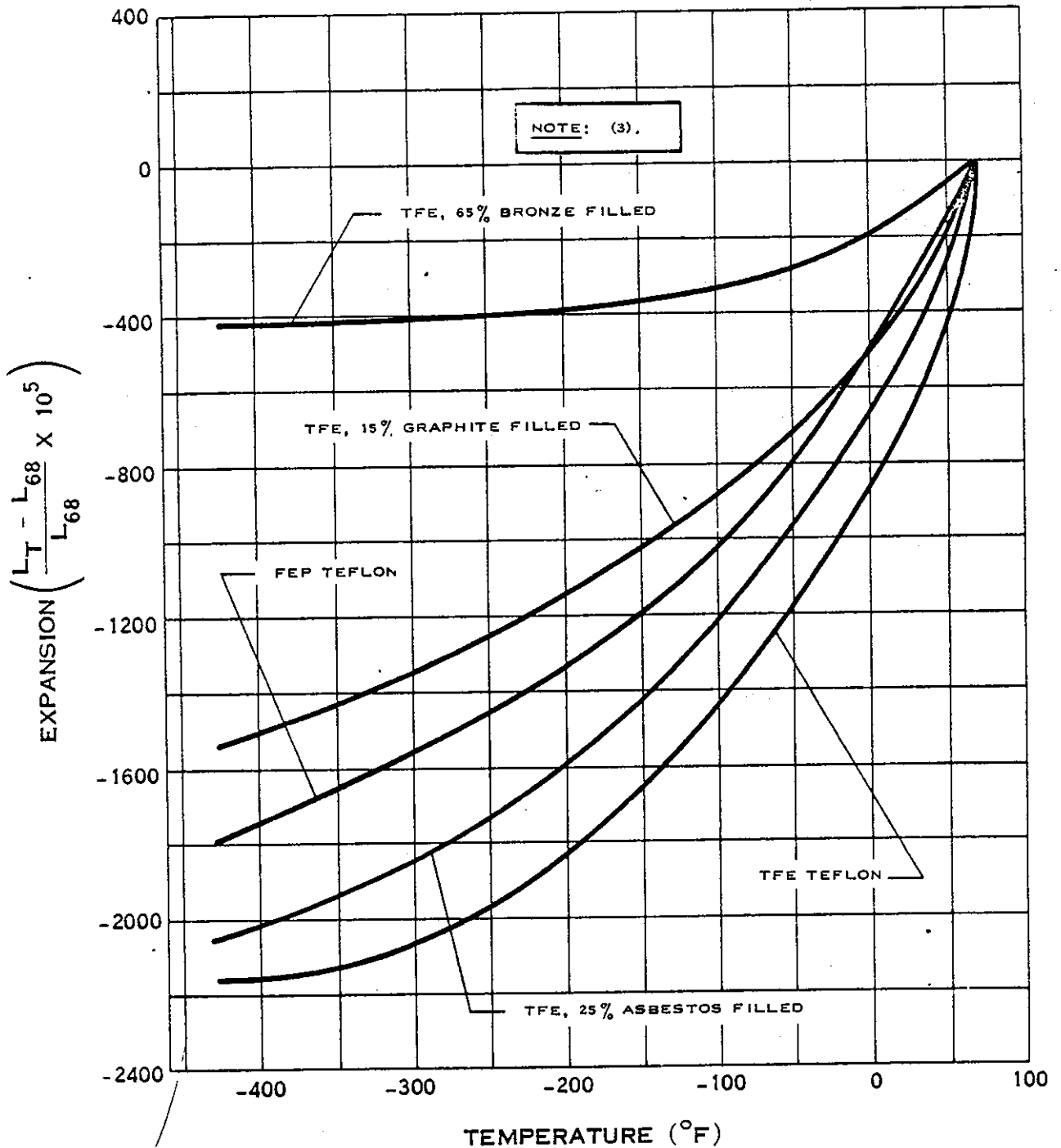
### COMPRESSIVE STRENGTH OF EPOXY - FIBERGLASS LAMINATE

Schwartzberg, et al., Cryogenic Material Data Handbook,  
Technical Documentary Report ML-TDR-64-280 AD609562,  
Air Force Materials Lab., Aug. 1964

(7-64)

T

### G.3.t



## THERMAL EXPANSION OF TEFLON\*

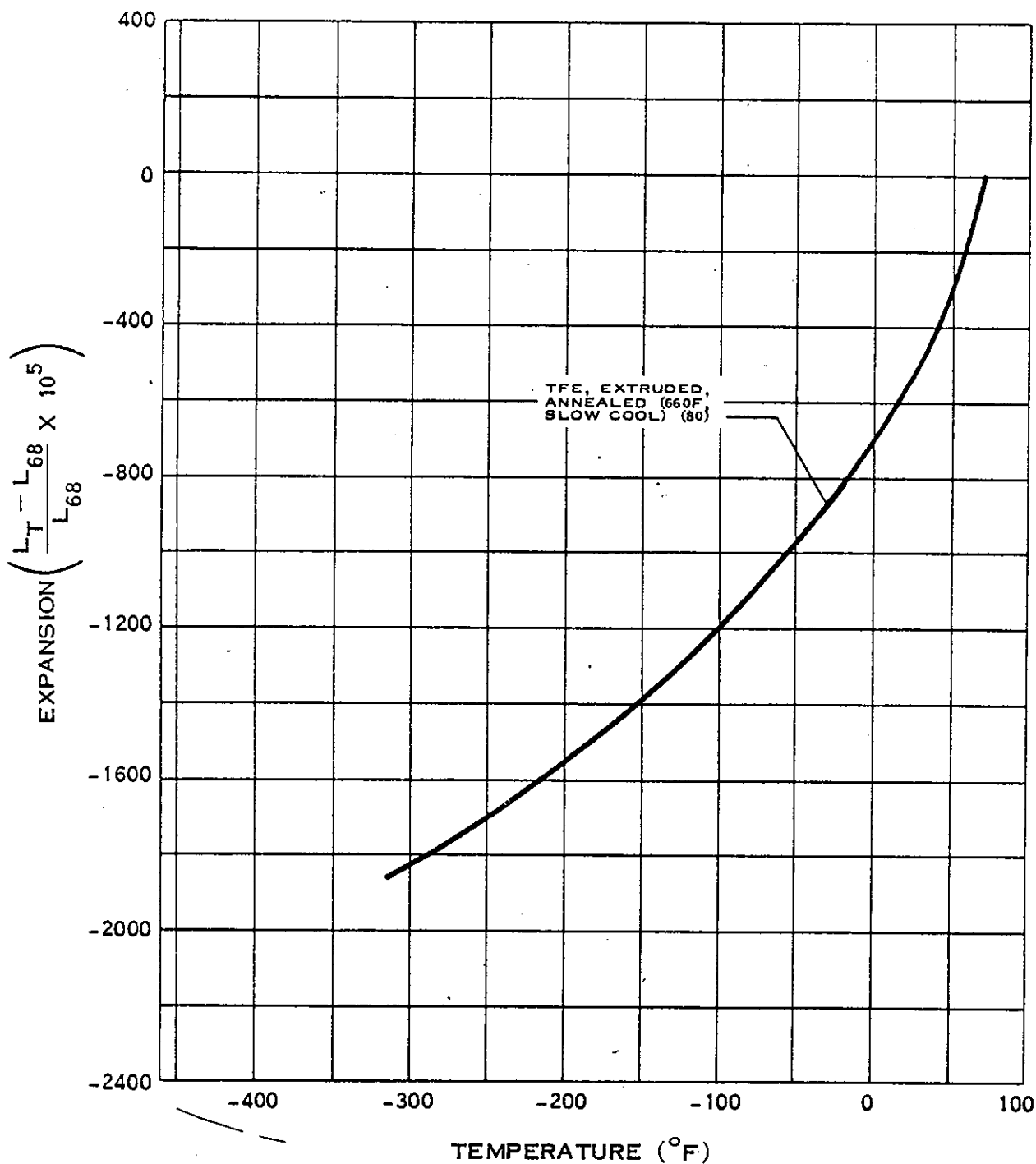
\* T.M.  
E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et al., Cryogenic Material Data

Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964

G.3.t-1



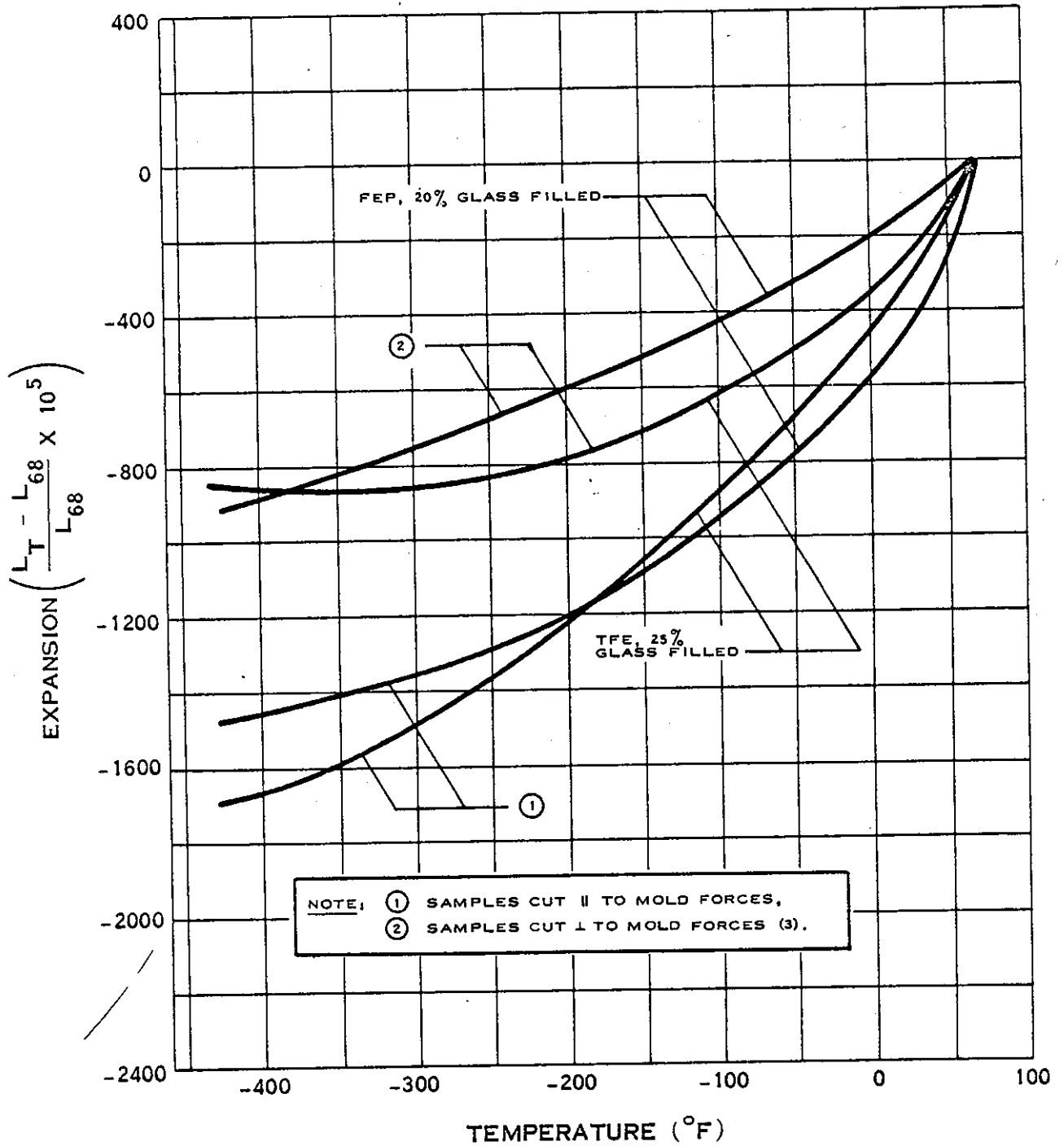
### THERMAL EXPANSION OF TEFLON\*

\* T.M. E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et. al., Cryogenic Material Data  
Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964

G.3.t-2

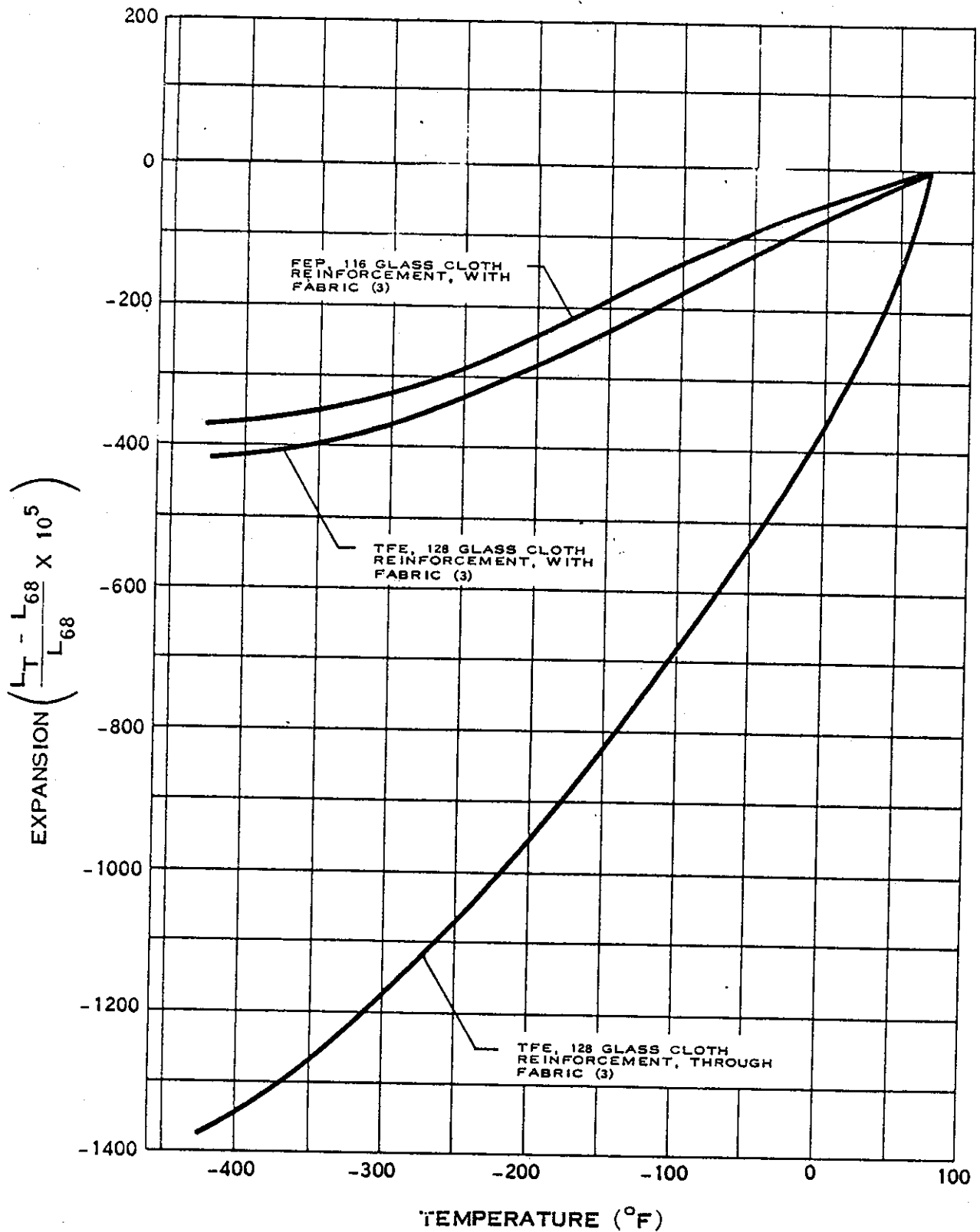


THERMAL EXPANSION OF TEFLON\*

\* T.M. E. I. DUPONT DE NEMOURS AND CO.

(7-64)

Schwartzberg, et al., Cryogenic Material Data  
 Handbook, Technical Documentary Report ML-TDR-64-280  
 AD609562, Air Force Materials Lab., Aug. 1964



**THERMAL EXPANSION OF TEFLON\***

\* T.M.  
E. I. DUPONT DE NEMOURS AND CO  
(7-64)

Schwartzberg, et al., Cryogenic Material Data

Handbook, Technical Documentary Report ML-TDR-64-280  
AD609562, Air Force Materials Lab., Aug. 1964